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QUANTIFYING REACTIVE MANEUVERS.

THESIS

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John J. Alt Capt USAF



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QUANTIFYING REACTIVE MANEUVERS

THESIS

Presented to the Faculty of the School of Engineering
of the Air Force Institute of Technology
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in Partial Fulfillment of the
Requirements for the Degree of
Master of Science

Ъу

John J. Alt, B.S., M.B.A.

Capt

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Preface

This study was undertaken because of interest from the Strategic Air Command (SAC) and the Air Force Avionics Laboratory (AFAL). Mr. James J. Foreman of the AFAL was instrumental in helping with the formulation of the experiment which was the heart of this research.

In the course of the study, I found that the elements for determining a value for reactive maneuvers are available. I identified those elements and hope to pursue this research at a later time.

I am thankful for the continuing assistance of Mr. Foreman throughout this study. I am grateful to Lt Colonel Pete Bobko, my advisor, and Major Dan Fox for their guidance and support during this thesis. I would also like to thank the following individuals for their material support. They are Mr. William McQuay (AFAL), Captain Dick DeRoos (SAC), and Dr. Robert Nullmeyer and Mr. Dave Grove of the University of Dayton Research Institute. Finally, I wish to express my appreciation to my wife, Sheri, for her patience and support, and for typing this final manuscript.

John J. Alt

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Abstract

There is currently no value of survivability attributed to an aircraft's reactive maneuver capability. In this experiment, exposure to enemy ground threats for various levels of information feedback to the aircrew were compared. This was done in an attempt to isolate the maneuverability factor. The Threat Model Penetration Simulation Analysis (TMPSA) model produced by the University of Dayton Research Institute was the penetration model used. The conclusion of this experiment was that only order of magnitude differences in capabilities can be captured with this model. It is recommended that two simple changes be made to TMPSA. These changes would allow more precise values for reactive maneuvers to be de-

QUANTIFYING REACTIVE MANEUVERING

I. Introduction

The purpose of this research was to examine the problem of quantifying aircraft maneuvering in response to electronic warnings of ground based threats. The objective is to examine one approach to solving the problem of quantifying aircraft maneuvering.

The research is limited to examining only bomber-type operations. It is hoped that the methodology developed can be extended to other aircraft by making the minimum number of assumptions necessary to achieve the objective stated above.

The EW Planning Process

Planning bomber operations begins with receipt of the target list. In the case of the Single Integrated Operations Plan (SIOP), this is prepared by the Joint Strategic Target Planning Staff (JSTPS). Strategic Air Command then plans aircraft sorties based on the targets on the list. In a conventional war, targets are assigned by the theater commander. In this instance, the bomber planners are cooperating with the theater commander's staff to develop the operations plan. Figure 1 shows generally how an electronic warfare (EW) plan is developed.

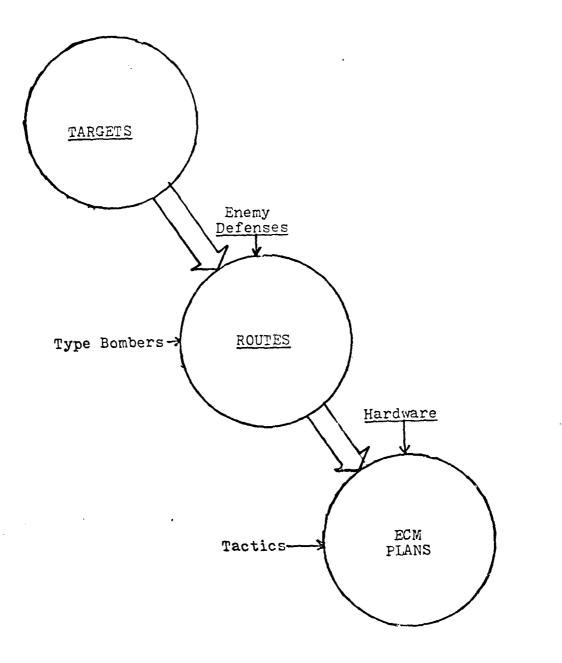


Fig. 1. The Domber EW Planning Process

With the targets assigned, the planners must now determine a feasible route for the bomber force. The goal is to achieve the objective, target destruction for example, with minimum losses. It is the responsibility of the Electronic Warfare Support Division (ESM) to gather information on the nature of the enemy's electronic defenses. The ESM staff is usually part of the intelligence directorate. With the raw data gathered by ESM operations, as well as other intelligence sources, an enemy radar order of battle is developed. type of radars and their locations, functions, and characteristics are determined or estimated. This intelligence estimate is used to plan the aircraft routes. Known point defenses, such as surface-to-air missiles (SAMS) and antiaircraft artillery (AAA), are avoided. Weaknesses in the electronic defensive network are exploited. Some of these weaknesses may be gaps in the radar coverage, low saturation level of the local command and control net, and poor types of equipment. Where electronic defenses must be penetrated, detailed information on these defenses is made available to the aircrews for study. With tentative routes established, the Electronic Countermeasures (ECM) plan is prepared.

The ECM plan consists of determining what hardware and tactics to use on the mission. The hardware is made up of the aircraft selected for the primary mission and the aircraft selected for support roles. Selection of the bomber is based on performance characteristics, such as range and speed, as well as ECM capabilities. Some bombers, such as

the B-52, have an assigned strategic mission. In this case. the ECM equipment on the airplane is tailored to the strategic mission. Some bombers do not have adequate built-in ECM equipment. On some missions, even the extensive equipment on a B-52 may not be sufficient to ensure a high probability of success. In these cases, support aircraft may be included as part of the plan.

Two examples of ECM support aircraft are stand-off jam (SOJ) platforms and defense suppression aircraft. The SOJ platform has a pure jamming and deception role. The SOJ aircraft flies out of enemy weapons range and uses high powered electronic equipment to jam and confuse enemy radio and radar operators. The primary defense suppression aircraft is called the Wild Weasel. The job of the Wild Weasel is to find enemy radar controlled SAM or AAA batteries and destroy them. Selection of the hardware depends on the tactics to be used; and the tactics to use depends on the hardware. That is, tactics and hardware are interdependent.

The ECM tactics employed are primarily dependent on the type of operation. The timing of the SIOP is designed in such a way that the bombers saturate the enemy's defenses in one area, then fly individual routes to the targets. During the initial phase of tha attack, the bombers support each other electronically. As the bombers diverge, the individual routes are designed to exploit enemy electronic weaknesses, such as those mentioned above. Each aircraft must then be prepared to defend itself. The self-protection tactics used

are developed by determining what equipment the enemy has and how he uses it, developing and testing new ECM equipment or tactics to counter the enemy capability, and training the crews to use the equipment and tactics thus derived (see Figure 2).

This same process is followed to develop the equipment and tactics for the use of massed bombers in a tactical operation. In the case of a tactical operation, self-protection may not be part of the ECM plan. Many other tactics are available. These ECM tactics include defense suppression, stand-off-jamming, chaff clouds or corridors, electronic confusion, and electronic saturation through the use of decoys. Thus, the choice of tactics is based on the type of operation, the hardware available, the routes chosen, and the targets assigned. None of the criteria for electronic warfare (EW) planning mentioned above is considered in isolation. The routes, targets, hardware, tactics, and type of operation are interdependent. Each is considered in light of the others before the final operations plan is established.

The Problem

This description of EW planning is only an overview.

The interdependence of the planning factors coupled with the growing diversity of the enemy's anti-aircraft equipment and organization pose complex problems for the operations planners.

Many of these problems have been solved through the use of computer models and simulations. Models have been devised to measure the effectiveness of ECM equipment against radars of

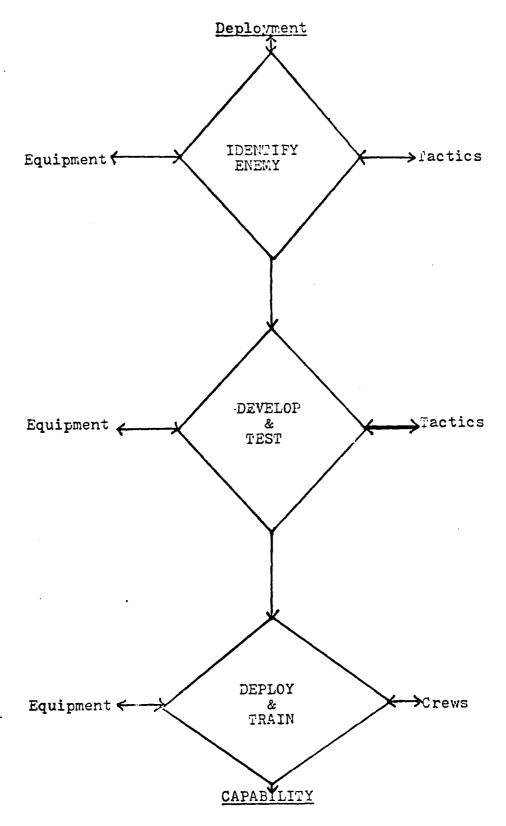


Fig. 2. ECM Tactics Development Process 6

most types. Models for many different scenarios have been developed, but none of these models has successfully included reactive responses (Ref 1).

One of the specific problems of this genre concerns determining a value, or modeling an aircrew reaction to a perceived threat. Although the problem of how to model maneuvers has been present in the past, recently it developed a greater importance.

The increasing mobility of the Soviet anti-aircraft forces is significantly complicating the job of operations planners. All of the latest Soviet anti-aircraft equipment is mobile (Ref 13:49). As a result of this mobility, a number of EW planning tactics are no longer as useful as before.

One of the main tactics of the EW planner is to avoid enemy defenses. As a result of the enemy's mobility, the planner is reduced to planning based on uncertain defensive positions. Avoidance is not as credible a tactic in this situation.

A second tactic, which is severely degraded by enemy mobility, is exploitation of the enemy electronic defense network. With a mobile air defense force, the enemy is capable of filling gaps in radar coverage; and, he can move equipment to locations where poorer equipment is operating. Today's weak spot may be tomorrow's strong point. Thus, the value of another tactic is reduced.

The last problem caused by the mobile defense concerns the impact on the aircrews. Before the mobile forces were

deployed, the crew could study a mission and be well prepared for the defenses to be encountered. Now the aircrew must be prepared to counter any and every threat the enemy can field.

As a result of the above, most of the problem of defeating enemy defenses rests on the aircrew's reaction once the threat is perceived. The planner's problem remains one of determining the best application of the forces available. This problem is complicated by a more dynamic battle situation.

The key factors of the problem of aircrew reaction to perceived threats are the aircrew detecting the enemy radar, the enemy radar operator identifying the bomber, and the subsequent reactions of both sides in the ensuing battle. If radar or some other electromagnetic device is not used, the situation is not an EW problem and is beyond the scope of this paper.

The aircrew detection and enemy radar operator target identification processes are two sides of the same problem. When the radar detects the bomber, the radar operator must see the "blip" on his scope and determine that the blip represents a bomber. Conversely, when the radar signal triggers the electronic warning equipment on the bomber, the aircrew must recognize the signal and the weapon type the radar supports. Although the problems of aircrew detection and radar operator interpretation are beyond the scope of this paper, the reaction times determined in the studies of these problems can be used in the research model.

The second key factor of the problem is responses of the

operator or crewmember to the detection. The specific actions of the radar operator are beyond the scope of the paper, except that they result in firing of a SAM or AAA.

The reaction of the aircrew can be active electronic countermeasures, physical maneuvering of the aircraft, neither of these tactics, or both of these tactics. Active electronic countermeasures are jamming the enemy radar signal, dropping chaff, and employing electronic deception techniques. Consideration of these reactive tactics is beyond the scope of this study. The problem of reactive ECM is an area that has not been modeled as yet. It is a considerably more complex problem than the problem to be addressed.

The second possible action available to the aircrew is to maneuver the aircraft to minimize exposure to the ground threat. This is an application of the avoidance tactic. As stated above, the objective of this paper is to examine an approach to solving the problem of quantifying the value of reactive maneuvering. With this value determined, the planning staff will have a better understanding of how much force will be necessary to achieve an objective.

Review of EW Modeling

Modeling EW is a relatively new development. The introduction of sophisticated radar-directed anti-aircraft weapon systems by the Soviets resulted in the need to develop airborne systems that warn aircrews of the impending attack. Other equipment was needed to deny the enemy radar operator location information about the aircraft. Modeling and simulating developed as tools to evaluate ECM equipment and tactics.

One of the earliest comprehensive efforts at simulating EW was the formulation and construction of the USAF REDCAP electromagnetic simulator in 1964-1965. The simulator was capable of evaluating ECM equipment against a single tracking radar. More radar channels and a variety of capabilities, such as chaff simulation, have been added to the simulator. Today the system is capable of simulating an entire air defense region against an attack by hundreds of aircraft. This simulator has been used extensively by the U. S. Air Force to evaluate tactics, ECM concepts, and EW hardware (Ref 3:1-2).

The history of the REDCAP simulator is a typical example of how simulation of EW by digital computers has grown. Today there are many models of air warfare which include EW (Ref 1). However, none of these models has adequately modeled reactive EW, including maneuvers (Ref 5).

A model, called the Threat Model Penetration Simulation Analysis (TMPSA), was developed for the Air Force Avionics Laboratory by the University of Dayton Research Institute to determine whether more accurate knowledge of threat location by a penetrator would enable the aircraft to increase its probability of survival. In the model, an aircraft seeks to maximize its probability of survival given knowledge of all threats lying within a certain distance (Ref 19:1-2). This flight path generation model is a first step in putting a value on maneuvers. The model accomplishes this by attempting

to minimize the amount of time the aircraft is exposed to lethal enemy ground fire. This research effort proposes to improve TMPSA as described above by refining the assumptions and explicitly accounting for some uncertainties assumed to be constants in the TMPSA model.

The Approach

The approach used is to compare feedback loops to determine a minimum lethality time. Feedback loops are information flows. The specific loops used are described later. The minimum lethality time is a function of the lethality of a defensive system at various ranges and the amount of time the aircraft spends within these particular weapon ranges.

In the postulated situation shown in Figure 3, the aircraft must navigate from the starting point (S) to the finish point (F). The aircraft track is denoted by the dotted line. The solid parallel lines represent the corridor the aircraft must remain within. When maneuvering is allowed, these represent the maximum lateral travel allowed for the bomber. Each circle represents a fixed enemy SAM or AAA unit. Three types are noted in Figure 3 as T_1 , T_2 , and T_3 . The total lethality time is computed by multiplying the amount of time the aircraft is in each circle, which is a constant (Δ t), by the lethality of the circle (P_{k1}^T , P_{k2}^T , P_{k3}^T). The approach used in the TMPSA model is to divide the lethal radius of the SAM or AAA battery into segments with an assigned probability of kill based on the range and azimuth of the aircraft to the battery site (Ref 19:2). Figure 4 is an example of a

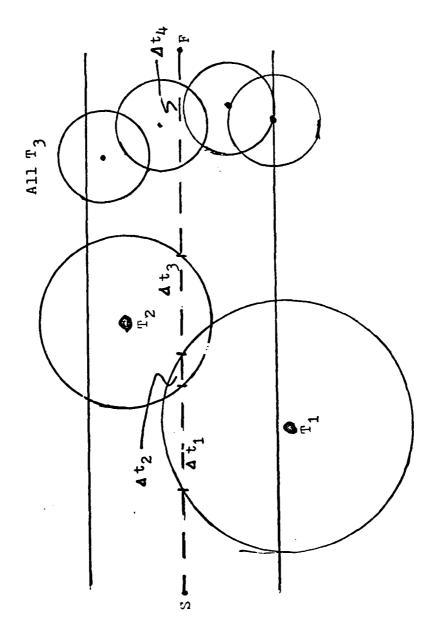


Fig. 3, Generalized Mission

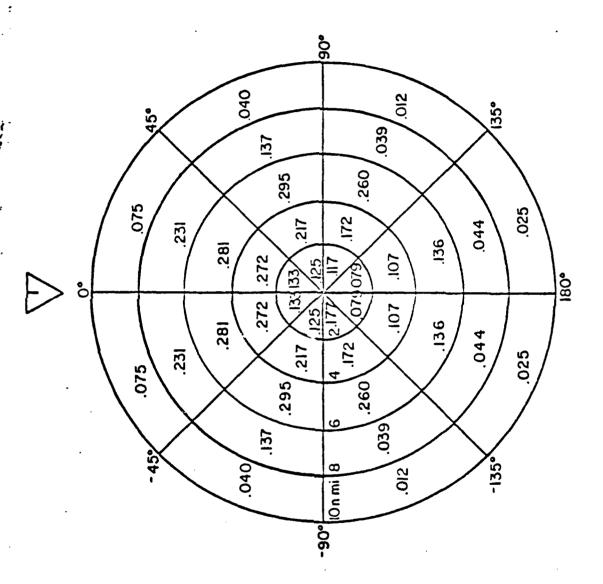


Fig.4. Example Antiaircraft Site Template

site template used in TMPSA.

Three levels of feedback will be examined. First, the situation will be considered with no feedback. Next, a feedback loop will use the TMPSA model flight path generator to determine the reaction with perfect information. The final feedback loop will attempt to model manual reaction based on less than perfect knowledge about the enemy defensive system locations. The missions with these three levels of feedback will be called the preplanned mission, the automatic mission, and the manual mission.

The purpose of the preplanned mission is to determine the total lethality time of the aircraft in the situation when no knowledge of the threats is available to the aircrew. In this situation, the aircraft is flown over its preplanned route and its total exposure to lethal enemy weapon effects is computed.

The preplanned mission will be constructed using the mission planning assumptions of an operational Air Force staff against a random distribution of ground threats. As part of the study of the unplanned mission, an additional unplanned threat will be introduced. The new situation posed by this threat will be examined to see how it changes the total lethality of the penetration. The threat location will be changed for each iteration of the model so the entire gamut of interactions from collocation with other defenses to no overlap of defensive fire can be examined.

The automatic mission will start with the same mission

as the preplanned, but as the airplane penetrates, the crew is assumed to have perfect knowledge of where all threats are once they come within the awareness range of the penetrator. Again, the total lethality will be determined, but this time the crew will be allowed to react based on the given information. As a variation of the automatic mission, a second set of lethalities will be computed for the case where small, random range and azimuth errors occur, thus simulating sensor limitations. In this variation the crew will no longer have exact knowledge of the threat location. Both of these automatic loops will use the flight path generator of TMPSA to determine the aircrew reaction to the threat.

In the manual mission, the input to the crew will be similar to the imperfect automatic loop. Crew reactions will be modeled based on reaction times from studies of human responses and the tactics prescribed by the major commands for these type engagements. The result will be lethalities as in the earlier cases.

This step-by-step approach to the problem should result in a realistic value for aircraft reactive maneuvering. This algorithm and the value it produces for maneuvering should enable planners and decision makers to employ forces more efficiently. An interesting result of determining the maneuver value is that it provides a way of determining the value of knowledge of enemy defensive locations. The emphasis of the approach, however, will be to proceed in small increments to achieve the objective of quantifying reactive maneuvering.

Chapter II The Preplanned Mission

Introduction

The purpose of this chapter is to establish a control model and scenario against which later refinements in the model can be compared. To this end the relevant planning assumptions used by the JSTPS will be outlined. Using these assumptions, a route segment will be defined, a threat array established, and the threat site template described. With these parameters established, the first basic model will be defined and run results shown. Finally, an additional threat will be added to the scenario and the model will be adjusted to treat this threat as having an uncertain location. This second basic model will serve as the control model.

JSTPS Planning Assumptions

The JSTPS attrition methodology applies the threat model to the penetrator on a one-on-one basis. The output of each engagement is a probability of kill (P_K) of the penetrator. These probabilities are then combined in series to yield a probability of arrival (P_A) at any particular point along the penetrator route. The computation is accomplished as follows:

$$P_{A} = (1-P_{K1})(1-P_{K2})...(1-P_{KN})$$
 (1)

for N threats encountered (Ref 17:3-4). In Chapter one, it was stated that comparison between the control model and the

modified models would be based on the lethality time. Since the JSTPS model determines a probability of arrival, these two approaches must be reconciled.

The JSTPS attrition model computes a probability of kill (P_K) for each threat site encountered. The P_K is a function of the probability that the threat successfully 1) detects and tracks the penetrator (P_d) , 2) fires its weapon (P_s) , and 3) the weapon, missile or AAA rounds, cause lethal damage to the penetrator (Ref 17:8-10). Mathematically this is:

$$P_K = (P_d)(P_s)(P_k)$$

Time is treated by determining how many shots (N) the site can make before the penetrator is out of enemy weapon range. This computation considers the geometry of the penetrator and threat site engagement, and the ability of the threat site to reengage the penetrator (Ref 17:11). This results in the $P_{\rm K}$ formula being revised slightly.

$$P_{K} = (P_{d})(P_{s}) \left[1 - (1 - P_{k})^{n}\right]$$

When all P_K 's are derived for the aircraft route, Equation 1 is used to compute the probability of arrival.

The TMPSA methodology is different in the way it treats time and in the resulting output. Each threat site template has a probability of kill for each segment. These probabilities are static probabilities which are functions of the range and azimuth of the penetrator to the site (Ref 19:2-5). The TMPSA program sums all the kill probabilities for all sites within whose lethal range the aircraft is located ($P_{\rm KT}$).

Then the incremental lethality time, called exposure (Δ E), is determined by:

$$\Delta E = P_{KT} \cdot \Delta t$$

where Δ t is the time increment. Total exposure E over an entire flight path is then:

$$E = \sum_{N} (P_{KT})_{N} \cdot \Delta t$$

where N is the number of time increments in the flight path (Ref 20:3-4).

The lethality time and the probability of arrival are determined by the same inputs. However, the TMPSA methodology considers the penetration problem using a fixed time increment. The JSTPS methodology is event oriented where time is a subroutine. The result is that TMPSA produces a scalar output (probability X time) and the JSTPS model produces a probability output (probability). The TMPSA result is a lethality time that is inversely related to probability of arrival (Ref 20:1).

Mission Segment Definition

For this research, a route segment was constructed based on the start and finish points, the track the aircraft flies, the distance from start to finish on track, and the speed of the penetrator. The symbols and their definitions for constructing the route segment are noted below.

XI = the starting point for the aircraft and the zero time location.

XF = the finish point for the aircraft and the run stop
time.

T = a straight line between XI and XF representing the penetrator track.

Any point on the ground can be measured from a point on the track by noting distance and angle measured clockwise from the track. Thus, along track is zero degrees. Although not too important in this model, it will have more meaning in later developments.

D = distance measured in kilometers between any two points on the ground.

VMN = speed of the penetrator. In this model, the
speed will be constant.

The last two parameters need further definition. The route segment is set at 100 km and the speed of the penetrator will be set at 350 knots. The length of the route segment was set arbitrarily. The sensitivity of the model to route length will need to be examined later. The penetrator speed represents a common B-52 low altitude training speed. As noted in Chapter one, bombers are being modeled. This is sufficient for this model, however, later model development will require specifying minimum and maximum speed limits. With the route segment defined, the threat array must be set.

Threat Array Determination

For this model, an artificial threat array is generated and threat system parameters will be arbitrarily selected.

However, for planning an actual bomber sortie, this would obviously be unnecessary. Having noted these caveats, the threat array determination method is outlined below.

A three-step method was used to establish the strategic (ie. fixed site) threat array. First, a corridor on each
side of the penetrator track was set based on the maximum
threat range. For this model, all threats represented the
same weapon system. Next, a grid system was devised as a
way of locating points on and around the route. Finally,
ten random numbers were selected to locate each threat on
the grid system.

Since the corridor limits depend on the range of the threat system in this model, the first step was to determine the threat range. In this case, the threat range was arbitrarily set at 10 km. The corridor width was selected to be 20 km (10 km on each side of track). For this corridor, a grid of one kilometer by one kilometer squares was considered appropriate. To establish the ten threat locations, ten random numbers were selected from the CRC Standard Math Tables (Ref 2:545). Lines one through ten of column eight were selected as the random number stream. The first two digits were taken as the x coordinate. The middle digit was ignored. The y coordinate was determined by the last two digits. The fourth digit was reduced to a zero or one. If the fourth digit was even, the digit became a zero, if odd, it became a one. The revised fourth digit and the fifth digit represented the y coordinate (eg. 75 translated to 15 and 85

translated to \emptyset 5). Table 1 shows the random numbers and the x and y coordinates derived. It later became evident that threats were needed outside the corridor. For this reason, another 27 random numbers were selected from the CRC Tables, column 7. The grid was expanded to include a distance equal to the awareness radius from each corridor at 25 km. In effect, the grid was now 70 km by 100 km. To establish each plotted position, the random number was divided as before. The first two digits represented the x coordinate. The middle digit was ignored. The y coordinate was more difficult to compute. If the fourth digit was zero through four, it was not changed. If it was five through nine, five was subtracted. If the last two revised digits were greater than 25, then 20 was added to the number to obtain the y coordinate. Otherwise, the last two digits are the y coordinate. Figure 5 is a plot of the final array. The final step in building the basic model was to define the threat template.

Threat Template Defined

The threat template represents the lethality of the threat system by range and azimuth of the penetrator from the threat site. The lethal radius of the threat system is divided into segments with an assigned probability of kill (P_K), as shown in Figure 4 (see Chapter one). As an aircraft transits a segment, the total lethality time of the penetrator is incremented by the segment P_K for each increment of time the penetrator is in the segment. Each segment P_K , in other words, represents lethality as described earlier

TABLE 1
Random Threat Site Location

Site Number	Random Number	Coord x	linate y
1 2 3 4 5 6 7 8 9	14194 53402 24830 53537 81305 70659 18738 56879 84378 62300	14 534 531 70 186 584 62	39 27 35 42 34 43 44 43 43 45
12 13 14 15 16 17 18 190 21 22 24 25 27 28 29 31 33 33 33 33 33 33 33 33 33 33 33 33	69179 27982 15179 394468 18602 7145940	621361795351364651879794668 975908147868677058360645069	929482450753094624989824268 11251026654165011

30.	29. 15 . 18 . 19 . 34 . 20 . 25 . 31 .	13• 35• 14• 32•	7° · · · · · · · · · · · · · · · · · · ·	• 7 .	28 • 16 · 33 • 38 • 23 • 23 • 26 · 22 • 37 • 33 • 38 • 23 • 23 • 24 • 22 • 37 • 37 • 38 • 23 • 23 • 33 • 38 • 33 • 38 • 33 • 33	27* 17* 36* 10 20 30 40 50 60 70 80 90 100 DISTANCE (Kilometers) - both axes	Fig. 5. The Basic Scenario
20	. 09	50	04	30	80 01	0	

in this chapter.

For this research, the threat template used was exactly the same as the template shown in Figure 4. The segments are two kilometers deep and subtend a 45 degree arc. The segment P_K 's are for a hypothetical terminal ground defense site. With the site template parameters set, we now turn our attention to the computer program to run the basic model.

The Basic Model

The computer program for the basic model is the TMPSA program. A copy of the program is in Appendix A. All variables are identified in this program listing.

The results for the basic model are shown in Appendix C.

The most important result is that total exposure equals 63.99.

The Control Model

The control model represents the no-feedback case. The other models developed will be compared to the results of this model.

There are only two changes to the input required. The number of threat sites is increased to eleven (NSITE = 11). With the increase in sites, another location is required. Using the same technique described earlier, the random number 56865 was selected from the CRC, line 11 column 8 (Ref 2:545). Since this site is not static, a range of locations is needed. It is noted that the only variability this new threat poses in this situation is with respect to the threat offset from the flight path. Thus, the x coordinate was

taken as 56 and the y coordinate was varied from one corridor limit to the other. The program output is in Appendix C. The run results are shown in Table 2.

TABLE 2
Control Model: Total Exposure Table

Mobile T	hreat Location	Total Exposure
<u>x</u>	Ā	
56.00	27.5	67.87
56.00	30.0	80.00
56.00	32.5	80.43
56.00	35.0	79.48
56.00	37.5	80.43
56.00	40.0	80.00
56.00	42.5	67.87

Chapter III The Automatic Mission

Introduction

In this chapter, results are presented for the case where the TMPSA program was used to determine the flight path through the threats. This is the opposite extreme situation from Chapter two. Thus, in this chapter we assume perfect knowledge of threat locations in determining the aircraft flight path. The previous chapter assumed no knowledge of threat locations.

The TMPSA program includes the capability to be used as the flight path generator. The procedures for determining the exposure were the same as used in Chapter two. The difference in using the full power of TMPSA is the ability of the program to choose among alternate routes. This is accomplished by the program through a change of the input data.

The procedure requires input of the aircraft speed (VMN) required to reach the finish point on schedule. It is important to note that VMN is the constant x component of the velocity. Next, the maximum speed (VMX) for the aircraft is input. This implies that the aircraft cannot deviate from the centerline of the corridor by an angle greater than:

$$\Psi$$
 = Arccos $(\frac{VMN}{VMX})$

If a deviation of greater than ψ is allowed, VMN would no longer suffice as the x direction velocity component. The third bit of information required is the awareness radius (R).

This is the maximum distance at which the aircraft becomes aware of the threat. Figure 6 shows the resulting geometry (Ref 19:4).

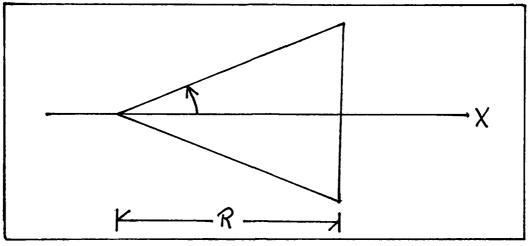
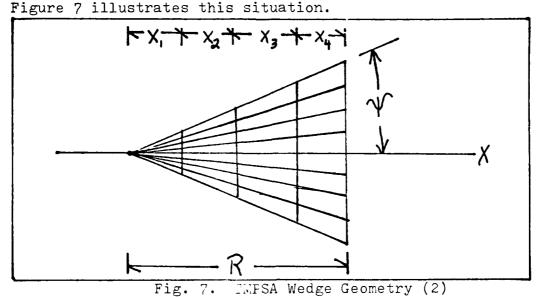


Fig. 6. TMPSA Wedge Geometry (1)

The next step was to divide ψ into a number of parts (J) which is input. This results in 2J + 1 rays emanating from the current aircraft location (x, y). Each ray is then subdivided into a number of steps x_1, x_2, \ldots, x_N . (Ref 19:9).



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The TMPSA program computes the exposure along each of the 2J + 1 rays through N steps, then selects the ray with the minimum exposure. The aircraft position is moved one step along that ray and the position and exposure are updated. Then the process is repeated (Ref 19:9).

The program makes two tests to keep the aircraft within specified limits. The corridor test ensures that the aircraft does not stray beyond the corridor limits. This is accomplished by eliminating any ray from consideration which would cause the aircraft to make its next step out of the corridor. The second test is the "wedge" test to ensure that the aircraft arrives over the finish point. This is accomplished by determining the maximum lateral distance the aircraft can be from the centerline and still reach the final point assuming flight at maximum speed. For the wedge test, note the geometry in Figure 8. If the aircraft is allowed to travel into the shaded area, it cannot maintain a constant x direction velocity and reach XF on schedule. Therefore, any ray causing the aircraft to step out of the wedge is eliminated.

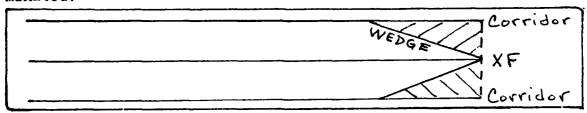


Fig. 8. TMPSA Terminal Wedge Geometry

Automatic Input

Two cases were considered for perfect knowledge. In the

first case, the awareness radius is set to 25 kilometers, the maximum number of steps the program can hold. This was done so that each step would equal one kilometer, thus the output would be comparable with the control model output. In the second case, the awareness radius was set at 100 kilometers, thus allowing the aircraft the capability of selecting its path based on knowledge of all threats in this scenario. This results in a four kilometer step size. The maximum speed allowed in both cases was 390 knots (722 km/hr) (Ref 18). The number of rays considered was eleven. This is the maximum capability of the TMPSA program.

Automatic Run Results

The run results for the two cases of the automatic mission are shown in Table 3.

TABLE 3
Automatic Model: Total Exposure

Mobile x	Threat	Location y	Case l 1 KM Steps	Case 2 4 KM Steps	<u>d</u>
56.0		27.5	42.16	67.31	25.15
56.0		30.0	48.17	78.29	30.12
56.0		32.5	50.01	62.27	12.26
56.0		35.0	50.36	62.67	11.91
56.0		37.5	53.98	69.20	15.22
56.0		40.0	40.41	64.78	24.37
56.0		42.5	36.44	60.38	23.94

Note that the results for the two cases described above follow a similar trend. A paired t-test indicates that there is a

statistical difference between the two results at the 95 percent confidence level. The column labeled d is the difference between the second case and the first. Let D = $\frac{1}{n}\sum_{i=1}^{n}d_{i}$, and compute s_{d}^{2} , the sample variance.

$$s_{d}^{2} = \sum_{i=1}^{n} (d_{i})^{2} - (\sum_{i=1}^{n} d_{i})^{2}$$

where, n = the number of d_i 's. For the data in Table 4, we obtain: D = 20.42, $s_d^2 = 51.75$, n = 7

The null hypothesis is that the mean of the deviations (d) is zero. The test statistic is:

$$t_o = D / \sqrt{s_d^2/n} = 7.510$$

The tabulated t for a two-tailed test with 95 percent confidence and six (ie. n - 1) degrees of freedom is, t = 2.365 (Ref 9:477). The rejection criteria for this test is, reject if $t_0 > t$ (Ref 9:267,269). Clearly, the hypothesis is rejected. Since step size causes a significant difference in exposure, the step size will be held constant throughout the experiment, if possible.

The above discussion concludes this section on perfect knowledge of the threat locations. In the next section, exposure is examined when perfect knowledge is not available, for example, due to sensor limitations, but an automatic flight path generator is used.

Uncertainty of Location

Due to the complexity of the geometry between the threat

sites, their templates, and the aircraft, the problem of uncertainty is resolved using simulation. The measurement of range and azimuth is simulated by adding zero-mean, normally distributed noise terms to the actual range and azimuth. The procedure for accomplishing this is described below. The algorithm is included as part of the TMPSA program in Appendix B.

The algorithm used by the program to accomplish the randomization of the site location with respect to the aircraft is based on changing the statistical scale. The algorithm begins with selection of a series of random numbers, R_{j} , from a uniform distribution between zero and one. A unit variance is generated by:

$$N_{A} = \sum_{j=1}^{12} R_{j} - 6$$

The result approximates a sample drawn from a truncated normal (0,1) distribution (Ref 12:90-95). Multiplying the sample by the standard deviation yields a noise term. In effect, the multiplication spreads the normal distribution. The noisy azimuth measurement becomes:

where, α_k' is the noisy azimuth measurement to the kth site, α_k' is the actual azimuth measurement to the kth site, and is the standard deviation of the azimuth measurement (Ref 20:6). The addition results in the normal distribution being moved from zero to the actual azimuth.

The noisy range measurement is determined by:

$$r_k' = r_k + r_k \sigma_r N_R$$

where, r_k' is the noisy range to the k^{th} site, r_k is the actual range to the k^{th} site, σ_r is the standard deviation of the normalized range measurement and N_R is a normally distributed random variable with unit variance generated the same way as N_A (Ref 20:7).

The location from the aircraft for site k is determined by the relationships above, ie.

$$x'_k = x_n + r'_k \cos \alpha'_k$$

 $y'_k = y_n + r'_k \sin \alpha'_k$

where (x_n, y_n) is the aircraft location. This computation results in an estimate of the site location for one measurement.

Clearly, if a large number of measurements were taken and averaged, the limiting condition would be to have perfect knowledge of threat locations. In actual fact, time is available to take only a finite number of measurements. The question then is how many measurements can a processor handle in the time it takes to make each step. The procedure to estimate this was to define the processing time per measurement, and the distance per step and speed in the x direction. Dividing the distance by the speed and then dividing this result by the processing rate will yield the number of measurements per step.

The last problem to be solved is the determination of the number of simulation runs required. Since there is uncertainty as to the standard deviation of the exposure for each run and the feasible range of the possible standard deviation, a formulation for the number of runs is:

$$n = \frac{(Z_{d/2})^2 \sigma^2}{(\sigma/b)^2}$$

where, n is the number of samples, $Z_{\bullet /2}$ is the risk to be taken (ie. $Z_{\bullet /2}$ is the two-tailed standard normal statistic for the level selected), and $\frac{+}{b}$ is the interval about the mean in which the sample value will lie between 100 (1- ϕ) percent of the time (Ref 16:188).

This ends the discussion of the formulation for the case with uncertain threat location. In the next section, actual parameters to be used are presented.

Uncertain Location Inputs

The first inputs necessary for the simulation are the standard deviation of the range and azimuth (σ_{r} and σ_{s}). In an earlier study using TMPSA it was determined that exposure increased greatly when $\sigma_{r} \geqslant 0.15$ and $\sigma_{s} \geqslant 5$ degrees (Ref 20:13). These critical values will be used. An argument will be offered later that sensitivity analysis of these parameters is unnecessary.

The next input parameter needed is the number of measurements per stcp. Assume one second is required for each measurement. The aircraft in this model is traveling at 648 kilometers per hour at one kilometer per step in the x direction. Therefore, the processor is averaging more than five measurements per step. Obviously, the number of

measurements per time is sensitive. The time of one second per measureme t is slow for modern processors, and is therefore, a conservative assumption (Ref 8:E-3)(Ref 15:9). It is further noted that for ten or more measurements the exposure rate converges rapidly to a low value (Ref 20:13-16). It is by this same set of circumstances that it was shown that the exposure becomes fairly constant for a suitably large number of measurements (Ref 20:13-16). Thus sensitivity analysis of σ_r and σ_{α} is unnecessary.

The formula for determing the number of runs indicates that 16 runs would be needed (rounded up from 15.37) to have a 95 percent probability that the average exposure over those runs will lie within the interval $\mu + \frac{\sigma}{h}$ where μ is the true average exposure, and b = 2. Since this is a very broad range, these initial runs are used to determine a sample standard deviation (s). This statistic and the t statistic are then used to derive another number of runs. In this case, the number of runs is determined by:

$$n = \frac{t^2 s^2}{d^2}$$

where, t is the tabulated t value for the desired confidence level (0), and the degrees of freedom of the sample runs, ${\bf s}^2$ is the estimate of the variance obtained from the sample runs and d is the half-width of the confidence interval specified (Ref 16:189).

Run Results with Uncertainty

The results of the first sixteen runs are shown in

Table 4. The cumulative exposure and standard deviation are indicated.

TABLE 4
Average Exposure: 16 Runs

	Total	Cumulative		
Run	Exposure	Average Exposure	Standard Deviation	
1	55.07	55.07		
2	55.61	55.34	. 382	
3	60.38	57.02	2.922	
4	56.94	57.00	2.386	
5	54.62	56.52	2.325	
6	54.08	56.12	2.306	
7	55.48	56.02	2.119	
8	58.57	56.34	2.158	
9	54.89	56.18	2.076	
10	57.51	56.32	1.002	
11	58.32	56.50	1.993	
12	55.36	56.40	1.928	
13	54.80	56.28	1.899	
14	55.72	56.24	1.831	
15	55.46	56.19	1.776	
16	58.18	56.31	1.786	

Clearly, the average exposure is converging to a value around 56. To compute the average exposure within \pm 0.5, however, with a 90 percent confidence requires n runs.

$$n = \frac{t^2 s^2}{d^2}$$

The t statistic is the t value for 15 degrees of freedom and α = 0.1. This equals 1.753 (Ref 9:477). From Table 4,

s equals 1.786. And d is the interval of ± 0.5 . Therefore,

$$n = \frac{(1.753)^2 (1.786)^2}{(.5)^2} = 39.2$$

By making 40 runs, it can be stated with a 90 percent confidence that the actual average exposure lies between plus and minus 0.5 of the computed value. Table 5 shows the average results for each case of 40 runs. The computer output is in Appendix C.

TABLE 5
Uncertain Model: Total Exposure Table

Mobile Thre	at Location v	Average Exposure	Standard Deviation
56.0	27.5	44.0	3.7
56.0	30.0	52.5	4.2
56.0	32.5	51.8	3.5
56.0	35.0	52.3	3.7
56.0	37.5	56.0	3.7
56.0	40.0	44.4	3.4
56.0	42.5	38.8	3.7

This completes the discussion and data collection for the automatic model. In the next chapter, some of the complexities of the human and machine interactions used to model the manual mission are introduced into the model.

Chapter IV The Manual Mission

The manual mission involves more specific modeling of the human factor in the experiment. First, an overview of human reaction time theory is presented. Next, current aircrew interactions are described, and the complexities of these interactions are discussed. From this a simplified aircrew reaction model is developed. Finally, this model is used to modify the TMPSA input data. The revised input is run using TMPSA, as shown in Appendix B, and the output derived will parallel the results of Chapters two and three.

Overview of Human Reaction Time Theory

There are currently two principal theories to describe human response times, the additive component theory and the variable criterion theory (Ref 6:431).

The older, additive component theory traces its origins to the experiments of Donders during the mid 1800's (Ref 14:2). One of the most recent applications of this theory is the method of convolution (Ref 10:3-4). Using this method, Kohfeld and Nullmeyer identified three component stages of response time; sensory-detection, stimulus identification, and response execution (Ref 10:11). The main contention of this theory is that these stages occur consecutively rather than simultaneously. Thus, the components are additive in nature.

The variable criterion theory was first proposed by

Grice in 1968 (Ref 14:4). In its present form, the theory postulates that "response evocation will result when the combined strength of the (sensory) processes satisfies a decision criterion." (Ref 6:431). For a simple reaction time experiment, sensory growth occurs with respect to time in a negative exponential fashion until it reaches the criterion. The criterion is described as a normal distributed random variable. At the point where excitatory strength reaches this momentary criterion level, a response occurs (Ref 14:8). A basic premise of this theory is that reaction time need not and should not be broken down into component parts to be described. This theory holds that the components vary from completely overlapping to no overlapping and therefore cannot be summed accurately.

Irrespective of the theory used, there are a number of factors which affect reaction time. The most important of these factors are noted as follows:

- a. The sense used.
- b. The characteristics of the signal.
- c. The complexity of the signal.
- d. The signal rate.
- e. Whether or not anticipatory information is provided.
- f. The response characteristics of the body member used. (Ref 11:228).

These factors will be used to build the simple aircrew response model.

First, however, a description of a current bomber

aircrew interaction is presented. This relies heavily on the author's seven years of experience as a E-52 Electronic Warfare Officer (EWO).

Description of Current Bomber Crew Procedure

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The simplest engagement is the one-on-one situation. In this case, the engagement begins for the aircrew when any crewmember detects a threat. Under normal circumstances, the EWO will detect a radar directed threat first on ECM receivers. Information about the relative location of the threat is passed to the rest of the crew. If the threat is immediate, the pilot will attempt to maneuver the aircraft to avoid being hit while the EWO applies electronic self-protection measures. If more time is available, the navigator may become involved. Based on the relative location transmitted by the EWO, the navigator will attempt to identify the threat from among the known enemy threats in the area. Then he will direct the pilot to follow a course which enables the bomber to avoid the lethal envelope of the threat.

Although the procedure is fairly simple to describe, the possible interactions result in a virtually infinite number of possible aircraft maneuvers. Variables include the time it takes the EWO to detect and identify the threat, the time it takes him to transmit this information to the rest of the crew, the time it takes for the pilot to "detect" the EWO's message, and finally, the specific maneuver the pilot chooses to make. These variables do not even consider the

time required to react if the navigator is involved in the process. In the next section, a much simpler model is developed. Its purpose is to include the man in the loop of the TMPSA program.

Simplified Crew Procedure

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To simplify the crew reaction model, the following scenario is used. Aircraft sensors receive the threat radar emissions. This information is processed using an algorithm like TMPSA. The output is then passed in the form of a digital readout to the pilot. The pilot is prompted to respond by observing the heading readout. He responds by making an input via the aircraft controls. Then, the aircraft mechanical response follows.

As noted earlier, the hardware sensing-processing time is assumed to be one second (see Chapter three). The crewmember response time to a visual prompt is on the order of 0.2 second (Ref 11:229) (Ref 7:305,307). However, for tasks such as the simple response outlined above, the maximum rate of response is two to three per second (Ref 11:231). These are mean times drawn from the probability distributions for reactions as described earlier in this chapter. They are sufficient for the purpose of this study. Assuming a conservative two responses per second reaction rate, the pilot can keep pace with the sensor-processor, except there will be a half second lag.

Another lag occurs when the mechanical input of the

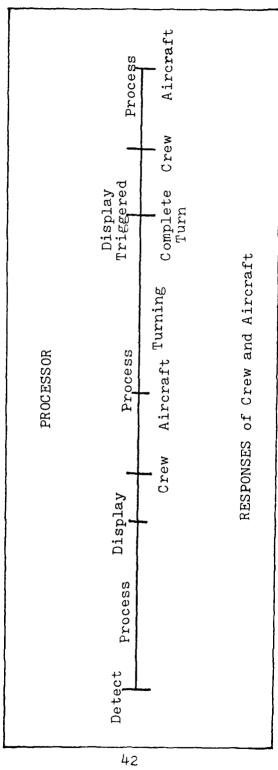
pilot is translated into aircraft movement. The first part is the aircraft response time. The second part is the time it takes the aircraft to complete the turn. The mechanical response time is considered a constant for like aircraft while the turn time is based on the aircraft turn rate and angle through which the aircraft must turn. Continuing use of the B-52 data, the mechanical response time is on the order of a half second (Ref 4). The time required to complete a turn is a function of the angle of bank, the altitude, and the speed of the aircraft.

Another way to view the simplified crew procedure is to consider actions along a time line (see Figure 9). The action starts with the sensor picking up a threat signal. After one second, a heading readout is displayed for the pilot. The pilot reaction time is a half second. Finally, it takes a half second for the mechanical translation of the pilot's input to start the aircraft turning and a certain amount of time to complete the required turn.

If the readout changes during this time, it is assumed the pilot would be too preoccupied to notice. In that case, the next cycle begins when the pilot rolls out on the first indicated heading. Thus, the delay is actually only the time required for the pilot and aircraft to react to a new heading. During that reaction, another heading is being processed. Figure 9 shows these simultaneous actions.

Manual Model Inputs

To make the results of the manual model comparable with



Penetrator Response Timeline Fig. 9.

the control model, it is necessary to minimize the changes to the input data. The method used is to determine how quickly the man-machine system can react. Assuming that the awareness radius (R in the model) remains 25 km, the size of each step (DX), the number of steps in the awareness radius (NSEG), and the number of measurements for each reaction (NM) can be computed.

Figure 9 shows how the event chain occurs. Figure 10 below portrays this event chain as a cycle.

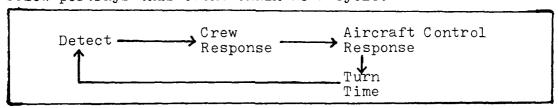


Fig. 10. Penetrator Response Cycle

In the previous section, the crew response time was assumed to be a half second, and aircraft mechanical response time was stated to be a half second. The turn time must be determined.

As noted earlier, the airspeed, altitude, and bank angle of the aircraft determine the level rate of turn. In this model, the altitude and airspeed are constant. The altitude is about 1,000 ft, and the airspeed ranges from 350 knots to 390 knots. The maximum turn \(\psi\) for this scenario is computed by:

$$\frac{\text{minimum speed}}{\text{maximum speed}} = \cos \ensuremath{\,\boldsymbol{\gamma}}$$

Using the airspeeds above, ψ = 26.2 degrees.

The only variable left is the angle of bank. At low

altitude, the normal angle of bank used by the B-52 is 12 degrees to 15 degrees. However, a turn of up to 30 degrees is possible, but hazardous (Ref 18). The fastest rate of turn for 30 degrees angle of bank is at the slower airspeed (ie. 350 knots). The maximum turn rate is 1.8 degrees per second at 350 knots, 1,000 ft altitude, and 30 degrees angle of tank (Ref 18). Therefore, to complete the 26.2 degree turn, the shortest time is 14.5 seconds. This represents the minimum time for the maximum turn. It is possible with the TMPSA program for the aircraft to turn from a maximum heading of + ψ to a maximum heading of - ψ , thus covering 2 γ degrees. Since this represents a rather violent maneuver, it is assumed that extraordinary measures such as this would be accomplished by exceeding the assumed parameters of the scenario. For this reason the 14.5 second turn time will be used. Also, any smaller turn can be made in this time using a shallower, and thus safer angle of bank. Therefore, this is used as the turn time constant.

The total response cycle is a half second for crew response, half a second for aircraft mechanical response, and 14.5 seconds for completing the maneuver. The sum of these is 15.5 seconds.

The size of each step is a function of the airspeed (in the x direction), and the total response time. That is:

$$\frac{648 \text{ km/hr}}{3,600 \text{ sec/hr}} \cdot (15.5 \frac{\text{sec}}{\text{response}}) = 2.79 \frac{\text{km}}{\text{response}}$$
In the TMPSA program this is DX.

Since the awareness radius (R) is kept at 25 kilometers, the number of steps (NSEG) in the awareness radius is:

 $DX = \frac{R}{NSEG}$ or 2.79 = $\frac{25}{NSEG}$ and NSEG = 9 by choosing the closest integer.

The last piece of information needed is the number of measurements per step (NM). This is the number of measurements which can be processed during each response cycle.

NM is computed by:

- (15.5 seconds/response)(1.00 measurments/second) =
- 15.5 measurements/response (round down to the nearest integer).

To summarize, the inputs for the manual mission are:

$$R = 25$$

$$NSEG = 9$$

$$NM = 15$$

Run Results

To be comparable with the results in Chapters two and three, the TMPSA program was run through forty iterations for each position of the mobile threat. The output is in Appendix C. The results are summarized in Table 6.

TABLE 6

Manual Mission: Total Exposure Table

Mobile Thre	at Location	Average Exposure	Standard Deviation
x	у		
56.0	27.5	34.6	4.2
56.0	30.0	46.8	4.4
56.0	32.5	44.3	3.2
56.0	35.0	44.6	3.6
56.0	37.5	48.7	3.8
56.0	40.0	39.0	3.5
56.0	42.5	30.5	3.4

Chapter V Results and Analysis

Results

In this chapter, the results of the missions described in Chapters two through four are tabulated and analyzed.

Table 7 summarizes the results for all four mission types.

TABLE 7
Summary Results Table

			Expo	sure			
<u>Index</u>	Threat	<u>Control</u>	Automatic	<u>Unce</u>	ertain	<u>Ma</u>	nual
<u>i</u>	y-coordinate						
1 2 3 4 5 6	27.5 30.0 32.5 35.0 37.5 40.0	67.87 80.00 80.43 79.48 80.43 80.00 67.87	42.16 48.17 50.01 50.36 53.98 40.41 36.44	44.0 52.5 51.8 52.0 44.4 38.6	3.7 4.5 3.7 3.7 3.7	34.6 46.8 44.3 44.6 48.7 39.0	4.4.26.85.4

The general structure of these results is intuitively appealing with the exception of the manual mission results. Figure 11 shows what one could intuitively expect to occur. As the level of feedback increases from no information on the left to much accurate information on the right, exposure decreases.

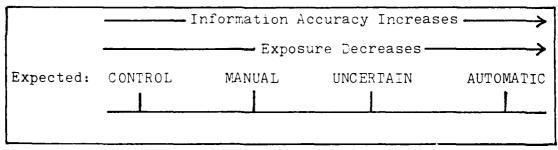


Fig. 11. Accuracy vs. Exposure Line 47

In the following sections, each step used to develop the final results is explained. How the changes in input affected the exposure is interpreted. Finally, comments are made to explain how much of the exposure change is attributable to the input change.

From this discussion, the reason for the counterintuitive mission results is explained. With the reason for this anomoly explained, the manual mission results are adjusted, then an analysis of the differences between the manual and control missions is accomplished to find the value of reactive maneuvers.

Control to Automatic

The only change in the input data between the control mission and automatic mission is the maximum speed. The maximum speed used for the control mission is 648 kilometers per hour. Since the minimum speed is also 648 kilometers per hour, the penetrator can only travel down the center path. This simulates no feedback to the aircrew of the status of enemy defenses. The maximum speed for the automatic mission is 722 kilometers per hour. Taken with the other input data, this situation simulates accurate and timely information reaching the crew. The crew is thus able to choose the flight path using the TMPSA algorithm to find a flight path which reduces total exposure.

The total exposure declined in absolute and relative terms as shown in Table 8.

TABLE 8
Control/Automatic Mission Differences

<u>i</u>		osure <u>Automatic</u>	d _{li}	Δ_{1i}
1234567	67.87 80.00 80.43 79.48 80.43 80.00 67.87	42.16 48.17 50.01 50.36 53.98 40.41 36.44	25.71 31.83 30.42 29.12 26.45 39.59 31.43	0.38 0.40 0.38 0.37 0.33 0.49

In Table 8, d_{li} = Control_i - Automatic_i

$$\Delta_{li} = d_{li} \div Control_i$$

The average relative exposure difference (Δ_1) is:

$$\Delta_1 = \frac{1}{n} \sum_{i=1}^{n} \Delta_{i}$$
 for all values of i

For the data in Table 8, Δ_1 = 0.40. The standard deviation (σ_1) is 0.06. The variance (S_d^2) is computed as detailed in Chapter three and σ_1 is the square root of S_d^2 .

It would appear from the above analysis that the effect of accurate knowledge of threat locations decreases exposure by about 40 percent. The relatively wide dispersion of these results seems to indicate that the scenario itself is a factor.

Automatic to Uncertain

Two changes were made in the input data in going from the automatic mission to the uncertain mission. The first was to change the standard deviations of the range and azimuth measurements from zero to 0.15 times the actual range and 5 degrees respectively. The effect is to introduce

uncertainty into the location of the threat sites. The second change between the automatic and uncertain missions was to average a number of range and azimuth measurements of a site before a location for that site is estimated. This has the effect of reducing the measurement uncertainty by the averaging process.

The net effect of these two changes is to cause the TMPSA flight path algorithm to make choices based on less than perfect information. Table 9 shows the increase in exposure that resulted from this uncertainty. In Table 9,

$$\Delta_{2i} = d_{2i} \div Control_{i}$$

For the uncertain mission, the mean values are used.

TABLE 9 Automatic/Uncertain Mission Differences

<u>i</u>	Uncertain	Automatic	d _{2i}	Δ_{2i}
1 2 3 4 5 6 7	44.0 52.5 51.8 52.3 56.0 44.4 38.8	42.2 48.2 50.0 50.4 54.0 40.4 36.4	1.8 4.3 1.8 1.9 2.0 4.0 2.4	.026 .054 .022 .024 .025 .050
	$\Delta_2 = .03$	4	= .013	

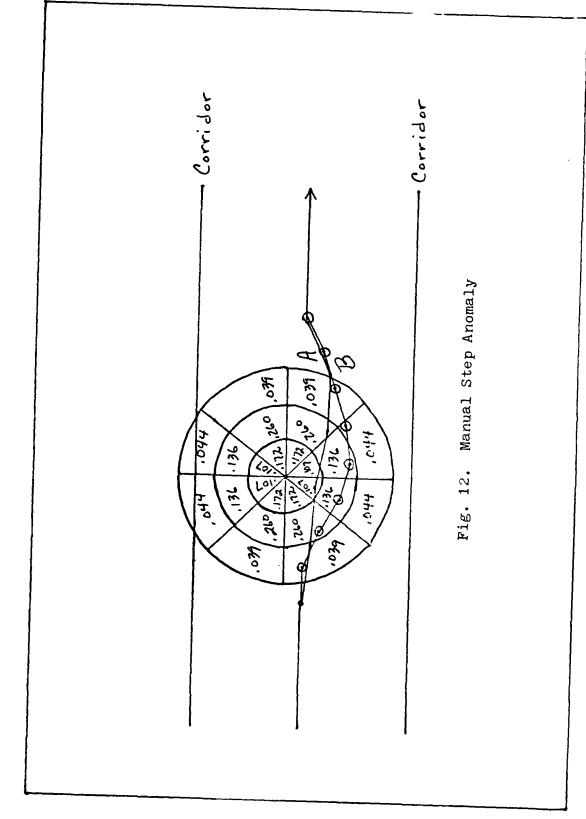
The above analysis indicates that there is about a three and a half percent decrease in exposure that can be directly related to the accuracy of measurement of threat location. Again, a wide variance suggests the scenario is a significant factor.

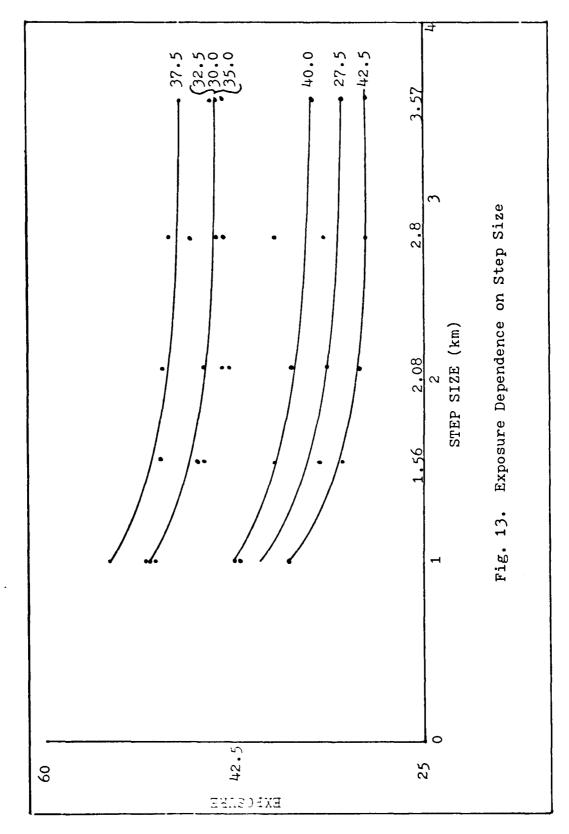
Uncertain to Manual

Two changes were made in the input data in going from the uncertain mission to the manual mission. The number of measurements of each site to establish a location was increased. The result of this change is to decrease the uncertainty in locating threats. The second input change was to decrease the number of steps in the awareness radius. Since the awareness radius was not changed, the effect is to take larger steps. As noted in Chapter four, the step size changed from one kilometer per step to about 2.8 kilometers per step.

This was not considered significant until the results of all missions were compared in Table 7. Careful consideration of the model dynamics offers a logical explanation. First, the locations of the threats are more accurately known because more measurements are being taken. Second, the increased step size is significant because it is larger than the depth of site template segments. This allows the airplane to step completely over the worst probability of kill segments without incurring a commensurate exposure penalty. Figure 12 shows how this can occur. Note that path A is a shorter path and still keeps out of the segments with large probabilities of kill.

A correction factor of 6.3, derived as shown below, was added to all of the manual mission values to establish a





revised manual mission exposure. This revised manual mission output is then compared with the uncertain output in the next section.

The correction factor was derived by showing the dependence of exposure to step size. To do this 40 simulations were run for each value of the mobile threat site (7) and for five step sizes. This resulted in 35 data points. The raw output is in Appendix C. Figure 13 summarizes these results. Smooth curves are drawn in for each of the seven scenarios. Although a number of data points are off the curves, none are off by more than one standard deviation. Only one curve does not connect the end points. The scenario with the eleventh threat y-coordinate at 27.5 is the only one where the exposure at a step size of one kilometer is not on the curve. This curve is based on the shape of the other six curves; and the point at one kilometer is, as noted, within one standard deviation of the raw data mean (see Appendix C).

For each scenario, the exposure is read from the curve at the 2.8 kilometer step size and subtracted from the 1.0 kilometer step size exposure value. The values are shown in Table 10. These differences (D_i) are then averaged. This average of 6.3 is the correction factor. The standard deviation is 0.5. The revised manual mission values are shown in Table 11 in the next section.

TABLE 10

Exposure Difference Due to Step Size

i	1.0 Value	2.8 Value	Di
1 2 3 4 5 6 7	40.0 51.0 51.0 51.0 54.0 43.0 37.0	33.0 45.0 45.0 45.0 48.0 37.0 30.0	7.0 6.0 6.0 6.0 6.0

Uncertain to Revised Manual

K i

With the step size difference accounted for, the main difference between the uncertain and manual missions is the number of measurements taken for each step. Since more measurements are taken for the manual mission, the results of this comparison should be that the revised manual exposure is lower than the uncertain. Table 11 shows the effect of increased information of threat locations.

$$d_{3i}$$
 = Uncertain_i - Revised Manual_i
 Δ_{3i} = d_{2i} ÷ Control_i

Mean values are used for the uncertain and revised manual data.

TABLE 11
Uncertain/Revised Manual Mission Differences

<u>i</u>	Expo <u>Uncertain</u>	sure Revised Manual	^d 3i	Δ _{3i}
1 2 3 4 5 6 7	44.0 52.5 51.8 52.3 56.0 44.4 38.8	40.9 53.1 50.6 50.9 55.0 45.3 36.8	3.1 -0.6 1.2 1.4 1.0 -0.9 2.0	.046 007 .015 .018 .012 011

$$\Delta_3 = .014 \quad \sigma_3 = .020$$

The above results indicate that increasing the accuracy of threat location information in this scenario does not significantly change exposure between the uncertain and revised manual missions. Thus, one of the three factors affecting exposure is eliminated from consideration. That is, sensors accurate to five degrees in azimuth and 15 percent in range are adequate if at least one measurement per second can be taken.

The second factor affecting exposure is the absolute accuracy of the sensors. If the accuracy of the sensors is perfect, as opposed to five degrees in azimuth and 15 percent in range uncertainty, it was shown that the exposure is reduced by about three and a half percent.

The remainder of the reduction in the total exposure is related to the ability of the aircraft to maneuver. This is the third factor affecting exposure. In the next section, the value of maneuverability is derived from a comparison of the control and revised manual missions.

Control to Revised Manual

The comparison of the control mission with the revised manual mission yields the value of reactive maneuvers. In the revised manual mission, a degree of knowledge of the defenses allows the penetrator to make decisions required to maneuver the aircraft. The two differences between the control and the revised manual mission are maneuverability and threat location knowledge. Table 12 illustrates this comparison

where:

$$d_{\mu_i} = Control_i - Revised Manual_i$$

$$\Delta_{\mu_i} = d_{\mu_i} \div Control_i$$

TABLE 12
Control/Revised Manual Mission Differences

<u>i</u>	Expos <u>Control</u>	ure <u>Revised Manual</u>	d _{4i}	d _{4i}
1 2 3 4 5 6 7	67.87 80.00 80.43 79.48 80.43 80.00 67.87	40.9 53.1 50.6 50.9 55.0 45.3 36.8	27.0 26.9 29.8 28.6 25.4 34.7 31.1	.40 .34 .37 .36 .32 .43
	Λ. =	38 🗻 = .05		

$$\Delta_{4} = .38 \qquad \omega_{4} = .05$$

These results indicate that the survivability increases by about 38 percent when there is adequate knowledge of threat locations and the aircraft is allowed to maneuver.

The next chapter discusses the analysis results stated above and judgements on the results are rendered.

Chapter VI Conclusions and Recommendations

The conclusions to be drawn from the above analysis must take into consideration a number of limitations discovered in the course of this research. A critique of TMPSA is followed by conclusions and some recommendations for future work in this area.

Critique of TMPSA

The goal of TMPSA is to determine how aircraft sensor measurement accuracy is related to aircraft survivability against surface-to-air weapons. The goal of this research was to use TMPSA to derive a value for reactive maneuvers. Following is a critique of the main weaknesses and strengths of TMPSA learned in its use.

The TMPSA program and supporting documentation have three categories of shortcomings. The first shortcoming is that a major claim of the TMPSA report is not factual. The second and third shortcomings are groups involving modeling shortcomings and user pitfalls. Each of these problems is presented and discussed below.

In the TMPSA report, the author claims that TMPSA uses an "algorithm to find the safest route through an arbitrary threat distribution." (Ref 20:1). In fact, while the algorithm attempts to maximize survival by minimizing exposure, it does not actually optimize. Consider how the flight path

is generated. Only a maximum of 11 rays or possible paths are examined. This immediately eliminates ar infinite number of possible alternatives. The subsequent movement of the aircraft is determined by stepping in the direction which offers the lowest exposure to the awareness limit in that direction, but does not consider alternate paths which may have lower exposure.

Figure 14 shows a small sample of the possible flight paths. The solid lines are the paths considered by the TMPSA algorithm. The dotted lines are alternate possible paths. The numbers at each node are the exposure for that node and the letters identify the made

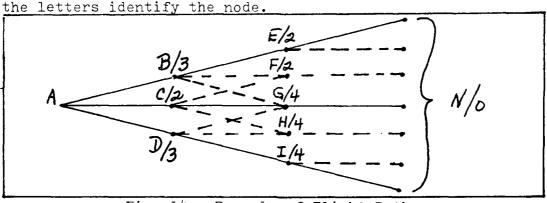


Fig. 14. Example of Flight Paths

In this example, the choices of flight paths available to TMPSA when the aircraft is at A are ABE, ACG, and ADI. The total exposure over the length of each of these flight paths is five, six, and seven respectively. The program would step to B. From B the available paths are BEN, BFN, and ECN where N represents all seven of the following nodes which are assumed to cause no exposure. They also could be viewed as all N points having equal exposure. The exposure for the paths

from B are five, five, and seven respectively. Observe, however, that if path ABEN or ABFN (total exposure equals five) are compared with ACFN, the ACFN path is safer with a total exposure of only four. Thus, TMPSA does not choose the optimum path. Although the program could be revised to accomplish the above task, the number of possible paths to be summed grows exponentially with the number of rays and the number of steps in the awareness radius. Therefore, this is not a practical solution for the usual situation being modeled.

The model has four additional shortcomings which have been grouped together as modeling shortcomings. Two of these shortcomings are related. They are lack of consideration of terrain and lack of consideration of cultural features. The factors of use of terrain and avoidance of cultural features are used extensively by operations planners to determine safe penetration routes. Terrain is used by the penetrator as cover. The penetrator will seek a flight path which causes terrain to be between the aircraft and the enemy fire control radar. Cultural features such as roads and cities are avoided by the penetrator as a way of avoiding contact with the enemy. Although not critical to the overall model, these two shortcomings reduce the credibility of the output.

The third shortcoming in this group involves the construction of the threat site templates. The only documentation of the values used in the construction of the site template used in the TMPSA study report is a parenthetical phrase

that the probabilities of kill and lethal radius were those used in a large scale simulation study (Ref 20:8). How the site templates are constructed is crucial to the model because they play a key role in determining exposure. To be able to judge the credibility of the cutput, the method used to generate the site templates is essential.

The last shortcoming of this type concerns where the aircraft is allowed to fly in the corridor. The aircraft has a constant velocity component along the attack axis. This is not realistic. Although the penetrator can maneuver sharply and sustain higher airspeeds to get back on time, in reality, in TMPSA it is incapable of such maneuvers because of the constant velocity component. An example illustrates this problem. Consider the problem in Figure 15.

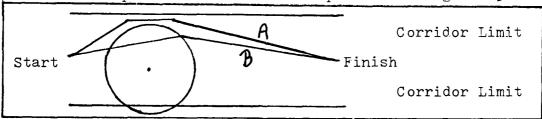


Fig. 15. Example of Velocity Constraint

In this figure, path A is longer, but at a higher sustained velocity the Finish point can be reached at the same time as the path B penetrator. The main difference is that the path A axis velocity component is allowed to vary. This is another crucial element in the total exposure computation which raises questions about the credibility of the model.

The last group of shortcomings are user pitfalls which the analyst must keep in mind when developing the input for the model. First, the model does not explicitly include physical time and space limitations for crew reaction and aircraft motion. Turns are instantaneous at each step.

Second, a constant altitude is assumed for the penetrator. Although the program can be revised relatively easily to accomplish three dimensional motion for the penetrator, the threat site templates would also require development in three dimensions and the lack of terrain consideration would grow in importance.

Third, the model literature does not explicitly mention threats outside of the corridor limits. In this research experiment with the model, it was found that the exposure declined when external threats were introduced. The reason deduced for this anomaly was that without the exterior threats, the penetrator would track to the edge of the corridor. Because there were no threats outside the corridor, those rays would have the lowest total exposure and the penetrator would thus remain along the corridor edge.

The last pitfall concerns the data input used by the authors of TMPSA. Although they studied the effects on exposure of accuracy of sensors by range and azimuth <u>separately</u>, they produced no comments on the combined effects on exposure of range and azimuth inaccuracies together.

Having mentioned nine specific criticisms of TMPSA, let us now turn to the positive aspects of TMPSA. The model has three good points. Within the context of the purpose of the model, these points are the concept, model flexibility, and

intuitively appealing results.

The basic concept of the model is to investigate all of the various possible paths within the awareness radius of the penetrator, then choose the safest path. Due primarily to computer hardware and time limitations, it is not practical to investigate each and every path. Use of the computer also requires use of non-continuous probability of kill distributions for the threat site templates and the stepping of the aircraft flight path. Althouth this is artificial, the problem can be resolved by decreasing the template segment sizes and the step size of the aircraft. Again, computer hardware and time limitations restrict the resolution that can be attained. However, these input parameters can be used to control the model to a large degree.

It is through judicious data input that the model derives its flexibility. By correct selection of the input, all of the shortcomings of user pitfalls can be overcome. Also, if a proper analysis is done, the threat template can offer a true representation of a certain type of surface-to-air threat verses a certain type of aircraft. Thus, by carefully planned input, the output can be reasonable.

In using this program, it was found that the output agreed with this author's intuition. The paths selected matched closely those an operational planner might select under the same circumstances. This was achieved only when the input data was true to the scenario. In every instance where the results did not agree with intuition an error was

found in the input data.

Conclusions

This model does not portray many of the factors involved in a penetration model. Examples of items not modeled are terrain and cultural features. But the purpose of the model is not to attempt to model all the nuances of a penetration. The goal of TMPSA, as stated in the beginning, is to determine how aircraft sensor measurement accuracy is related to aircraft survivability. The purpose of this research is to use this model to examine how maneuverability affects survivability.

The analysis in Chapter five shows that the effects of maneuverability and accuracy of threat location knowledge are intertwined. From the above discussion and analysis in Chapter five, it is concluded that the TMPSA model is adequate for studying the effects of sensor accuracy and aircraft maneuverability on exposure if the input is properly prepared. However, for the reasons listed below, the model output cannot be used to establish ratio relationships among the various input variables, specifically the accuracy and maneuverability input variables.

There are two reasons for the above assertion. First, the aircraft travels through the threat array by large steps and the threats are represented as segmented probabilities of kill where the segments are relatively large. These discontinuities alone are enough to destroy the ratio relations.

Second, the model does not optimize survival. For this reason, there is no point from which to measure the lowest exposure. In fact, some runs of the program yield lower exposure with inaccurate measurements than runs which have no measurement inaccuracies!

With these caveats in mind, it can be said that the results are comparable on an order-of-magnitude scale. That is, if one set of runs results in twice the exposure of another set of runs, it is safe to say that the second set of conditions will yield a safer penetration profile than the first set of conditions. It would probably be erroneous, however, to assume the second conditions are twice as safe as the first.

From the above statement, it is inferred that the goal of relating sensor measurement accuracy and maneuverability to aircraft survivability is accomplished in a macroscopic sense. However, a clear mathematical relationship between sensor measurement accuracy and aircraft maneuverability, and aircraft survivability derived from this model is not supportable.

Having concluded that the results above are insufficient, what avenues are available to improve this rough procedure?

Recommendations

One recommendation was noted above; that is to increase resolution of the model by decreasing the step size of the

aircraft and decreasing the size of the threat template segments. Another recommendation is to expand to three dimensions. Each of these procedures increases the degree of reality being modeled. They also increase the computer running time significantly.

Two other possible approaches to this problem of quantifying reactive maneuvers are suggested. One is to try to follow every possible discrete flight path from the awareness radius limit back to the present aircraft position, one step at a time. Using a dynamic programming algorithm, all but the smallest exposure branch for each node is eliminated until the present position is reached. Then a step is taken on the last branch. Returning to Figure 14 and the example above, the technique would work as follows. In this example, the awareness radius is two and the step size is one. To node B from nodes E, F, and G, the smallest exposure is two from E and F, so G is eliminated. To node C from nodes F, G, and H, all but node F are eliminated. To node D, all three nodes G, H, and I remain possible (all equal four). To node A, the total exposure from node B is five (two plus three at B). node A from C, the total exposure is four. And, from node D, the total exposure at node A is seven. Eliminating all but the smallest, yields the optimum flight path.

The second approach would be to rewrite the TMPSA program in terms of a computer simulation language. With TMPSA written in a simulation language, a large number of runs could be efficiently run. The mean total exposure can be determined with a tighter distribution about the mean when

more runs are made. The result would be that the analyst could have more confidence in evaluating the interaction between maneuverability and sensor accuracy on exposure.

The last recommendation is to make a change in the input to allow study of different degrees of maneuverability. In the context of TMPSA, increasing the maximum speed results in increased maneuverability. In this regard, it would be better to include some physical limits on the ability of the aircraft to turn. Even the best aircraft in the world cannot turn on a point.

To this author, the most promising direction for future work in this area is to combine the last suggestion above with the dynamic programming recommendation. It would be relatively simple to revise TMPSA with these two changes. Then a three dimensional matrix of data points relating scenario, maneuverability, and sensor accuracy could be built and the interactions of these factors analyzed. From this, a true independent value for reactive maneuvers might be developed.

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$\label{eq:Appendix A} \mbox{FORTRAN Code Listing of the TMPSA Frogram}$

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                 AC(L) = CISTO/NY
                 &F(L) =/ LOH 9/:/4
                YS(L) = \sqrt{2}(L)^{2}S^{2}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^{2}S^{3}(L)^
                WFITE(6,74 ) XS(1),47(_)
350 0047 FUE
=SET-HP ( (F FC PAYS
                 DO - J=1, NT (YS
                       PK J(J)='.
                       rv [fr= •
                        T == J-1
                        PO(J(J) = -257 + 7747 \times 2021
```

```
#SET-UP LOOP FOI PAY SEGMENTS
        DO " JET NEED
     PSITI1= .
     PSIT - 7= .
          YP: =X! +! 177
          VEE=YE+T (04:70)(257)(1))
*SET -UP LOGE FO CITIES
          00 8 1 L=1,1517F
     IFL^G=1
            7 1 = 5 / K ( _ ) - X = 6
            1 7=5 YY (E) -YE C
            T = ST T ( T1 + T1 + F 7 + F 7 +
     IF(0.67. . ) 60 TO 2"
     ALPH' = .
     90 TC 2
     ALPH: = 1741 3(13,73)
24
     THETO=+35 (751J(J)=3_945)
25,
     TH(THET(.E. .P.) THYT$=?.↑ ?↑→5H/YA
     KY=KYK(L)
     CALL PKTAP.
     의∼ITE (+ • - 111) EK
5011 FC - 4 T (4H , 3HFK=, F43, 3)
     IF (IFLAS, 15, 1) FO 70 251
     PSITT 1=PSI /E++PK
     IFL.G=2
     T1=XS(L)-Y=R
     72=Y'(()-Y 22
     9=S0=T(T1, T1+T2, T2)
     GO TO 27
     PSITTZ=PKI - EZ+PK
265
300
          CONTINE
     IF(J.E0.1) @(TU(U) = #37771
          PKSEG=PKS=S+PDIFF?
     WRITE (F, C: (T) J, PKT J( )) . PKT G
501 FORMST(1H ,540KTJ(,10,34), PKS1G=,2F1 .3)
       CONTINUE
400
*SUM PK FOR PMY *J*
        PKJ(J)=PYJ(J) + PV(F5
     MRIT (6,00 t) 1,600 N O
5081 F(R) AT (1H .44-PK)(. 1.14) = .51 .2)
See Contitue
* SOF T -- PV
     WRITT((,, , !) (PKJ(J).PRTJ(J),FKTJ(J),J=!,NFAYS)
5 1/4 FOTH ( (44 , 35 . 3)
     CALL PRSOF
```

```
*ELIATNATA FAYS THAT
*FAIL CONSTRAINT TESTS
     J J=
     X \otimes P^{i_1} = X^{i_2} + \mathbb{D} X
     00 / J=1,454Y5
       AH05=AH4, 27 (BP2) (1) / JA
       71-77-Y1 2
       70=Y1-Y1-08
       51 > FF (J) = 11 L > 0 ( - 3, T 1)
     GAMP (U)=3 J (GOME+ ()) 10 (++,))/1/
     WRITE (C, ) F () F, es. J(D) + T (P(E) I J(J)) , YN, YREF , GNEE ((J)
7099 FORMAT(1H , TE, AF1 .3, F1). 7)
基本基本基本多的1.10倍(1.10倍)。
TETT
       1F(AFS(G VPR(U)), 3", 230) 5( 70 4
       11=11+1
       D1 : J(JJ) ===7 J(J)
       GINPE (JJ) = GLMPE ( ))
       PKJ(JJ) = ?<J(J)
     PKTJ(JJ)=FKTJ(J)
6.7.
     CONTINU
     N ?=J J
     WEIT (0, 3) 02
5005 FORMIT(18 ,3HNR=,) )
*TEST==MOTE THAT ONE FRATE
     N > I != 1
     IF(N: .LT. 2160 TO
     80 7 J=2,42
       IF(PKJ(J). 4: . PKJ(J+1)) 37 70 7%
       NMIN=NMI '+1
700 CONTINUE
750 WRITE (Fife F) KMIN
5005 FORMAT(14 ,5HhMin=,[4)
750 IF(NMIN.GT.1)60 10 777
#ONE MINE-UM
     INDEX=1
     50 70 11
*SELECT FIY ON PARITION
#ABSOLUTE WILLIE OF THE
770 THETE .
     I 40, Y=1
     77 5 J=1, 4414
       41/9=497(75*J(J)=- 171(1))
       IF(THETT._T.210) 70 73
       TH 71=113
       In OCY = J
SELL DOM DINE
```

```
816 PSINN=FSIJ(INDEX)
     DY=PX · TAN (PSIKN)
     WRITE (E, ' ( 7) 1403 Y, POT 44, DY
5007 FORMAT (14 ,104TN/3X,03144,0Y=,31,151 .3)
*TOTAL PK, FLIGHT PATH
     PKTOT=PKTOT+PKTU (EKOTK) FOOKST
     GO TO S
*ALL DO > F
850 T1=X0-X10
     WR17 (6,56 8)
5308 FORM T (1H .84L00F -5 )
     T2=Y1-Y1
     DY=DY+T?/T
*DISTALC FLOWS
909 DELW = 181 W + 3007 (1X 1Y+34 10Y)
*UPDATE FIRE FOSITION
     X := X \cdot + \cdot \cdot A
     A+=A++[ A
*PRINT HOW FOSITION
     WRITT (E, 2F ) XN, YN, PRIMA, OFTOT
*AT TARGET?
     T1=X1-X4
     TZ=YT-YN
     DT=80F1 (T1 F1+T2+F2)
IF(DY=DX)911,92 ,2 7
910 IF(DY=LE=(.5)30 70 351
 921 IF(Y' . NE. Y .) 30 TC FT
     IF (YT.NE.Y') GO TO 3
921
     DY= .
     GO TO 94
 937 TEMP=TAH(TEMT1)
     DY=DY*TEKE
940 X4=X4+CX
     Y N=Y > + [ Y
     95LN4=05L8 +3327 (34534+3443Y)
     WF1" (t, 25
                  ) x 1, Y 1, PT 141, TY 15 1
     WRITE (L, 27 1) XT, FELME
 95
950
    WRITE (6,27 1) XT, FFLAV, DP1, DP2
     IF (7 00. En. 1) 945 1 XT = PX 101
     CATI TEFKTO / PESEXP
     I 40=[1 F+1
     WPIT: (+,32 ) HATID
320 FORMAT(FOR PATTO OF TOTAL TXEGSTOOT TO DISSUING =.F1 .7)
     GO TO 1
```

606 *END OF TIPHE

12: 11: 9999 E() - 17 (FE() - 1) EGOLUT (THINK) X * 1 CHOOLLE IN EFECHT BY THE INDIFFERENCE STANFORM SAFETHER FORMAT(2F1 .1,I) FC9: 4T (17: 1251 -1)

284 565 245 23. 237 FORMIT(1 Y,C(K,F, -2), (X, F), (1), (X, F), (1), (X, F), (2), (X, F), (3), (X, F), (3), (X, F), (X, F) 20f HINTH= } 75.../177 / 347---, 77.77 / 0 HING. OF MIRSUNIMENTS= , 12/ X , 1744/18 / 3 FAN US = , 17/ X , 1744/14 / 77.47/19 / 0 HINTH=) 75.../17/ X , 1744/18 / 0 MINTH=) 75.../17/ X / 0 MINTH= (MINTH=) 75.../17/ X / 0 MINTH=) 75.../17/ X / 0 MIN ESHXH-CTATE (*1X, CHXHA) TITTO, SX, EARC (TAD), 2X, BHEX DOSJOEZ) FORM: T(10/4, T(40/4, T) - 7, T) ---/, 14, 10HX - 10T & TED Y-10TATED/) 1(-AVET)=,F:1,-2,17, 13475341(f:G[r]=,Ft ...,SH DEG.///)X,23HEK F/7/--FOR . T (1827/7 X) 1 FO DOWN (HTPH] --- //10X, FORMIT((* 4, F). 7))

3011 FORMST (18 , 21. 3, 17 . 7) HOUNT COH

ب 29

FORMET (1H , FX,

7.4.4

AL SHAP SIGHAE DISTR

VEDUCO.

3/20

```
SABBLALING SKITSE
*TO 031614 FK V 18. 18 FJ 137734 GF
*DISTARCE TO STUE(E) 487 1724154
*ANGLE (TH TE)
     ODMMTN/CT/T_/D, 11:14(2), 424GS(2), 55702(2), PK, 2KT43(1, 3, 2),
L = EL(2), THEF1, PI, KX
     THEFTITU(XX)
      TOMES = 1 S FO(KX)
      PK= .
*DISTANCE TO SITE VS LETHAL PARTIES
      IF(D=F((KX))) ,2 ,1
 #D<FL,000 FUTF F. 43
      M=(0/13+P)+1.
       GO TO 3
 ▼D=-E,OHTTE FIFT
     4=11 " Gr (KX)
 *TEST ANGLE, CPT (I'V BECT CP
      IF(THETA=FL) 5 ,5.,57
 #THETA CPT
      N=(THETA*T MP1/FT) +1.1
       GO TO L'
 *THETA =PI
       N=NSEC2(KY)
 50
       PK=PKTAF(M,N,KX)
 6 U
       GO TO 1
 # ERROR
       #RITS(6,00))||--||A
||FORM:||T(7/5,4,13485||CP-+PKSSPL,2X,6HTH(T)=,8244.7)
 8 C
 9:
       STOP
      RETURN
 100
       E1:0
```

SUB OUT INT PKSONT *USES # PURPLE TOTAL TOTAL OUT *TO AREN OF TOT! - PK FT 9 7404 *RAY IN ("C"NFI G 0 05%

50

200 RITHEN END

COMMUNICATION (TING NOTE , PK 1(11) , FAIJ (11) , FKIJ (11)

```
MEXT= ME VAC-7
     30 1 = 1-1,421
     IFL 5=
     4547=81145-1
     00 ' J=:, 3*I
     IF(FKJ(J+1).35.FKJ(J))30 TO 4
     T1=F<J(J)
     T 2=P5 I J ( J )
     TZ=PKTJ(J)
    DKJ(1)=[KJ(1+1)
     PSIJ(J)=P5 'J(J+1)
     PKTJ(J)=PK:J(J+1)
     PKJ(J+1)=T1
     PSIJ(J+1)=*?
     PKTJ(J+1)="3
     IFLAG=1
     CONTINUE
     IF(IFLAG.ED..)GO TO 211
100 CONTINUE
```

Appendix B FORTRAN Code Listing of the Revised TMPSA Program

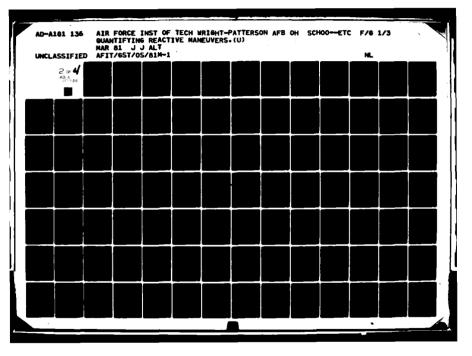
```
PAGG- AN OF THE (INDITED TO 1797) TAKE #3 4707 , C. PLOSEOU (PUL)
斯林斯斯亚斯斯斯约尔亚克茨克斯(西亚克加尔克斯沙特)尼亚
*PROGRAM TO COMPUTE (FITHIM
*PATH BETMEEN & CHECKENINGS
*MULTIPLE TYPE TIET
*MAX LINT: OF 1 PAYERS
    RAYC=11
    STICE = E'
    SITHREL
    STTE TYP CED
    RING /SITE TYPE=:
    $3000096/3108 14Pk=13
*SUBSCRIPTS--
    I=ST*F0
    J=1, Y9
    K=STIF TYP"S
    L=87 81
    OCM 18/07/ 12/7, 1 19/71), 27/860 (1 ), 43 02(2 ), 94, 9/713(1 , 4,2 ),
                - F(E)*, HIEV *13*KA
    DIMENSION MY(3 ), MS(1 T), TX(1 ), TY(1), ), ITMP(1 ), 344P2(11),
             x<(; ), $XX(*, ) * 2XX(; )
    PI=3.1.185 25
*CARD INCHT**** ******
THO. ANGLES, STEES, STEET, STEET TYPES, MC. OF MENSUREMENTS
    PER SITE, FANDARD DOWNAFTON OF GLAGE & ATIMUTH
    I 40=1
    RATIO= .
    REAT(0,1 ) JRLICO, 4783, 4833 F, 0 YES, 44, 5194 NO, 51944D
10
    WRITE (C, 1 ) USLID THERETH PRITE, TYTE, NY, SIGHA GEORGE
4100 FORMAT(14 ,511 ,2F1 ,1)
*TEST END OF FILE
    IFUNE(). F. ) GO TO TR
*CHECKPOINTS
    READ(),11 | ) Y3, Y5, YF, Y7
    WEIT ((, a ) ) XI, Y, , < T, <=
420 FUNCT (14 , FF. . 1)
```

```
*AWARENESS FARTUS, MINEMAK A/7 VELOCITY, CO : RIGOR WIGHT
     READ(S,11) ) R, VEE, VMY, 40
     WEIT (1, 2 1) 7, Ver , 1 14, 17
*SITE DATA ---
*NO. RING., SHOT PE, THIS FARE OR 2 KEI, TYPE
       Me. C. (K) = 46: U. (K) \J
     WHITE(F, 1 ) NAMES (K), METOT (K), INTV(K), ITP(K), METOT (K)
    CONTINUE
*SITE PK 327"
     DO 3 KHI TYPO
      17 = 1 1 4G5 (K)
       MT= NS202(4)
       WRITE (F., 2 ) ((PKT/7(4, 4, 4, 4), 5=2, 11), 6=2, 67)
    CONTINUE
30
*SITE COU ELAATUS,TYSE
    00 1 L=1, RITE
     401 FORMIT (1H .2F1 .2,35)
6 "
     CONTENUE
*END CARD INPUT FIRE
#INITIALT PATEON CONCORDS
     NRAYS = UCLICE+1
     OP1=SECONE (DP)
     PSI= .
     DPSI = .
     PSIMU=: .
     PKTO1 = .
     DELWHE! .
     TEMP=NSEG
     DX=F/TEMP
     T ミドヤ= VトセノV・Y
     IF(T-MF.GT.1. ) TT F=1.
     PSI=/ C(S(T-4P)
701 FOR JULY (FS 11 .+.T) //
TEMP= USETO
     DPS:=(1. 1 787)/77 F
     T 758=115=5
     CONST = 3: . . 1747 (1 _ M3 /M4)
     SIGHTF=SIC AD PI/ .
```

```
*SET-UP ALRAY R LATING
*SITE NO. TO SITE TYPE
     DO 7 L=1, SITE
        KK([)=
        90 - K= ., VTYPS
          YK(!)="((1)+1
          3 F (33 Y P(_) . 50. 387 (7) 130 10 4
        00.733497
###K # * * EDECE -> C MOTOH
        L!. = L
     $10P
     COMPRISE
*LETHAL 1 1 DO HS
     06 -- K=1, TYPS
-- PE(K)=== 35(K)+1+1+(4)
9
     CONTANU.
*INITIAL //O PORTION
111 Y 1=Y"
     V:(=V)
     XT=YF
     YT=YF
#END THITTPALITATION ---
*PRINT IMPHIT SU MARY
     00 12 K=1,41YPS
        117 = NET GS (<)
        NT=NSTC2(()
120 CONTINUE
*DISTANCE-+STIF TO A/C
      WRITE (F,CF E)
6005 FORMITO 1H
                                 TZ DN(L) ALPHAL(L))
                   Ti
200 00 21 L=1,NEITE
        71=5Y(L)-K%
        T2=SY(L)-44
0N(L)=00 f(T1 T, +TT T2)
600 FORMAT(LH 71=+F1 .7.7472=+F1 .1)
    . IF(T: . 15.1) 30 TO 31
      ALPHILL(L)=
      G0 70 81
211 ALPHIL(L) = 1000(T1,T1)
WROT (0,0 2) II, 11,00(L), 1LPHAL(L)
61.2 FO MIT(1H ,1F1 ,2)
211 0015 INUE
```

```
*TEST FOR SITES ATTHIN MASSENERS CATTUS
      ISIT =
     DO 23 L=1,NSITE
        IF(D)(L).31.R)60 [7 23
     K = KK(L)
      IF (XH. GT. (PK(L) +FL (<))) 37 TO 27
        IS_TT = 1S 1T: +1
        SXY(151T )=1Y(L)
SYY(161T )=5Y(L)
        KKK(3011) = KK(f)
     DIS((1811F)=DN(L)
     ALDB. F ([SI'E) = / LF'''_(_)
23
     CONTINUE
     WRIT-(+,01 3)18375
60 3 FORMAT ("H ($172=,0))
IF(3517E.ET. )30 7: 47
*AT LEAST OFF PONCT > P---
     RND=1 - 1710+101
     CALL FAGGET (PAG)
     70 T
            L=1,181"E
     719T:= .
     ₹[DH:F= .
     WRITE (F. 3.3)
     70 F LL=.,41
     RN= .
     00 3 till=1,52
     Y=RA, F(-4)
     ?्स±5 % +Y
370
      CONTINUE
     ₹N=""--.
     01STF=0ICT >+(1+813545624) *0IST(U)
     ₹%= .
     00 35 - LLL=1,12
     Y=R3\F(-1)
     RN=PN+Y
     CONT' NUE
380
     RN=PV-E.
     ALPHOP=ALPHAP+ALFHAR(L)+FTGMAF FN
     WRITE(E,37 1) FN,51GMAP,31ST(L),/LPH4 (L),S1GMAF,31STP,4LPHAP
      CONTINUE.
36÷
     AD(L)=DISTRIM
     AF(L) = ALOP P/NM
     XS(L) = LS(L) + CCC(LE(L)) + Cd
     YS(L) =1 7(L) 47, V( = (11) +44
     WRITE (1,31 T) XO(L) ( (_)
     Some this
35
=SET-NP LOCE FC RAYS
             J=1, VF: YS
       PKJ(J)= .
        BRILLE .
       TTWP= J-1
        Pm: J(J)==>2.+"///
```

```
*SET-UP LOOP FO RAY STG4TATS
        00 4: . I=1, NSEG
     PSITE1= .
      PSITE 2= .
          XP:=YK+T+DX
          YPE = YE + T = TY - TY + (TRT 1( 1) )
*SET-HP LOOP FO SINGS
          56 3 .=1,15175
      IFL49=1
            7 4=5 4X (1 ) + X= 1
            12=5 44 (E) -40 m
            [=80 [("1/"] +""""]
     IF(0.9%. . ) 30 TO 2
279
      ALPH' = .
      50 70 21
     ALFHS=ATAN (TE,T1)
240
     (*F@_4=(5)U[20) 30 1=172HT
25i
      IF(TUTT: CI.T.) THE WEST TO THE H. T.
      ΚΧ=Κ<Κ())
     CUFF BK. VE.
     MPITT (- ,- 11) FV
5011 FORMAT(:H , THEK=, F11, T)
     IFC:FL/6.1 .1) 60 70 25
     PSIT-1=PSI:F1+PK
     IFL29=2
     T1=X*([)-Y2?
     T2=YS(L)-Y3R
     D=S0:T(T1*:1+72:12)
     GO TO C
     PSTT 2=PSTTF2+PK
26
         CONTINUE
300
     IF(J. E0.1) PKTU(U) = P3TTE1
          PKSEG=PKSEG+FSTTT?
     WRITE((), () 11) J, PKT J()), PKT J()
5016 FOSMAT(1H ,5HPKTU(,10,3H),PKS76=,2F1 .3)
400
        CONTINUE
*SUM PK FTF BIV #J#
        PKJ(J) = F /J(J) + F K (523)
     WRITE(1, 1) 1.543(1)
5001 FORMAT(14 , HEY)(, T2, T4) =, T1 , ()
500 00 011110
*SOFT--PK
     MEITER: - - > (E) (U) D - 73F J (D) - 747 J (D) - J=1.45 3 Y (D)
5004 FO 4 7 (20 , 35 , 05 )
```



```
* ELIMINAT FAYS THAT
*FAIL CONSTIAINS TESTS
                  JJ=
                  X \times P = X + Y \times Y
                  00 / J=1,491YS
                         YMF7 = YH+75 W(PSCJ(I)) 3K
                         71=Y1-Y1-2
                         72=47-48 39
                         51: PF (J) -17: (8 (1 %, T1)
                  SAMER (J)=7 (30 MR) (J) 11 .+.1)/1
      WRIT ((:,)) - (:,) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:) - (:)
            ##(YÑE: 1.2.. .CE. Y45E.GE. )63 EU
***∵WJEGE TEJE
                          IF(186(6 MP:(U)).37.987136 TO 2
                          JJ= JJ+1
                          rs, J(JJ) =>S: J(J)
                         GA-BE (JJ) = 31 (FE (J)
                          PVJ(JJ)=~<J(J)
                  コペナリ(リリ)= 5 (でリ(リ)
60 BORTHART
                  44=11
                  WEIT (F, 7) YP
5050 FG2 34T (14 ,3450=,55)
TTEST--MOVE THE . ONE PRMENT
                  NHIMES
                  IF(% .LT.2)30 TO 74.
                                       J= 2 , 4 3
                         JF(PKJ(J).NE. PKJ(J-1)) GO TO T.
                        THE THEFT IN
700 CONTINUE
   750 WRITE (E, HI E) WMIN
5076 FORMAT (1H ,5HRMIN=,IT)
750 IF (NMIN.GT.1)GO TO 711
#ONE MINI 'U'
                  IND=Y=1
                  60 TO 61
*SELECT FRY OH PISTS OF
*ABSOLUTE WILLE DE TEST
770 THET := COO.
                  140~4=1
                  00 1 J=1,4514
                         4:3=235(031J(J)-7\100(H))
                          15(TH5"#.25./1010101 73
                        THE TABLE .
                        I: \cap Y = I
809 CONTENIC
```

```
810 PSIMMERSIU(FNOLX)
      9Y=9x~754(-5744)
5817 F000007 (19 , 1947) 6744, 745, 745, 31, 251 .3)
*TOTAL OK, FUIGHT OUTH
      DKILL (ALL C TELKIC THEKLICH POLAN COURSE
      60 TO G
*ALL DN > F
     T1=X"-XN
W: I" (F, " F S)
859
5000 FORM T (14 , 142005 AF )
      12=Y1-Y1
      DY=04~12/1
*DISTANCE FLOWS
908 DELWHEDELW .+ SORT (DX10X+3410Y)
*UPDATE W/O FUS TICK
      X E = Y^* + E Y
      A 1. = 4. + L A
*PRINT NEW FOSITION
+AT TARGET ?
      Ti=YT-Yh
      T2=Y1-YN
      DT=900Y (74: T1+T2: T2)
IF(DT=FX)9 ...,02 ,2 '
910 IF(DT+LE+D+3)30 TO 331
 92: IF(YT+1 E+Y 4) 30 TC (3)
      IF (YI.) E.Y a) SO TO SET
920
      DY= .
GO TO 9
 930 TEMP=TAR(T9/71)
      DY=DY+T+ND
943
     X #=Xn + L.X
      Y N=Y1+[ Y
      DEFIN = [ F[ N ] + 2 0 2 7 ( 7 Y ) 7 Y + 3 Y * 3 Y )
 951 WEST (F,27 ) YT, 17 JUS
954- SP2=7E(() 5(0F)
      IF(TYD.EC.1) FIGURY =FKTOF
RATIO=FKTOFKTKICTYP
      WELT: (1, 26 1) Y'', Y , 277 44. PYTCT
      INDEIND+1
32% FOR MIT (70H ENTITY OF 1971 - XMMLH)
                                                 TO BASCLINE =, Ft . 3)
       50 70 1
```

3. 29.7 28... **₩** 25.7 4. 235 191 1101 9999 38-1 FORMT (4H , TO) 1= + 1 . HOWER (IN TOX TO IN FORWET (1847/7 X, 175 TO RIMOR, , 17, 2 H 1925 NOT RATCH LIST, (31)) E017 (1 (1) 7 / / / 2 / 3 / / 18 = 3 + 60 + 0 3 = 9 0 10 131 OF JAPHT FERRITICALLY, 124811 CONTA---/11X, 104X-ROTATED Y-ROTATEDAL FOR (1 ((Y, F . 7)) FUT 117 (251 -1935) ECCE I (103 7775 X) ECCES TABLE FLICHT PATH INPUT SUITESY---// Y, 1 WIN E01-11 (1 E1 .1) FCPM: T(11,251 .1) MF ITT (1, 20 3/17, 12 (1) 3.4 DIST ALPHAR SIGMAR DISTR COSTANCES, Foregray X, 1 HOISTALE CONTROL

円とり

al min

```
SUPTIME PRIME
*TO 087679 FK VILIE 65 FJ907709 GF
*DISTANCE TO SITE(S) AND ATTAITH
*ANGLE (TH TA)
     COMMUNICATE REND, INT V(1 ), WINGS (2 ), RSICS(2 ), RK, PMT13(1 , 1, 2 ), 8 ), RE(2 ), THEFT, PI, KX
     TEMPETITY (KK)
     TEMP: = 1 SEC (KX)
     PK= .
*DISTANCE TO SETT VS LETHIL RADIUS
     IF(F-FL(KX))1 ,2 ,3
*D<RL, COTTUTE F NG
     M=(0/T(~P)+1.
     GC 70 3
*D=#L,08774 8189
28 M=48 YGS (KX)
*TEST ANGLE, 091 NIN SECTOR
     IF (THETA-FILE ,E ,E)
*THETA <PI
     N= (THETA: TOMP1/F1) +1. '
40
     60 TO 6:
*THETA = PI
     N=NSFC2(KX)
50
60
     PK=FKTAP(M.N.KX)
     50 70 1
#ERF 72
     WRITE (6 .C. ) THETA
80
     FURNIT(/// 4,13450 %--><5351,2X,00(H 7)=,310.7)
9:
     STOP
     २ इसमा ध
100
     END
```

```
SUB BUTTHE PKSOAT
*TO ARRANTE TOT _ PK FIN TACH
*RAY IN ALCENIT 3 0405
     COMPREZOCC TYMERY", PKJ(11), PSJJ(11), PKTJ(11)
     NRMS=1F5YS-1
     00 1 1=1,401
     IFLAGE
     1-3444M=[M. N
     70 J=1, 24I
     TF (F<U(U+1).GE. PKJ(J)) 50 TO
     T1=F<J(J)
     T2=PSIJ(J)
     T3=FKTJ(J)
     PKJ(J)=PKJ(J+1)
     PKIJ(J)=PKIJ(J+1)
PKIJ(J)=PKIJ(J+1)
     PKJ(J+1)=T:
     PSJJ(J+1) = [2]
     PKTJ(J+1)= [3
     IFLAG=1
50
     CONTINUE
     IF (JFL/G.EG.1) GO 10 211
105 CONTINUE
```

200

R FTURN END Appendix C
Raw Output Data

Appendix Cl
Basic Model Output

OPTIMUM FLIGHT PATH INPUT SUMMARY---

CHECK*T 1	OHECKSI 1 1.686, 35,135 2 156,000, 35,135		AZC VELOGITYMIN=548.5 MAX=645.5	MAX=645.5	CORTONS BAMMA	19 14 • •
NRAYS=11	NO STEPS=25	01	NO SITES= 37	NO SITE TYPES= 1	§ 1 1	
NO. OF MEASUREMENTS =	SEMENTS =		SIGMA (RANGE) =	د . د .	SIGMA (ANGLE) =	

PK NATA--EACH SITE TYPE

SITE TYPE: 10. SITE TYPE NO = 1

	231				200
. J. F.	.137	205	.217	125	JANUAR INGS
.:12	£ 5 1 .	. 253	.172	.117	92745456
2	144	• 1 3F	.1:7	.179	CICES

SITE DATA--X-ROTATED Y-ROTATED

14.13	39.11	134	1
r 3)	27.	1.0	1
24. :3 53. : 3	35,11	11.1	1
53. 3	4 2. 1	1 7	
81.0	31.13	11 b	
7' • -3	4 3	1 f 3	1
81.0 7'1 181	4 3. 7	1 . 5	1
F.F. 37	1, 6, 5 3	1(1
84.3	\$ 6.00 \$ 3.00 2 5.00	11.5	1
84 • :3 52 • :1	25.	111	1
69	49.71	1	1
84 · .3 62 · .1 69 · .1 27 · .)	₹ 2. ` კ	130 100 100 100 100 100 100 100 100 100	1
15. 3	40. 1	1 ,	1
39. 3	\$ 2. \ J \$ 9. \ 1 8 h. \ 3		
6 . 1	1.8)	1	1
18.3)	2.13	1.0	1
15. 3 39. 3 6. 3 18. 3 71. 5	54.01	1: 4	1
	85.11	1(4	1
94 57. 3 38. 1	50.11	10.	1
57. 3 38. 1 56. 11 18. 21 35. 31	37.000000000000000000000000000000000000	1: 0 1: 0 1: 0 1: 0 1: 0	1
56.13	15.73	1.0 1.0 1.0 1.0 1.0	1
18	13.13	1:0	1
35.11	2:.11	15 û	1
67.3 47.3	59.13	11.0	1
47.33	14.77	194	1
E C . 43	6.11	162 100	1
E(.4)	2 2. 5 5	4 : 11	1
18.:7	54.*1	100	1
83.13 76.13	59,11	1.3	1
76.13	58.37	10.0	1
96 . 13	49.33	100 100 100 100 100	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
76. 3	18.75	106	
94.31	49.33 18.79 82.30 54.53 2.33 16.73	125 1.:	1 1 1 1 1
45.10	82.10 54.10 2.1) 16.10	1 . : 1 · i'	1
6 13	2 ;	1 v i.	1
65.)	16. "	1	1
65.) 89. /}	1 3.	161	1
		- -	-

PROGRAM OUTPJI---

	X-ROTATED	CETATED	ANG (FAD)	EXPOSURE
	1.93	35.17	(• 0 t = 0	0.00
	2.00	35.33	instit	(.09
	3.00	35.77	instici	6.00
	** D (35.31	₹ • ↑ ₹ 1/ ↑	₽•60
	5.00	35.11	(• . t	• 42
	A. 91	3= 23	5.000.25	.83
	7.56	35.)1	6.000	1.25
	8.00	35.11	101	2.53
	3.00	35.13	1.11	3.82
	100	35.11	₹ • / 1 v ±	5.39
	11.20	37.13	\$ • . F # f	7.12
	12.0.	37.19	• 1 1 1	8.85
	13.33	35.11	1.00 Put	11.74
		35.11		
	1 > 50			13.92
	13.05	35.11	\$ • f f & £	15.1.
	15.0.	35.13	1.06.1	17.18
	17.00	35.11	f. USi	20.13
	19.00	35.13	Ǖ17 u l	23.08
	13.00	35.13	iectic	24.96
	2 4 5 (35.1)	i o' i o i	26.63
	21.30	35 • 33	4.00(4)	28 • 54
	22.36	35.13		30 • 26
	23.51	35.17	1.000	31.21
	2 1. 84	35.71	€ • : E 25	32 • 61
T1=	6 • 1 T2 =	.3 <u>.</u> n		
	25.00	35.13	1.000	32 • 45
	25 • 6 f	35.11	6.0736	33 • 04
	27.00	37.97	to the	33 • 64
	28.06	35.3)	Souteb	34 • 39
	29.00	35.01	5 C L I	35.15
	31.00	35.33	3.31.5	35 • 39
	31.30	35.11	6.00 6.00	35.64
	32.14	35.7)	i. iii	35.78
	33,30	35.11	5 • v L 0 t	35.92
	34.00	35.33	5 . L & L	36.06
	35.50	35.11	1.0000	35.06
	35.00	35.71	6 C . C	36.36
	37.00	35.13	りゃくしょう	36 • 55
	38.00	35.71	1.00	36 • ⊍€
	39.10	35.11	0.21.41	36.06
	4 . * * * *	35.11	اب:: ٠٠	35 • 46
	41.31	35.77	10,551	36.96
	42.1.	35.5:	•	36.16
	4 7.31	35.10	• . (2)	36.66
	44.31	27.31		36.06
	47.37	77.33		36.05
	4 .)	37.13		38.47
	1	71		36.92
	69.1.	77.19	1.5000	37.36
		77.11		37.61
	F	77.03		35.79
	F 1. ^ .	32.33	•	39.77
	5 2. 1	35.11	•	46.98
	,		• ,	.

57.30	35.71	5 • € € 5 €	42.13
54.35	35.13	r. Fig.	42.69
50006	35.13	1 • 1 · 2 · 0 · 0	43.19
F 5.33	35.1)	1.01.2	43.71
57.01	35.11		43.90
F 3 • 01	35.13	•	44.14
59.30	35.15	3.6.6.6	44 • 34
€ 0 €	35.1)	5 . F	44.43
61.37	35.1)	a extet	44.43
62.01	35.13	· · · tul	44 • 65
£ 7	35. 15		44.66
F4.0	37.11	tettet	44.66
6	₹=. 11	4 • • • •	44.05
E 1.01	35.13	111	44 . 83
6 T. 31	39.13		45 • 1 :
63.76	37.11	£ • 51 • 7	45.32
E 3. 7.	37.11		45.54
7 • 1	35.13		49.77
71.7	37.13		45.83
72.5	35.1)		
73.00	370 13	i • · (L L	45 • 9 ÷
	37.77 77.33		4E • 33
7 /• 31	35.13	0.04.50	46.67
73.15	35.13	· • · · · · ·	48 • 15
76.55	35.1)	3.11.1	49.43
77.00	35.11	E . Trum	50 • 19
79.51	37.33	1.5	52.06
79.50	35.33		53.92
80.00	35.33	1.01.01	55 • 78
81.86	3=.11 35.11	1.4	57 • 64
82.00		1.04.21	59 • 31
23.61	37.1)		60.97
80.52	35 • 43	0.0100	62.64
85.00	35.33	U . 1 7 97	62.92
86.31	35.13	6.0(7)	63.21
87.00	37.13	1.0061	63.52
88.00	35.11	0 0 . C . C	63.72
87.00	35.17	i. Iti	63.93
91.00	35.11	4. 6777.	63.99
91.00	37.73	for Cut	63 • 9 3
92.5	35.12	forter	63.99
93,01	35.13	1.181	63.99
Q . , 1:	35.73	! . !	63.49
99.05	37.71	Gal Dut	63.99
95.5.	35.03	to the	€3.99
97.00	37.11		63.99
93.17	₹₹. 11	To the	63.99
93.3.	35.13	10-135	63.99
10	37.17	· · · · · ·	63.99

Appendix C2
Control Model Output
(see Table 2)

OPTIMUS TEST PATH INPUT SUMMARY---

	. 133 . 272 . 281	SITE T	PK DATAFA	NO. OF MEASUREMENTS =	VRAYS=11	CHECKPT 1
. E 37	125 217	SILS TYPES ILS	DATAFAC4 SITE TYPE	DREMENTS =	NO STEPS=25	135.186 35.185
6 6 6 (2) (3) (3) (4) (4)		// U L	4 0 H	-		생 생 생 생 생 호 네 3 네 3 네 3
* * * * * * * * * * * * * * * * * * *	• 17 c	STIE TYPE NOW 1		SIGMA (RANGE) =	NO SITES= 38	A/C VELOCITYMIN=648.0
				ر ن• ن	NO SITE TYPES	MAXH OAB . 1
				SIGMA(ANGLE)=	150 H	CORFICE NINTH 25
				• 7 7 5		10 0

SITE DATA--C=TATOS-Y GETATED

152557186 1525	37.00	100 100 100 100 100 100 100 100 100 100	111111111111111111111111111111111111111
90 • 77 75 • 7 9 • 7 • 5 • 5 • 5 • 5 • 5	9.13 18.11 2.13 2.13 2.13 15.13	1	1 1 1 1 1 1

PROGRAM OUTPUT---

	X-ROTATED	Y-ROTATED	ANG (RAD)	EXPOSURE
	1.01	37.33		
	2.00 3.00	35.1) 35.13	9.000 19.000	0.00 (.00
	4.50	₹5.1)		i e e ii
	5.00	35.13		• 42
	ó. C 5	37.11		.83
	7.91	37. 3	e • f :	1.25
	13.8	35. 13	6.0037	2.53
	9.5:	37.31	• 5 (4.1)	3.82
	1 . 0:	37.01	cobtat	5.38
	11.01	35.11	to it.	7.02
	12.21	35.11	Contrat	8 • 88
	13.50	35.11	to fut	10.74
	14.50	35.13	• CLERT	13.02
	1:.55	35.33	in int	15.10
	15.30	35.11		17.18
	17.01	35.31 35.33	1 . · · · · · · · · · · · · · · · · · ·	2.13
	18.70 19.70	35.21	i elitat Melitat	23 • (; \$ 24 • 95
	2 (• 1) u	35.11	i of i of	25.83
	21.50	35.13		28 • 34
	22.0€	รู๊ก•ู้ า ทั่	1.6111	31 • 26
	23.75	35.11		31.21
	24.00	35.73	1 . 1	32.01
T1=	(• T2=	•		
	25.5r	35.00		32.45
	25.00	35.13	• . .	33.34
	2'.0(35.00	· • . (· (33 • 64
	28.09	3=.31	for Cur	34 • 39
	29.11	35.11	じゃくじょし	35 • 15
	37.00	35.77	100	35.39
	31.56	35.11	1.3660	35 • 64
	32.00	35.13	(• (u (35.78
	33.00 34.00	35.33 35.33	. (35 • 92
	35.60	37.17	6.3191 1.166	36 • 46 35 • 46
	36.00	35.33	E-9191	36.46
	37.58	35.13	1.01.1	36.46
	38.00	35.10	Cactua	3b•u6
	39.00	35.1)	! ()	36 • 95
	47.00	35.33	0.194	36 • 66
	41.00	35.17	1.11	36.16
	12,00	35.33	6.031.04	36.86
	₹ ₹,57	37.11	• • • •	35 • ∞ 5
	I + ₂ ⊕ \$ (35.11	b a w to b a	36 - ⊍6
	L 3	35.33	* • (C ()	36.06
	45.00	3	forther	36 • →7
	47.7t	35.13	•	36 • 92
	43.00 10.00	35.13	1. F.	37 • 30
	49.66 5 .6	35.11 37.11		37 • 81
		37.17		39 • 01 40 • 32
	5 2. (v	35.	•	41.64
	26000	<i>y</i> · •		41.04

53.00	35.33	1 • 1 ((43.17
54.38	35.11	1.000	44.34
F5.50	35.11	• 1 66	45.61
FA.Di	33.11	Last of	46.87
57.11	35. 13		47.29
53.71	35.33	(tol	47.71
F9.31	35.11	1.011	47.97
6.36	35.11	2. iti	48.17
61.00	35.23	1. (3)	40.24
62.31	35.1	1.1.1	45.53
63.80	35.13	• 1	46.53
6 . 3 .	35.33		48.53
65.51	3=.11		48.53
65.21	35.11		46 • 75
FY. C.	35.07	• 1	48.97
13.63	35.33		49.19
£9.50	3F. 1	Control of	49.42
7.666	37.11	1.0	40.64
71.01	35.11	i • . ↑ = i	49.71
72.00	35.11	5 • 5 4 • G	49.77
73.01	37.13	i	50 • <u>2</u> 6
73.56	35.33	1.01.02	574
70.01	35.13	coftet	52.32
75.36	35. 1	Lock 'I	53.31
77.0.	35.00	0.0000	54.67
73.61	35.1)	1.11	55.93
79.00	35.13	Confut	57.79
203	35.31	1.6 16.56	59.65
81.35	35.13	[61.51
82.51	35.11		63.18
83.10	35.33	1.00.30	64 • 84
84.00	35.13		66 • 51
85.30	35.31	6.5000	65.79
85.00	35.11	600000	67.08
87.03	35.23	1.000	67.39
88.50	35.33	36.26	67.59
89.00	35.13	[• ≥ f (i	67 • 8 5
9(:	35.1)	1.46.01	67.07
91.5	37.7		67.67
92.0	37. 1		67.87
93.06	37.11	. []	67 • 87
96.55	35.19	• 6 1 3 4	67.87
9	3= 1		67.87
од. 1	7 T		67 • 87
97	35.)	• • • • • • • • • • • • • • • • • • • •	67 • 87
95.B.	3=. 1	10.00	67.87
93•0 93•1	37.11		
		• . !	57 • 87
10 .05	35.13	1 . 1 21	67.87

SITE DATA---X-ROTATED Y-ROTATED

14.03 14.03 14.03 14.03 14.03 14.03 15.03 16	323+344-43-73-73-73-73-73-73-73-73-73-73-73-73-73	10 1 10 10 10 10 10 10 10 10 10 10 10 10	111111111111111111111111111111111111111
90 • 73 75 • 03 94 • 13 45 • 93 60 • 23 60 • 3	49.77 18.77 52.71 54.77 2.7 16.77	12 0 17 0 11 0 11 0 17 0 1 0 1 0	1 1 1 1 1 1

PROGRAM OUTPUT---

•	X-ROTATED	Y-FOTATED	ANG(FIG)	EXPOSURE
	1.01	35.73	€ • F " 1	L • J J
	2.30	35.11	1.1.1	じ• J C
	3.00	35.03	V • 1 ()	0.00
	× • 22	35.11	•	ក ូ ប្រព្
	5.85	35.33	1.11	•42
		37.13		. 83
	÷ , , , , , , , , , , , , , , , , , , ,		. • • • • •	
	7.0	37.33	• • • •	1.25
	8 • ှိ မှ	35.93	. •	2.53
	9.00	34.17	1 • C 1 1 ×	3.82
	1 (4.3)	35.11	1. 1.	5.38
	11.35	35.1)	· • !	7.82
	12.00	35.11	C	8.88
	13.01	35.1)	1 • 2 ° 3 ° 3	11.74
	1 4.11	35.11	• 1	13.12
	1 - 00	35.13	1.00	15.13
	15.01	37.1		
			• . • .	17.13
	17.0i	3=.11	s of the C	20.13
	12.00	37.03	1 6 C C	23 • 98
	19.00	35.73		24.93
	2 1. 11.	34.33	. •	2E • 83
	21.13	35.33		28.54
	22.01	3=	1.41	31. 25
	23.30	35.11	to:tat	31.21
	24.5.	3=.11	•	32.01
T1=	6 T 2 =			02.461
• • -	23.36	35,11		32.45
	26.30	35.11		
				33 . 04
	27.00	37.11	1.0	33 • 64
	28.00	35.13	www.tif	34.39
	29. 90	35.11	6 . n 6 6 5	35.15
	3 •	35.11		35.39
	31.95	35.1	C. 20 51	35 • 64
	32.43	35.13	• Lti	35.78
	33.00	35.11	3.00000	35.92
	34.00	35.3)	· · · · · · · · · · · · · · · · · · ·	36.46
	35.00	3=.13	deal Pt	36.66
	35.00	35.13	f. Hab	36 . 96
	37.00	35.11	1.1(1	36.36
		32.33 35.33		
	33.10	•		36.36
	33.00	35.01	6.0664	35 • 66
	r • J3	35.13	$e = a k \otimes e$	36 • 0.5
	41.00	35.11	10100	36 • 3 5
	4 2.5 €	ৰল্ ুণ	* • · · · · · · · · · · · · · · · · · ·	36.76
	43,30	35.11		36 € € 6
	431	35.13	• . *	36 • 96
	65.	78.11	•	36 ⋅ 06
	45.23	3=. 11	1.00	35.47
	67.7	3~. 1		36.92
	43.7.	3=.))		
	10.0.	37.73		37 • 78
			• • •	38 • 64
	5 . 9 9 5 4 3	35.13		41.91
	51.0h	37.7	•	43.17
	5.3.50	35.1)	• :	45.14

53,36	35.73	333263	47.98
5 i	3=.19	6.5566	56.13
55.20	35.13	Cattle	52.27
F6.11	35, 11		54.42
F7, F;	35.10	for Lat	5€ • ∪ 6
	35,11	1	57.71
50.11	3=.11	1.000	99.35
€ 1.18	3=.11		59.70
£1.76	35,19	Coctai	59.92
62.11	3= . 1	t o . S et u	67.33
	35.11	tooi (1	66.52
E 3.2:	37,11	t was a constant of the consta	66.06
64.16	37. · · ·		61.65
E 1, 00	37 4 13	• 1	
€ 5.00	35. :)	• • •	61.38
67.23	35.11		61 • 11
63.00	35.))	tattat	61.33
63.00	37.17	• 4 4	61.55
7 .00	3=.11	1.	£1.77
71.15	37.31	:	61 • 64
72.36	35.33	coacti	61.91
73.01	35.13	E	62 • 39
74.01	35.01	t • (€ t	62.37
75.00	35.13	6.0665	64.15
70.00	35.11	€•3€6€	65 • 44
77.31	3⊼.33	1.166 1.666	66 • 20
78.00	35.13	\$. F & F	58.06
73.60	35.11	1.31.01	69.92
80.00	35.11	for a to to to	71.78
81.30	35.31	0.3641	73.64
82.00	35.31	1.1161	75.31
23.31	35.00	L.0081	76.98
84.36	35.13	0.√ bàt	78.64
85.30	35.00	1.3696	78.93
86.55	35.33	boutul	79.21
87.CG	37.11	1.0111	79.52
85.73	35.11	1.1565	79.73
83.30	35.71	"	79.93
91.00	35.11	1. C C 4 C	83.40
91.26		i. ili	80.01
92.00	37.19 37.19	101111	80.40
93.20	77.17		81.00
	3.	i e ciul	80.33
9 .00	3=.11		81.03
9:00		• 1 K*	81 • 0
Y 30 30		* 10	87.03
97.55			
98.71	37.37	.:(81.03
93.00	35.37	y • × € 25	8
10 . 35	₹5. `}	• • • • •	8 1 . 35

SITE DATA---CETATOS-Y DETATOS-X

14.13	39.11	110	1
53. 1	27. 1	1. 1	1
24.43	35.13	1. (i
53. 1	2.	1. 2	1
81.1	4 2 . 1 3 3 C . 1 3	1	1
7: . 23	k h	1 7	1
18.)	43.12	1 :	1
56.4	46.73	1: 0	1 1 1 1
81.	4.4.73 43.13	1.	1
62.11	25.03	11.7	1
55.02	32.51	1. 0	1
69. 3	49. 1	1! č	1 1 1 1 1
27.	5 2. 1	1/ 0	1
15.)3	5 2. 1 4 0. 1 1	160	1
39.53	5 1	1.6	1
6: 13	5 4. 1 1 8. 11 2. 1	1 0	1
18	2. 1	1 · 3	1 1 1 1 1
71. 3	54.10	1.5	4
94.	6 F.	1 1	1
57.12	5	106	4
38.13	17.		1
56.1)	4 E 4 5	105	1
18.33	15.17 13.1	1 B 1 D	1 1 1
	2	4	1
35.30		1: u 1: 0	1
67.13 47.13	59.1)	1.6	1
	14.33		1
56.01	6.33	196	1
55.33	2 2. 2 1 5 4. 2 1	1 0	1
18.J3	54627	1. (1
83.11	59.")	11.6	1
75.33	58.00	11.6	1

91 • 37	49. ~ ;	1 :	1
7E.)	18.	100	1
94. "	5.2. 1	1 _	1
45. 3	34.37	1 0 a	1
51.11	2.	1 "	1
66.13	16.55	1 ^ i.	1
£0. 1	18.37	1 : 、	1

PROGRAM OUTPJT---

	X-ROTATED	Y-ROTATED	ANG(FAD)	EXPOSURE
	1.0 i	35.11	• • •	0.00
	2.00	35.11	t	3.30
	3.0n	35.13		0.00
	4. 7.	35. 77	C . C . P	မ 🕶 🗗 🐧
	3.00	35.33	• • • • • •	• 42
	5.55	37.31	•	. 8 3
	7.21	37.13		1.25
	3.20	35.11	• [1]	2.53
	9.30	35.11	. • • • •	3.82
	1 1000	3=. 1	• • • • •	5.38
	11.00	35.13	100501	7.62
	12.0.	35.01	•	8 • 88
	13.00	37.33 37.33	1.71	11 . 74
	14.61	35.11		13.02
	17.57 13.01	37 € 17 38 € 18	• • •	15 • 17 17 • 18
	17.01	37.43		2' • 13
	18.00	35.13	e tui	23.56
	19.20	35.11		24 • 96
	2	3=. 11		25.83
	21.	35.11		28.54
	22.00	3=. 1	6.00	31.25
	2 3 . 3	35.11		31.21
	2 * . 3 :	35.13	testar	32.1
T1=	ü.″T2=	• •		
	25.00	35.73	1.5	32.45
	28.70	₹5. 13	6.000	33 • 44
	27.01	33.11	i. bif	33 • 6 4
	23.66	35.31	for killing	34 . 39
	29.0.	35.11	i. Ivi	35 • 15
	3 / . 8 (35.13	4 6 4 6 4 5	35.39
	31.90	37.17 37.13	6.3566 6.1636	35 • 64
	32.00 33.00	35.33	resttt.	35 • 7 8 35 • 9 2
	34.31	35.13	6.6176	35.92
	35.00	35.73	1.01101	36 • 86
	36.10	35.73	1.61.00	36.96
	37.50	35.11	10)(01	36 • ú 6
	39.00	35.11	Lauf (r	36.05
	33.0c	35.00	6.6657	36.06
	43.03	35.11	E . 1 . 2	36 • ≥0
	41.00	35.7)	10.101	36.66
	42.3:	35.11		35.05
	43.00	35.73	5. (3)	36 ⋅ 06
	4 4 . 30	35.73	"• L"•	36 • 96
	6 .56	3=.71	1.1.1	36.35
	46.16	35.13	t • • • • •	36 • 47
:	47.32	35.33	(• • • • • •	37.33
	44.5	35.11	1.0	36.19
	ይዓ.በር መጠቀ	37.33 37.33	1.00	39 • 92
	5 .07 51.11	3= 11	v • v	42.19
	52.55	37.11		44.73
	カニ・リン	370:1	' • .	47 • 5 :

53.00	35.11	1	50.22
F4.5(3=. 17	10.00	51.93
57.30 53.10	35.77	6.000	53.64
r to the	37.11	. • L	55.35
F7.00	37.13	10 · 14 · 4 · 4	50.51
59.01	3=.11	Barton	57.66
Fair	35.11	1 . 51	
	330 1	10-131	58.45
6 2.51	35.03	(·	59.34
E 1.00	35.33		60.10
6.2000	75 43		
€2.15	35.17	10 to 10 to 1	61.57
€3.00	35.1		61.61
€ - • 5 0	37.1	(• · · ·	60.95
	35.11		
6 5 • 0 €		•	61.19
€5.8.	35.33	% • € * . €.	61.31
67.14	35.13		61.53
	77 13		
63.36	35•1)	to • an exist	61.76
€ 3.00	35.11	1.6	€1.98
7 .::	37.23	1.0	62.2.
7	7	• • •	
f 3.0 2 2	35.1		62.27
71.11 72.00	35.11	to a section.	62.33
73.60	33.23 37.11	1 . 1 . 1	62.82
71. 10	7		
74.00	37 • 11		63.31
7	37.13	1.1.26	64.58
74.33	35.13	201661	65.87
77.00	72 12		
	35.11		66.63
78.01	37.11	1.7551	68.49
79.00	35.11	· · · · · · ·	70.35
		(0)	
8 • 25	37.13		72.21
81.00	35.13	€. Use	74.07
82.06	35.J]	Coulde	75.74
83.01	35.3)	1.500	
			77 • 41
84.00	35.33	1.021.06	79.97
85.30	35.13	100630	79.36
86.93	35.11	botout	79.64
87.00	35.21	100000	79.95
88.10	35.33	0.5666	80.16
89.01	35.11	1.11	81.36
en.gu	35.23	じゅうじしし	8r • 43
91.00	35.11	Lather	8. • 43
92.30	35.31	intlut	80.43
	37.73		8: • 43
63.00		ke car	
64.30	3=.73	! !	81.43
9 3	35.17	1	8 . +3
93.00	37.31	to'tat	8"•43
97.01	37.11	to tot	81 • 43
20.00	35.13	Lockat	81.43
	35.13		
99.00		• • • •	8 • 43
10 .00	35.11	$C \bullet a f \circ f$	8ۥ+3

SITE DATA--X-ROTATED Y-ROTATED

13.00 13	32343443434617077177777777777777777777777777777777	1.0 1.0 1.0 1.0 1.0 1.0 1.0 1.0 1.0 1.0	111111111111111111111111111111111111111
76. 12 76. 12 9. 12 65. 12 65. 13	18. 13 52. 1 14. 13 15. 13 16. 13	1.0 1.0 1.0 1.0 1.0 1.0 1.0	1 1 1 1 1 1

PROGRAM OUTPJT---

	X-ROTATED	Y-ROTATED	ANG(FAD)	EXPOSURE
	1.08	35.10		9.31
	2.30	35.00	1.10	1.00
	3.00	35. 11	•	G_{\bullet} as
	4.50	35.30	(· · · · · ·	0 • 0 J
	9.33	35.11	• • • -	. 4.2
	5.00	35.)1	•	.83
	7,35	35.11	1.11	1.25
	3 • € €	35.	• ;	2.53
	9.00	37.11		3.82
	1 * * >	35.01	1 • •	5.30
	11. 00	35.11	feater	7 • ∪ 2
	12.7%	3=. ")	v o i i a l	8.88
	13.00	37.73	1.566.25	10.74
	14.00	₹ # • ` \$	• • • • •	13.72
	16.00	3*• 13	نوائها	15.1 3
	15.1	35.00		17.18
	17.38	33.13	តិ∌្តសិស្តិ	23.13
	13.0.	35. 3	201767	23 • ⊕8
	13.0	37.11	te. Cal	24.96
	2	35.73	1.11.1	26 • 83
	21.0	37.	to fai	28 • 5 +
	22.1.	37. 21		31.25
	23.16	35.13	* • 1.64	31.21
	24.33	35.13	1.16.5	32 • (1
T1=	0 • ↑T2=	: • *		
	25.00	35.33	1.11	32.45
	25.10	কুল•ু উপ	to. Chr	33 🗸 🗸 4
	27.00	35.13	for tit	33 • 64
	28.00	35.00		34 • 39
	29.00	35.11	0.0100	35.15
	37.10	35.11	Fa ful	35.39
	31.00	35.33	1.0000	35.64
	32.01	35.30	1. • 2 11 0 1	35.78
	33.00	37.13	t • c * 0 €	35.92
	34.00	35.13	Lautin.	36.06
	35.00	35.11	bestel	36 • ⊕6
	35.01	35.33	6.7631	36.06
	37.09	35.13	I was to be	36 . 66
	33.DI	35.11	6.00	36 • 16
	33.31	32.11	L. Jiif	36.36
	6 4.33	35.9)		36.06
	41.50	37.34	Leibie	35 • 46
	42.64	35.33	(• ' » •	36.16
	43.37	35.33		36.06
	44.7	35.23	• 1 2	36.96
	4 to 6 12	35.1)	t • 1 (1)	36 . 16
	46.00	33.44	s • 6 4	3F.89
	47.70	35.11	€ ⊕ ∞ ° ∞	37.75
	45.72	35.31	اع تيوي	39.61
	49.06	37.	N	4! • 3×
	F	35.31	7 · 12 f	42.01
	51	33.00	• • •	45.15
	5.2.31	37.13	• ° ≤ ↓ €	47.92

```
F 3. 00
                             35. 17
                                        * • • f % t
                                                        50.63
                 54.00
                             35.33
                                        52.65
                             33.11
                 53.11
                                                        53.89
                            37.13
                                        50.31
                                                        55.14
T1=
            [ . T 2=
                             35.13
                 57.30
                                                        5º .78
                                        for their
                 5 ....
                             35.13
                                                        56.57
                                        . . . . .
                             35, 13
37, 23
                 53.30
                                                        57.37
                                        d. def
                 F ...
                                                        58.25
                                         . . . H
                 E1. . .
                             35.11
                                                        59.11
                 F 2.5"
                             37, 13
                                        f . ( )
                                                        59.48
                 £3,10
                             3: 11
                                        1.000
                                                        59.72
                 F .. 31
                             35.11
                                                        59.86
                                        1.000
                             35.11
                 E 7 . C .
                                        11-6-11
                                                        65.50
                 €5.C.
                             37, 11
                                                        61.36
                                        . . . . . . . . . . . . .
                 E 1.30
                             35.13
                                        10016.1
                                                        66.58
                             37.73
                 63.1.
                                        i. lu.
                                                        65 • 81
                 €3.11
                                        west wi
                                                        61..3
                 7 . .
                             3=. 23
                                        61.25
                                        . . .
                             37.13
                 7...
                                                        61.32
                                         - \frac{1}{4} \cdot f = f
                 72.10
                             35.31
                                                        51.38
                 73.00
                             37.11
                                        · . . tal
                                                        61.87
                 74.30
                             35.13
                                        Colbic
                                                        62.35
                 75.66
                             3=,11
                                        1.6.194
                                                        63.63
                             37, 11
                 75.16
                                        5 . . ! b c
                                                        64.92
                             35.11
35.11
                 77.00
                                        1.6.6.6
                                                        65.68
                 78.33
                                        0.000
                                                        67.54
                 79.01
                             35.13
                                        1. 1. 1. 2.
                                                        69 • 45
                 8 500
                             35.13
                                        1.1111
                                                        71.26
                 81.0i
                             35.33
                                        1. 66 8 23
                                                        73.12
                 82.30
                                                        74.79
                             35.71
                                        2.00606
                 83.50
                             35. 13
                                        F . T F . 3.
                                                        75.46
                 84.0£
                             35.13
                                        deatest
                                                        78.12
                             37.91
                                                        78.41
                 85.gr
                                        Section.
                 86.00
                             35.11
                                                        78.69
                                        1.0126
                             35.33
                 87.00
                                        Taction
                                                        79.00
                                        7.000
                 83.00
                             35.31
                                                        79.21
                 89.5.
                             33.31
                                        * octif
                                                        79.41
                 o ...
                             35.13
                                        30.601
                                                        79.48
                             35.11
                                        1.015
                91.00
                                                        79.48
                92.31
                             35.13
                                                        79.43
                                        1.00 1.00
                93.70
                             35.11
                                        5 · :
                                                        79.48
                94.01
                             35.13
                                                        79.48
                94.06
                             35.13
                                                        79.48
                95.;
                             37.19
                                                        79.48
                 97.00
                             37.71
                                                        79.48
                 16.50
                             37.43
                                                        79.43
                                        . . . . . .
                 93.00
                             35.33
                                                        79.48
               16 .:
                             37.11
                                                        79.46
```

SITE DATA--CETATOS-Y GETATED

PROGRAM OUTPJT---

	X-ROTATED	Y-POTATED	ANG (F 10)	EXPOSURE
	1.11	35.11	[[.[0.42
	2.00	35.13	•	0.06
	3.00	35.11	(• . (L f	(• u .,
	4.50	35. ; 3		l • J ∪
	5.00	35.11	(• . ! . !	•42
	6.00 7.00	35.1)		.63 1.25
	5.51	35.11	u•(€.€ .•:	2.53
	3	35.3)	f • f (i f	3.82
	1 .00	35.32	i et lut	5.38
	11.7	35.1)	iostet	7.02
	12.5.	35.11	t • : : :	€.08
	13.01	35.33		1(.74
	14.00	37.13	6. 141	13.72
	17.3	35.11		15.11
	10.00	37.93	1.1.1	17.13
	17.94	35.13	tection	2(• 13
	18.76	35.13	i. tu	23.28
	19.17	35.13	3.	24.96
	2	35)		26.83
	21.01	35.03	10.5	28 • 5 -
	22.03	35.1)	1.1146	36 . 25
	23.73	35.13		31.21
	24.00	35.39	foctil	32.01
T1=	t • 3T 2=	· •		
	25.55	35.13	(32.45
	26.31	35.11		33 • 34
	27.00	35.11	Callet	33.64
	28.80	35.13	0.00	34.39
	29.00	35.11	ដឹងជាស្រាប់	35 • 15
	3:00	35.11	to to:	35 . 39
	31,35	35.33	0.3500	35.64
	32.00	35.77	£ o : C o C	35.78
	33.00	35.33	Course	35.92
	34.00	35.73	0.1000	36.06
	35.06	35.33	6.0686	36.06
	35.00	35.11	(.0096	36.06
	37.01	35.13	Section .	36.46
	35.52	37.77	1 · 1::(36.25
	39. 00	35.73	i • u i u l	3€ • ₺6
	k `• ^ (35.11	1.00	36.06
	11.11	₹ ~ \ 1		3€ • € 6
	42.	77.	t • ' +	36 • 16
	4.24	75,11		36 - 36
	dy to a	33.11		36.06
	1.5	35		35.06
	4 ₹	37.11		36.47
		33.11		37.33
	4 2	35.33	C. or C.J.C	35.19
	. 1 ⋅ 1	37.11	· • • · · · · · · · · · · · · · · · · ·	39.92
	F .	37.11	1.00	42.19
	Flo	35,11		44.73
	5.2 € € 2 €	35.12	correct	47 • 5 0

5 3.06	35.13	1.0000	50.22
54.00	35.33	Locket	51.93
55.26	35.03	* • *: : (53.64
53.00	33.33	· · · · · · · · · · · · · · · · · · ·	55.35
57.°C	35.11		55.51
F8.00	35.2)	to the	57.66
\$9.36	35.1	E • . 1 . u	58.46
€ 3.06	3=.73	2.01818	59.34
F1.50	33.93	inclut	66.10
65.00	35.11	7.00000	61.57
E 3. J.	35.11	u. tot	6: 81
6+076	35.J1	1.1.1	61.95
61.10	35.71	1.10	61.49
65.11	35.73	0.46.00	61.31
67.86	35.11	H	61.53
63.00	35.11	u of the	61.76
69.90	35.11	a a 1 Cut	61.98
7 . 50	35.11	1 •	62.23
71.75	35.1)	• 1	62.27
72.33	35.10	5.3000	62.33
73.24	3=.11	i • . t . t	62.82
74.35	35.00	10.1.1	63.36
75.00	35.31	Contat	64.58
75.30	35.13	0.000	65 • 87
77.00	3=.11	(• . (• (65.63
73.96	35.11	1.1661	68.49
79.25	35.13	5.0635	7t • 35
83.00	35.11	^•C(u(72.21
81.00	35.11	1. (1)	74.97
82.16	35.13	1.0511	75.74
83.00	35.10	100114	77.41
84.00	35.73	1.1618	79.07
85.01	35.11	1.101	79.36
85.30	3=.13	10000	79.64
87.06	35.11	1.0161	79.95
83.60	35.13	1613.0	86.16
89.26	35.13	1.1.6	80.35
91.30	35.21	i e sal	8: •43
91.00	35.11	1.00.11	86.43
92.3	35		89 • 43
93.33	35.		8: 43
9 . 33	3=.		85.43
97.71	35.31	. •	80.43
0- 1	37.15		8: 43
97.33	37.13	tock in	8: •43
59.00	77.11	• !	80 • 4 3
9 7	35.23	1.1.1	81 • 43
10.00	37.21		86.43

SITE DATA--CETATOS-Y CETATED

14 . 3	39.11	1: .	1
14. 3 13. 3 14. 3 15. 3 16. 3	27.1	11.0	1 1
53.13 24.15 53.13 81.13 78.13 18.13 84.13 69.13 10.13	3 F	1 .	1
53.13	+ 2	1 . i	1
81. 7	3. • 1	1.5	1
7	46.	100	1
18. 1	43.13	10.	1
55.13	L 10 - 1 ?	1.5	1
70.12 18.12 85.33 84.03 62.12 86.33	⊊ 3 , 11	1 (1
62. 😘	25.53	1 5 ∪	1
56.35	4 . 7 9	100	1 1 1
69. 1	د چ. د ک	1 i	1
27. ;	. 2.	1	1
10.0	9.	11.1	1
39• 😘	54.77	1	1
39. () 18. () 71. () 94. () 57. () 38. () 18. () 36. () 67. ()	18.37	10.	1 1 1 1 1 1 1 1 1
18. 3	2. 1	11 i	1
71. 3	54.	1	1
94. :	55.70	100	1
57.1	5: • 1	1	1
38 • 33	17.13	161	1
55.11	15.37	1.	1
55.11 18.15 36.18 67.13	1 3.	100	1
36. 3	2 : • ' !	14:	1
67.3	5.9•1.3	1 f e	1 1
47.13	14.77	3. C. C	1
6 !}	6.13	3.1 0	1
55.13 18. 5	22.11	100	1
18. 3	54, 1	1.0	1
83.15 75.13	59.19	1.0	1
75 • 13	F 8. ' 1	1.5	1
9: • 3	• 9•	1 4	1
75. 1	18.77	1.00	1
9~• 1	0.20	11.0	1
45. 2	24.	1	1
5	۷.	1. C.	1
83.25 75.33 90.03 76.33 98.3 65.3 89.3	23 43 44 4 2 4 4 2 4 5 5 1 6 5 5 1 1 1 2 5 1 6 5 6 7 6 7 7 7 7 7 7 7 7 7 7 7 7 7 7 7	1:000000000000000000000000000000000000	1 1 1 1 1 1 1 1 1
89	1.5.	1	1

PROGRAM OUT OUT ---

	X-ROTATED	Y-ROTATED	ANG(FAC)	EXPOSURE
	1.32	37.33 37.33	•	6 • 6 · 6 · 6 · 6 · 6 · 6 · 6 · 6 · 6 ·
	3.0	33.11		. • 2%
		₹₹ . ; ;		L • 94
	• • •	35.33	• • • •	• 42
	າ	37.13		.83
	₩ . 7 • 1 6	3. 10	•	1.25
	3.14	37, 33		2.53
	3.1	37.13	•	3.82
	1.**	35.13	•	F . 38
	11.	35.13	• 4 4	7 • . 2
	12.01	33.1)	1.0166	8.68
	13.00	3=	1.0	1.74
	10.30	35.11	. () (13.c2
	15.	35. ` `	• '.	15.1
	1 10 m	37.51		17.13
	17.8.	37.1	1.000	2 .13
	19,00	35.13		23 • 19
	19.5	77.1	· 1 *	24 • 95
	2	37.13	to a second	25.63
	21.10	37.11	San Algebra	25.54
	23.11	35.77		√ 3(• 2b
	23.00	35.13	• 171	31.21
	2	3=.11	i. i.	32.81
T1=	6.8T2=			
-	25.00	37.33	1.1.1.35	32 • 4 F
	25.30	35.3)	1.01	33.14
	27.1	35.13	1.00	33.64
	23.05	37.33		34.39
	29.31	3=.1)	6.188	35.15
	3 1.00	37.13	C • 16 16	35.39
	31.30	37.13	teresi	35 • 64
	32.00	35.11		35.78
	33,10	35.13		35.92
		35.13	0.145	
	3 - • 3 t			36 • 65
	3.0	35.11	• .	36 • 96
	30.01	35.11 35.11	. • . √ . €	36.06
	37.11	5- 1	5 • 1 f • 5	36 - 15
	38.05	3=.11	to the same	36 • 66
	39.71	37.33		36 • 06
	4 1 6 7 7	3.000	* • 1 * • 4	36.76
	41.11	35.11	• • • • •	3r 6
	42.	37.17	1.6	36 • სნ
	43.00	37.11	• • •	36. 6
	4 " ."	35.1)	1 • C	36 • 26
	47.00	35.11		3€•.⊎

			77 17
46.11	35.13	Colbet	36.47
47.06	33,11		36.92
	** `		27 76
48.5%	35.11	· 1 1	37.78
49.00	35.1)	1.01.65	38.64
\$ 10 € €	35.31	· · · · · · ·	4:.91
F1.30	35.33	1.66.66	43.17
52.5	37.15	• 111 • 111	45 • 14
£3.30			47.98
2 3 6 2 8	35.13		
\$ 5. PX	3=.73	1.0066	51.13
F 4.3 L			
	37.17	10.10.	52.27
53.07	35.33		54.42
£7.11	35.33		56.06
53.48	35.11	io sus Topius	57.71
		•	
F 9 . 01	35.13	' o s i u i	59.35
Eu. " "	37.11	* • * 1 . <i>f</i>	59.75
C 2 0 2		• • • •	
F1.6%	37.11	F 1	55.92
62.0.	35.13	· · Car	61.33
C 20 4 2	33. 13		
63.10	3		50 - 52
4 34			
€ 0.35	35.13 35.13	€ • • • • • • • • • • • • • • • • • • •	6. • 66
€1 • A €	35. 1	• (6i •5h
		•	
65.Ju	37. 11	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	6: • 88
67.1.	3~. 1	F.	61 • 11
	3 • •		
63.70	35.51		61.33
E 4	35. 1	a of the state of	6: •55
		•	
7 4.21	35.11	· • · • .	61.77
	• • • •	• •	
120.	<u> የ</u> ጠል ነፃ		61 . 3 .
72.70	37.37	1 4 1	51.91
	4		
73.1	35,11		62.33
74.00 77.00	35.7J	0. 606	62.87
		•	
77.3	35.31	2 · 3 · 2 · 2	64.16
75.00	35.11	to that	65 . 44
13103		U ● U E E E	
77.35	3=.17	, • f* { - f	66 • 23
78.00	37.31	to. Cat	68.05
79.00	35.00	10.666	69.92
	35.11	folts.	71.78
8 30			
81.00	35.33	ca the	73.64
0.2 0.3			
82.00	35.11	10000	75.31
87.26	35.11	1 - 1 6 2 2	75.98
84.00	35.11	6.0766	78 • 64
85.00	35.13	10.000	78.93
86.01	35.17	5 · • • • • • • • • • • • • • • • • • •	79 • 21
87.70	35.11	/ .	79.52
88.05	35.00	1.6 1.24	79.73
89.01	37.11	•	79.93
		• •	
9	37.11	• • • •	87 • 90
6 · · ·	3		8
2 L T		•	
Ç 2.	37.	•	5° • 3 3
93.3.	3/ . 11		8: • £ J
2301.			
g .i	3	• /	8. •
9	37,11	• 1	8' • • !
9		• ' •	
Q , N	37. 3	• • • •	81 🕠 🗅
9 ·	35.33		
A. • () (t • 1 •	80.00
¢ 3, ***	7	• ' ('	827
o 3.	77.37		8°•j
18 .)	37, 3		8 :
	-	· ·	

SITE DATA--X-ROTAFED Y-ROF17ED

14. 3	3 C. 13	1	1
53.43	27.	1	1
24. 3	30,11 27,11 36,11 42,11	11	1
53. 1	42.	1 .	1
81.3	3 1	1	1
71.53	4 2. 13 3 1 44. 13	1 :	1
18. 3	53. 7 44.73	1 i	1
56.,3	44.73	1 1.	1
81 . 7	25,11	1 %	1
62.)	3 4 1 1	1 1	1
56. 2	+ 2. 5 3	11 C	1
69.)	49.77	100	1
27. 3	5 2. 1	1	1
15. 3	# G* _ 3	100	1
39 • ∴3	+2.53 49.53 52.53 54.53 18.53 64.53 17.53 17.53 17.53 17.53 17.53	10.0	1
60.23	18.17	1.5	1
18. 3	2 }	1. "	1
71	B 4 . 5 3	1 (0	1
94.1	R.F. C.	110	1
57. 3	£ . • * 3	460	1
38 🕡 🧎	17.53	1 r	1
55.11	1 % - 3	1 :	1
18.33	1.3.10	166	1
36. : }	5 • 7 1	1 ,	1
67.3	59 . 1	11 6	1
47.10	59.1 16.19 6.19 22.19 54.73 59.18 58.11	1. e	1
60 •3	€∙10	100	1
55.3	2 2	1 7	1
18.30	54.73	11.5	1
83. ;}	59.11	11 6	1
76.::	58.11	100	1
97.13	49.	1	1
76 • 13	18.	1 -	1
5253 51 52 53	70.7.10.7.10.7.10.7.10.7.10.7.10.7.10.7	1	111111111111111111111111111111111111111
45		1	1
64.	2. 7 1(. 7 18. 7	1 i	1
66.	16.	1 .	1
89. ?	1 2	1. i	1

PROGRAM OUTPJI---

	X-0CTITED	Y-FOTATED	ANG(FAD)	EXPOSURE
	1.0i	33.11	Continu	0 • +9
	2.30	35.13	1.666	ូ. ប្
	3,10	33.11		€ • 33
	֥ 0 i	37.13	5.7155	3 • 3 €
	5.56	35.33	6. 166	• 42
	A. []	75.33	To tell	• 83
	7.35	35.31	fautuf	1.25
	8.10	35.11	1.3660	2.53
	9.00	3=. 13		3.62
	1	35.11		5.38
	11.90	35.11		7.32
	12.50	37.3	1 . 11	8.88
			(4)	10.74
	13.0u	35.11		13.62
	14.81	35.13		
	13.38	35.13	\$ 6 CH	15.10
	15.3.	₹ २३	(5 %)	17.18
	17.1	3 F • 3 1	. • i ku i	21.13
	18.00	37.00	5.00	23.18
	13.00	35.10	Faltur	24 • 95
	2 1 • 7 !	35.71	* • • • · · · ·	26 • 83
	21.85	37.11	in the	28.54
	22. 1	35.11	5 . 1 . 1 . 1	3*•26
	23.00	39.93	i er	31.21
•	24.16	35.13	(32.11
T1=	0.3T2=	• *		
12-	25.00	35.33	1. 1. 1. 1	32.45
	25.33	35.11	1.36.6	33 • 84
	27.30	35.11	1037	33.64
		35.13	6. 835	34 • 39
	2 8.00			35.15
	29.00	37.73	1.1100	35.19
	330	35.13	bestel	
	31.33	35.11	ត្តព្រះប្រជុ	35.64
	32.Ja	35.11	boufit	35.78
	33.90	35.13	្រូបដ្ឋា	35.92
	34.00	3= 13	i out it	36 • 56
	35.00	37.31	Tactus.	36.06
	35	37.00	to the state of	36 • ≎6
	37.33	3=.13		36.56
	33.39	3 11	• * •	36 • 36
	33.10	35.11	5 • 1 · 5 · 6	3€ • ∪5
	4 . 1 .	35.19		36 . 16
	L 1.	37.11	F . 15	36.06
	42.00	37.17	•	36.46
	43.01	3 . 11	•1.(1	30.15
	L	3		35 . 16
	47.30	35.1)	• . • * . l	36.06
	يل و ک	2.20 (1)		30 0 0

46.00	35.11	1.00000	36.47
67.00	₹≈.13	1	36 • 92
40.11	35.:1		37 • 36
49.00	35.13	1	37.81
5 . 60	35.13	6.00	39.61
51.00	35.11	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	40.22
F 2	35.73	•	41.64
53.00	35.11		43.07
5-1-13	35.11		41, . 34
57.00	35.11		45 · 61
5 10 U L	37.73		46.87
15.00			47.29
57.03	35.11	1	47.71
53.00	3=.13 35.13	/ ,	47.97
F 9 . 5 3	37.		
E •3:	35.11	• • •	48.17
E1.5"	35.11	A Company	48 • 24
62.01	3=. 13	to. fut	48.53
€3.00	37.12	1 • . 1 1 2 • 2 10 6 1	48.53
€4.0E	3=.33	V • 17 12 € 1	48.53
€1.•04	3F • 3J	4. N. 18 9	48.53
66.1.	37.13	s · · · Cul	48.75
F 7 . 34	37.01	• • •	43.97
€3.15	37.13	tolet	49.19
€3•1°	35.13	6.66	19.42
720	35.33	i •! ^u! • ! !!!	49.64
71.00	37.73	• tat	49.71
72.50	35.33	1.000	49.77
73.00	35.31	7.1041	50 • 25
74.10	37.03	Sankle.	5(.74
75.56	35.73	folkut.	52.52
75.36	35.21	i • i f u i	53.31
77.56	35.73	r • " t c c	54.07
73.00	35.31		55.93
73.6.	35.31	· · Lui	57.79
30.08	35.13	C. H.C.C	59.65
61.00	35.11	0.000	61.51
82.06	35.11	5.20gt	63.18
03.88	35.31	i wii i ni	64 • 84
84.35	35.11	8.0166	66.51
85.00	35.17	5-65-66	65.79
85.00	35.13	Cockut	67 • 48
87.00	35.33	 € (1 € 1 €) 	67 • 39
85.50	35.11	. • t* ∪ t	67.59
83.13	37.11	• - • - •	67 • 8 J
9 . 11	3=.13	io tet	67.87
C1.;	37.1	• 1	67 • 87
92.11	35.31	. • . • • •	67 • 87
03.00	5-632	. • • • •	67.37
9 4 . 62	3= 1	6 • 1 · 1	67.87
¢ 👝 🔭	7.	• · () i	67 • 57
ç.	37.1	• 1 •	67 • 37
91.1	₹*•:*	• t	67.87
Çy, T	70.00	• • •	67.87
c a ,	27.1	•	67 . 07
11 . 1.	22,13	1.0000	67.57

Appendix C3
Automatic Model Output
(see Table 3, 1 KM Column)

OPTIVE TIGHT PATH INDUT SUMMARY---

PK DATAEACH SITE TYPE SITE TYPE 1.1 SITE TYPE NO = 1 COLUMNS=RINGS, ROWS=SECTORS 133

SITE DATA--CETATOR-X CETATED

16. 3	39.13	1 i	1
14.3 53.1	27. " ?	1 (1) 1 (1)	1
	35.11	1	1
53	42.13	1	1
81. 3	3: • 1	1	1
7:. :	40. 7	1 .	1
18. 3	k 3. 3	1 -	1
5 F . 3	61.77	1.5	1
F3. 12 24. 11 53. 13 81. 13 18. 13 62. 13 56. 3	5.3.1	10.	1 1 1 1 1
62. 3	25.11	1	1
56. 3	27.51	1	1
69	49.33	1 .8	1
27.3?	E 2.11	1 .	1
15. 3	4 C	1.4	1
39. 3	54.13	1.1	1
55.3	18.73	1 t	1
24	2• * *	1	1
71.13	51.77	1.0	1
94. 3	5 F . 7 3	1 .	1
57.3	5 • 1	1 , 5	1
38.33	17.	1.4	1
55.43	15.31	1 1 G	1
18.3	13.33	150	1
35 1	2 • 1	1.1	1
67.3	50.1	11 e	1
47.33	10.13	1 5.3	1
E 33	ۥ13	11.5	1
55.33	22.11	11 e	1
18.33	54. 3	1 i u	1
83.1	59.11	10.5	1
75 - 1	58.17	1 0.0	1
913	49. it	17.6	1
76.3	18. 1	1	1
36.1 47.1 47.1 55.3 16.1 55.3 16.1 75.1 91.1 75.1 94.1 65.1 65.1 65.1	327 U 3 U 3 C 2 U 5 U 5 U 5 U 5 U 5 U 5 U 5 U 5 U 5 U	1	111111111111111111111111111111111111111
45.73	54.11	1 : :	1
6. • 13	2. " 3	11.0	1
68.3	15.7	1	1 1
89 - 13	18.11	1 1 i	1

PROGRAM OUTPIT---

1.00	
3.02	
5.72 32.5+ 4571 6.0 6.01 32.15 4571 6.0 7.20 31.55 4571 2 9.11 31.7 4571 2 9.11 31.7 4571 2 1.30 31.33 4571 .5 11.01 29.53 4571 .5 12.03 29.11 4571 .5 13.30 28.51 4571 .5 14.01 28.51 4571 .5 15.10 29.51 4571 .5 16.11 28.12 4571 .5 17.03 28.12 4571 .5 17.03 28.12 4571 .5 17.03 28.21 .7914 .9 18.01 28.73 4571 2.6 21.10 28.73 4571 2.6 22.07 28.73 4514 2.9 23.33 .4914 2.9 23.34 28.73 4514 2.9 25.16 2	
8. 61	j
6. 6	•
7.17 31.55	:
9.6. 3 .574.675 11.00 31.038.75 11.00 20.538.75 12.02 23.114.575 13.00 28.514.575 14.60 28.514.75 16.10 28.514.75 17.00 28.514.75 17.00 28.514.75 17.00 28.614.75 17.00 28.734.75 19.00 28.734.15 200 28.734.15 200 28.734.15 200 28.734.15 200 28.734.16 200 28.734.1	,
9.6. 3 .574.675 11.00 31.038.75 11.00 20.538.75 12.02 23.114.575 13.00 28.514.575 14.60 28.514.75 16.10 28.514.75 17.00 28.514.75 17.00 28.514.75 17.00 28.614.75 17.00 28.734.75 19.00 28.734.15 200 28.734.15 200 28.734.15 200 28.734.15 200 28.734.16 200 28.734.1	2
1 . 30	•
11.00	L
12.03 29.11 1571 .5 13.00 28.51 4570 .5 14.00 28.51 100 .5 15.10 29.51 100 .5 16.10 28.12 4170 .5 17.20 28.21 .7910 .9 13.30 28.31 .6571 2.6 2.00 28.73 610 2.6 21.30 28.73 601 4.9 23.30 23.33 .4914 5.7 23.30 23.33 .4914 5.7 23.30 23.33 .4914 5.5 24.30 23.37 .4870 8.1 25.30 23.33 .4914 5.7 23.30 23.33 .4914 5.5 25.30 23.55 .914 9.8 27.00 23.55 .914 9.8 27.00 23.55 .914 9.8 21.00 37.14 .607 9.8 21.00 37.52 .3856 10.4 33	3
13.35 28.51 4576 .5 14.56 28.51 4576 .5 15.76 28.51 4576 .5 16.76 28.12 4176 .5 17.30 28.21 .7916 .9 14.37 28.31 .4916 1.4 19.30 28.73 4571 2.6 21.30 28.73 451 4.9 22.37 28.73 451 4.9 23.30 23.33 .4914 5.7 23.30 23.33 .4914 5.7 24.31 28.73 451 6.5 25.30 23.33 .4914 5.7 24.31 28.73 .4576 8.1 25.30 23.33 .4914 5.7 25.31 29.37 .4576 8.1 25.31 29.37 .4576 8.1 25.31 29.37 .4576 8.1 27.60 23.37 .52 .914 10.2 27.60 27.50 .914 10.2	3
15.16	3
15.16	3
16.1.	
17.29	ۏ
14.31	3
19.70	1
2 . f n 29.73 f l 3.4 21.11 28.73 l l 4.9 22.07 28.73 l l 4.9 23.33 23.33 914 5.7 24.31 23.37 614 9.5 25.30 23.47 4571 8.1 25.30 23.55 914 9.5 27.50 23.55 914 9.8 23.30 33.14 4071 10.2 23.30 33.14 4071 10.2 23.30 33.14 4071 10.2 23.31 33.23 914 10.2 33.31 33.23 914 10.2 33.31 31.31 36.56 10.4 33.31 31.31 36.56 10.5 34.32 31.37 18.28 10.7 35.31 31.32 18.28 10.7 36.31 31.32 18.28 10.7 36.31 32.11 18.28 10.7 36.31 32.21 18.	3
21.32 28.79	
22.00 28.73 00.0 4.9 23.00 23.33 .0914 5.7 24.30 28.37 .0514 6.5 25.30 23.47 .4576 8.1 25.30 23.47 .4576 8.1 25.30 23.55 .0914 9.8 24.30 33.14 .6076 10.3 29.30 33.23 .914 10.2 29.30 33.23 .914 10.2 33.00 33.23 .914 10.2 31.05 31.71 .36.56 10.9 32.10 31.37 .1828 10.6 33.31 31.37 .1828 10.7 39.30 31.74 .1828 10.7 39.30 31.74 .1828 10.7 39.31 31.32 .1828 10.7 37.31 32.31 .1828 10.7 36.31 32.21 .1828 10.7 36.31 32.23 .1828 10.7 36.31 32.23 .1828 10.7	2
23.33	
24.31 28.37 .1514 6.5 25.70 23.47 .4571 8.1 25.70 23.55 .914 9.5 27.70 23.55 .1914 9.8 21.70 37.14 .4571 17.3 29.71 37.23 .914 10.2 31.71 35.52 .35.56 10.4 31.71 .35.56 10.5 32.56 32.11 .16.28 10.7 36.51 31.74 .18.28 10.7 36.51 31.32 .18.28 10.7 36.51 31.32 .18.28 10.7 37.01 31.32 .18.28 10.7 36.31 32.21 .18.28 10.7 36.31 32.23 .16.28 10.7 36.31 32.23 .16.28 10.7	4
25.10 29.55 .914 9.8 27.00 29.55 .0914 9.8 21.30 31.14 .4076 10.0 29.30 31.23 .7914 10.2 33.00 31.52 .3856 10.4 31.30 31.71 .3856 10.5 32.66 31.37 .1828 10.6 33.31 31.37 .1828 10.7 34.00 31.75 .1828 10.7 39.30 31.74 .1828 10.7 39.30 31.74 .1828 10.7 39.30 31.74 .1828 10.7 39.30 31.74 .1828 10.7 39.30 31.74 .1828 10.7	
25.10 29.55 .914 9.8 27.00 29.55 .0914 9.8 21.30 37.14 .6576 17.3 29.30 37.23 .7914 10.2 31.00 37.52 .3656 10.4 31.00 31.71 .3656 10.5 32.66 31.73 .1628 10.6 33.31 31.37 .1628 10.7 34.00 31.75 .1628 10.7 39.30 31.74 .1828 10.7 39.30 31.74 .1828 10.7 39.30 31.74 .1828 10.7 39.30 31.74 .1828 10.7 39.30 31.74 .1828 10.7	
24.30	
29.30 37.23	
31.00 31.52 .3856 10.4 31.00 31.13 .3656 10.5 32.10 31.13 .1628 10.6 33.01 31.37 .1628 10.7 34.00 31.75 .1628 10.7 39.00 31.74 .1828 10.7 39.00 31.74 .1828 10.7 39.00 31.74 .1828 10.7 39.00 31.74 .1828 10.7 39.00 31.74 .1828 10.7	
31.05 31.13 .3656 10.5 32.16 31.13 .1628 10.6 33.16 31.37 .1628 10.7 34.06 31.55 .1628 10.7 39.30 31.74 .1628 10.7 39.30 31.74 .1628 10.7 37.31 31.32 .1628 10.7 37.31 32.11 .1825 10.7 33.31 32.23 .1725 10.7	
32.16 31.13 .1828 10.67 33.16 31.37 .1828 16.7 34.26 31.55 .1828 10.7 39.30 31.74 .1829 16.7 35.50 31.32 .1828 10.7 37.30 32.11 .1828 10.7 33.31 32.23 .1728 10.7	
33.31 31.37 .1828 16.7 34.00 31.55 .1828 10.7 39.30 31.7+ .1828 10.7 36.51 31.32 .1628 15.7 37.51 32.11 .1828 10.7 33.31 32.23 .1728 10.7	
34.26 31.75 .1828 10.7 39.30 31.74 .1828 16.7 39.30 31.32 .1628 19.7 37.31 32.11 .1828 10.7 33.31 32.23 .1728 10.7	
39.30 31.74 .1828 18.7 39.50 31.32 .1828 19.7 37.50 32.11 .1828 19.7 33.30 32.23 .1728 18.7	
36.51 31.32 .1628 15.7 37.01 32.11 .1828 10.7 33.31 32.23 .1/28 10.7	
37.5. 32.11 .1825 10.7 33.3. 32.23 .1/25 10.7	
33.34 32.23 .1/25 10.7	
33, 34 37, 23 41/25 15 46 7	
	d G
_	
h .) 32.55 .1825 117	
41.00 37.35 .1826 10.7	
42.3, 37.33 .1825 1°.7	
47, 7, 77, 72 ,1828 16.67	
u.,j. 77.+7 .1828 1(.7	7

45.00	33.53	•1 t 2 s	11.479
46.78	33.37	• 27 + 2	11.02
47.33	34.75	.1128	11.65
43.00	32.44	.3656	12.29
49.36	34.3+	(916	13.16
5 1.33	34,25	0914	14.56
51.50	34.15	44	15.76
5 2.06	34.37	91/	18 . 7.4
53.36	3+.15	914	19.75
	34.07	514	21.98
F 3 6 2 1	37.33	011	22 • :2
F-12 + 7 L	37.33	+ + 1, 6 F	23.17
57.32	33.43	4571	24.61
50.54	37.77	.271.2	25 • 24
53.30	33.35	•1828	25.82
E . 01	3 + • 1 +	.1828	25 . + 0
E1.30	34.52	.3696	26.61
£ 2. 45	34.51		
		• L 91:	25.81
E 3.36	3 + + 73	4.514	25.53
€ -• 30	3 • • * 3	• . 91!	25. 94
67.31	35.33	915	25.94
E6.15	34.33	914	27 • 17
67.32	3 - + 3	4571	27.23
Esals	₹₹ . 33	- • ≥ 57 .	27 • 23
69.00	33.13	•1526	27.46
7:.5.	37.53	4571	27 • 46
71.75	33.+1	- 27 - 2	
			27.46
72.50	37.12	2742	27.87
73.00	32.3+	1828	28 • 29
74.86	32.55	2742	29.57
74.35	32.33	2742	31 . 86
75.31	32.13	2742	
			32.42
77.00	31.57	- • 457 i	33 . 98
73.00	31.42	1828	35.49
79.35	31.23	1828	37 • - 3
87.13	31.15	1828	37.74
81.[[31 • +3	• 3f 5 6	38.39
82.11	31.71	.2742	35.83
83.50	31.31	.1825	39.42
84.39	32.23	.3656	46.02
85.00	32.77	.4571	46.77
85.00	32.35	.1828	41.53
87.06	3₹•3+	•3€5€	41.77
88.00	33.72	. 3€5€	42.62
89.00	34.11	. 3F SE	42.16
9	34.51	• 4.F.7.L	
		• ~: 7 %	42.16
91.1.	37.03	**** 7.	42.16
92,55	31.13	- · ` \ . '	42.16
93.50	3~• * 3	- • ' . :	42.16
9~, []	37.,3		42.16
	77.13		
C(), ()		+ · · · / ·	42.16
9 7 • 2 1	22.13	= • • • • • •	42.16
0 1	۲۳ ۰۱3	(c)	42.16
93.00	7	- , t	42.16
و رو و و	7 1	- ()	42.16
1(+1.	3 - • 51	• : ₹ 7 .	42.16

SITE DATA--CETATOS-Y CETATED Y-ROTATED

14. 3	3 9, 13	1 / L	1
53	27 1	1	
24. 1	30. 1	1 1	1
53.01	4 2.	1 .	$\overline{1}$
81. 3	3 . • •	100	1
7: . :	نهادي الم	1 (1
18. 3	→ 3• 3	1.	1
562	44.00	1 .	1
84	43. 1	1 3	1
62.11	25.	1 '	1
56. 1	3 1 . 5	1	1
69.	+ C. 11	1.9	1
27. 1	3 2. * *	1	1
15	5 Q j	1	1
39	56, 31	1	1
6: • "]	18.11	1 10	1
18. 7	2. ' '	1.	1
71. 3	54.12	1. ti	1
94.13	55.11	100	1
57.13	5 1 1	1 .0	1
38.11	17. 3	1 9	1
55.13	15.03	1.3	1
18.33	1 3. 3.7	1.	1
35 • / 2	2 . 1	1 6	1
67.1)	59.15	1 ែ ម	1
47.30	14.7	11 7	1
€1:•13	6.50	1. t	1
55.1	22. * *	111	1
18.J)	50.07	17 to	1
83.11	59. 1	1 00	1
76.1	58.7	1	1
96.3	4 G. ^ J	100	1
75	18.11	1 i.	1
15258715.1131131131313131313131313131313131313	23 + 3 + 3 + 3 + 3 + 3 + 3 + 3 + 3 + 3 +	1/0 1/0 1/0 1/0 1/0 1/0 1/0 1/0 1/0 1/0	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
453	54.0	1	1
80.)	2. 1	1	1
E5.)	16.	1 .	1
89.13	18.73	17 t	1

PROGRAM OUTPJT---

X-ROTATED	CETATCH-Y	ANG (FAT)	EXPOSURE
	-		
1.50	34.51	7	
2.11 3.11	34.12	4571	ا فر ت 🔹 🖰
4.29	33.52 37.13		5 - 7 J
5.25	37.5+		f.jq 0
5.01	32.75	4.71	
7.00	31.55	4571	0.00
9.33	31.17	- 4571	. 22
9.00	3 . 5 7	- · · · · · 7 :	4
1 .50	31.13	4571	•51
11.50	23.53	- · 457:	•58
12.31	29.13	- • • • 7	• 58
13.00	29.51	4571	• 58
10.00	28.51	*	• 5 8
15.30 16.30	23.51	11. 17.	•50
17.01	2°•12 2°•21	• 914	•53 •99
13.50	28.33	915	1.41
19.00	29.73	457:	2.69
2 1.53	23.73		3.46
21.00	23.79		4.22
22.50	23.73	1.11	4.98
23.50	2 P . 33	.3916	5.74
24.01	23.97		5.53
25.00	23.47	• • • 71	8.14
25.03	29.55	• ,51t	9.58
27.38	23.55	• . 914	9.80
23.00 29.00	31.1+ 31.23	•4571 •1914	19.52
33.33	31.52	•3656	10.23 10.45
31.53	31.33	•3656	10.52
32.35	31.13	.1828	19.66
37.03	31.57	.4575	16.72
34.06	32.1"	. 57	10.72
35.0≥	32.35	•1828	10.72
35.01	32.5+	• ∡ ≿ 2 €	10.72
37.00	32.72	•1 5 2 8	10.72
33.01 39.11	32.91	•162£	15.72
43.60	37.13 37.23	•1828 \\$36	10.72 172
4 1 . 2	37.45	•1828 •1828	10.72
42.5	33.55	•1628 •1628	10.72
63.7	37.33	•1625	1).72
44.	39.12	.1828	172
4.5.3.	34.21	.1629	172

46.33	3+.53	•4576	16.72
47.01	34.73	• 91L	11.17
43.33	34.33	. (16	12.3
4 - 0 0	34.37	. 914	12.69
50.00	35.)5	.0914	14.63
51.00	37.15	• 611	16.38
F2,32	35.24	• 91 6	19.34
53.37	37.24	- • It w.	2 31
		• 4 .	
5 · • 3 0	35.2+	164.	22.46
E	₹5.2+		26.63
		• • • • •	
55.0C	35.24	- • · · · · ·	26.74
		• •	
57.50	34.75	57'	29.06
E3.33	3 3+	• 6 9 1 4	36.33
53.53	34.33	•.91:	31.9+
	35.12		
€ - 6 U C		.1828	32 • 32
E1.35	35.51	• → 5.7 "	32.63
E2.00	35.33	2742	32.72
E3.3.	38.35	2742	32.77
E Hart	37.1+	· 914	32.91
65.30		914	
	3=,23		32.91
E 5. 3)	39.32	•. 911	33.13
67.00	34.33	- · L 7 7 '	33.19
E 9.00	34.34	1571	33.26
69.00	37.35	57:	33.25
	33,35	7 (33.20
71.35	33.35	.457.	33.25
72.85	37.57	 27≈2	33.68
73.50	37.23	2742	34.09
7∾.00	33.33	27.42	35.38
75.30	32.72	2742	35.06
F 50 F J J			
75.00	32.++	2742	38.22
77 65			
77.00	31.35	5570	39.78
78.35	31.57	2742	41.29
79.01	31.33	2742	42.81
30.00	31.11	2742	43.54
81.00	31.443	.3696	44.19
82.50		3656	44.84
	31.11		
83.00	31.29	•1828	45.44
84.00	31.73	•4576	46.33
85.10	32.27	.457[45.79
85•₿[32.77	•6571	47.54
87.00	37,25	• ~ 5.75	47.79
88.01	33.54	• 3656	48.13
89.01	3+.12	•36 à5	48.17
9	34,51	· 327.	48.17
91.4	77.11	•457£	48.17
92.	37.11	:	48.17
	2 U Z L		
93.66	35.1		48.17
94.3	33.11		45.17
	2 * • 1	→	
o . ^ :	35. 1	- • • • • • •	45.17
	3= -1		
90.0h		• • • •	49.17
0,	33.11	- (48.17
Ç. F. J. J.	37.01	~ •	46.17
çn,;;	7"."	• •	48.17
		•	
10:00	731	 ±?	48.17

SITE DATA--CETATOS-Y CETATED

14. 3	39.11	1.	4
93.1	27.1	1.0 1.0 1.0 1.0 1.0 1.0 1.0 1.0 1.0 1.0	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
24. 3	35.53	11 0	1
53.1	2.	1	4
81. 9	3/400	1.6	1
7 2	81. 3	111	1
18.3	43.1	1.1	1
56.00	4 L	1.	1
86.13	43.33	100	1
62. 3	25.	1 6	1
58.33	32.57	100	1
69. 1	49.11	11	1
27. 1	5 2. 11	1	î
15. 3	49.	1:	1
39.33	43.77 43.77 25.77 45.77 52.77 54.77 54.77 54.77	1. 0	1
5. · 1	18.	1 1	1
1E • 23	2.1	1.1	ī
71.00	54.15	1	1
90.0	55.73	1::	1
57.3	5 " 3	11.9	- 1
38 • }	17.00	11 3	1
55.3	15.00	110	1
18.3	13.77	1	1
36.01	24.53	101	1
67.1	59 . *)	100	1
47.03	14.77	135	1
6(.*)	F. ^ ?	1 0	1
55 · }	2 2. 3 3	150	1
18.73	54. : :	1()	1
83.1	59,1)	136	1 1 1
76 . 3)	5.8• " "	1.0	1
90 }	¥ 9. 3 3	1.;	1
75.3	1 %. 1	1	1
9++3	5.2.13	1	1
343	23.00.00.00.00.00.00.00.00.00.00.00.00.00	1, . 12, 11, 11, 5 11, 6 11, 6	1
6.03	2.	1 5	1
65.1	16.71	11	1 1
89.1	18	1. 4	4

PROGRAM OUTPUT---

X-ROTITED	CETATES Y	ANG (FAD)	EXPOSURE
1.16	34.51	4577	J • 6 7
2	3+ • ∶2	→ • 1 1 1 1	. • : :
3. ₽	37.52	- 4571	2.0
4 🕳 🗦 💆	33.17	44571	6.23
9.53	32.5+	55.74	0.91
6.01	32.35	57:	0.00
7.35	31.75	4 E 71	د له و د
a . t .	31 7	1571	. 22
9.00	31.57	7'	• • • • •
10.	3 • 13	1.7	.51
11.5:	23.33	- 4571	• 58
12.33	22.11	6.5.7	•58
13.35	23.51	()	•53
14.52	23.51	- • • •	•58
10.00	29.51	-	8 ذ ٠
10.19	22.12	7	•58
17.	29.21	. 514	. 99
15.20	23.31	91-	1.41
19.05	20.73	. 671	2.69
2	23.73	-	3.45
21	20.73	-	4.22
22.20	23.73	:tu:	4.98
23.13	28.33		5.74
24.88	29.97	914	6.59
25.30	23.47	• • 57	5.14
25.30	23.55		9.58
27.30	29.55		9.85
28.10	3*•1+	457:	18.62
29.50	3:.23	• 51 6	10.02
	37.52	•3656	10.45
31.21 31.21	31.33	•3656	
		• 3 6 2 6 • 1 6 2 6	10.52
32.00	31 • 13		10.66
33.90	31.57	•4571 •4571	10.72
34.00 #8.30	32.17	•51 ()	16.72
35.33	32 • 25	. 011	16.72
35.55	32.35	611	10.72
37.55	32.44	. 616	12.72
3 3.10	32.53	• L 5 1 4	10.72
39.00	32.1.	457	10.72
b . • 31	32.13	• S1-	10.72
41.3.	3?.??	• 61 L	172
42.1	32.32	. 61-	15.72
43,10	32.+L	• <u>61</u> .	10.72
4 · 1 2 c	32.51	• 91 :	15.72
63.35	32.57	911	11.14

46.30	32.37	.3656	11.35
47.30	32, 17	.7511	12.
1.9.1	33.+5	.3156	13.45
			15.32
49.33	33.45	-outet	
53.53	33.45	t.i	17.59
F1.31	33. + 5	Tout	213
52.24	37.+5		22.68
53.00	37.45	- 1 1 61	25.17
5 1 6 5	₹₹.45	-,	26.97
57.30	33.+5	Cul	27 • 93
55,00	33.+5	-	28 • 97
57.33	32.35	- 45.571	31.39
54.30	₹₹. 24	.2742	31 • 45
59.1	37.52	.27.2	32.41
6	37.31	.27 +2	33.37
£1.10	34 • 13	• 38 5 E	34 • 19
62.01	34.23	. 1 C14	34.5
63.14	34.37	• . 5.1 ×	34.81
E+. C.	34.45	. 914	35.2
E	34.55	916	35.15
63.01	34.5+		35.16
			35.16
67.81	35.15	4. 7. 5	
63.70	33.55	157	75.16
€ 94 5 L	33.3+	.1828	35 • 16
735	33.35	457	35.16
71.00	33.3+	• +57.	35.16
72.35	33.55	27 - 2	35.57
73.50	33.23	2742	35.99
74.00	33.11	2742	37 • 27
70.30	32.72	2142	38.55
75.31	32.++	2742	40.12
77.35	31.35	4176	41.68
78.00	31.55	2742	43.19
73.20	31.33	2742	44.7.
8 3 . 3 3	31.1)	2742	45 . 44
			45.39
81.30	31.43	.3(55	
82.50	31.77	.2742	46.68
83.60	31.35	.1828	47 • 23
84.00	32.1+	.1828	47 • 57
85.21	32.53	.457	48.63
85.01	33.31	.3655	49.38
87.6[37. 73	.3656	49.53
88.30	37.73	.3656	49.87
84.0r	34.15	.3656	51.61
9	31.55	•4E7i	51
01.7	3 - 1+	•4576	5031
92.00	75.14		5' 461
93.3:	₹5.14		50.01
9 (1.1)	37.14	- • •	5
91.	33.1+		5 1
93.34	35.14	- • • • • •	57 • 51
97.J	35.1+	- ← () . (51
99.51	37.15	51"	5 .01
93.31	35.15	-,	5 1
10	355	-, 7	5 • 11
_ , _ , ,		· ·	

SITE DATA--CETATOS-Y CETATED

14.]	3c, 1	1 0	1
F 3 4 3	27.	11.6	1 1 1
26. 3	35.13	1 . e	1
24.] 53.] 81.] 71.] 18.] 56.] 62.] 56.] 56.] 19.]	+ 2. ^ *	1 " 5	1
81. 3	3.,	1000	1
712	\$ 4. °)	1.5	1
18. 3	4 3 . 1 3	1:5	1
55.33	40.73	1 " -	1
84.)	4 3. 3	11. "	1
623	2 10 - 1 1	11.6	1
56.3	35.13	100	1
€9• :	4 C 1	17.6	1
27.)	52.13	1 '	1
15 1	49. 1	1. i	1
39. 1	54. 11	1 🕀	1
E5 . 3	18.77	16.6	1
18. 3	2 1	1 .	1
71.3	54, 17	170 4 170 4	1
94.13	8 5 6 7	1	1
57.3	5 _ 1 1	175	1
38.13	17.11	1()	1
55, 13	15,00	1. 7	1
18	13,13	100	1
36.73	2 (• 7 7	100	1
1 ن • 67	59.33	116	1
47.33	14.23	10 b	1
60.13	6.33	100 100 100 100 100 100	1
55.11	2 2	1. 0	1
18 . 59	81.01	1:6	1
83. 😘	59.11	10%	3
76 3	3 8. " . "	1, 9	1
90.03	4.9.03	1	1
76. 3	18.73	1 ()	3
94 . 12	5.2.33	100	
53.03 53.03 53.03 55.03 56.03 56.03 56.03 56.03 56.03 57	323+344423455454 - 585113251777733177732777343442354545455511325463777777331777	1() 1() 1()	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
6L.:	2. 13	101	1
68.13	100	1.	
89. 1	18.7	171	;

PROGREM OUTPJI---

X-ROTATED	CELTICA-A	ANG (FAD)	EXPOSURE
1.31 2.07	3 + • 51 3 + • 72	5 F 7 5 - 6 5 F 7 5	
	344 12		4 . 89
3 [37.52	- • ÷ f 7 ·	
⇔ ଶୁଣ	37.15	4571	u • ⊎ 5
ა•ქ.	32.5+	571	6.03
n • . ′	72.15	1 7 7	1.00
7.57	31.55	17"	1.00
3.30	31 . 7	 7	• 22
9 ∶	31.57	571	• 44
1 1.77	3 🔒 3	- •4575	•51
11	29.53	57:	•58
12.00	23.13	457	•58
12.00 13.00 14.00	23.51	- · · F 7 :	•58
1	20.51	-,	, 5 ∂
1 3	20.51	- • (•58
10.0	23.12		•58
17.30	23.21	. 914	. 99
13.1	23.73	. 911	1.41
19.01	23.13	-57:	2.69
2	22.73	* • i es	3.46
21.0i	22.73		4.22
	23,73	~ • C 1 / C	4.98
22.36			5.74
27.03	22.33	014	5.59
2 0 2	23.37	. 914	8.14
25.30	29.47	.457i	
25.34	23.55		9.58
27.30	22.55	• (514	9.83
29.30	3 . 1+	•4571	10.32
29.30	3. • 5 ₹	. 614	10.23
3 : . 🗓 u	31.52	•3656	10.45
31.Ji	31.11	• 3656	10.52
32.00	31.13	•1828	11.66
33.51	31.57	• 457,	15.72
3 4. 0 5	32.17	• € 57.	13.72
39.30	32.25	. (14	15.72
34.01	32.35	. 914	10.72
37	32.44	• 64t	16.72
34.11	32.31	. 61%	1:.72
39.10	72,34	7	10.72
<i>i</i>	32.17	C 11	172
41	32.22	91	10.72
42.00	72.32	. (1)	11.72
	32.41	. 51-	172
43.31	32.53	.1628	10.72
Acceptance	37.14 37.13	•1828	1.94
L. C.	3 (• 1 3	• ¥ C & C	J ● 3 †

45.90	37.15	.3656	11.17
67.17	37.23	6. C1.	11.81
43,31	33.5+	•3f5€	12.36
43.00	33.5+		15.13
F .34	33.54	- • 1€ -1	17.39
51.38	32.5+		19.94
52,16	33.73	• C.1 k	22 • 48
£ 3, **	37.77	- 1 4 5	24. • 98
53.11 54.11	33.73	- € . t	25.77
Fo. 16	77.73	- • • 1 € 4	27.75
56.00	33.73		28.73
57.3	33.34	3656	29 • 94
53.71	37.53	.1826	31.32
53.18	37.71	.1828	32.28
E . 30	33.33	.1528	33.39
£1	34.23	.3(56	34.22
62,30	34.3	91 k	34.53
£ 3. ± 3	34.45	C 1 4	34.84
	34.35	611	35.04
6 (4.33	34.53	511	3F • 18
€ 5.31	34 • 7 +	91	35 • 18
66.36	34.25	57.	35.13
67. ((37.75	L F 7 .	35.13
61.71		1825	35.18
E 9	37.9+	157	35 • 18
70.00	37. +5	.457	35 • 18
71.55	37.9+	2"42	35 • 6 1
72.10	33.55	2742	36.02
73.00	37.33	2742	37 • 3)
74.50	37.11	2742	38.58
75.36	32.32		40.14
76.00	32.53	27 42	41.71
77.30	32.1+	575	43.22
78.00	31.35	1828	44.73
79.30	31.77	914	
8 1. 36	31.57	- • ; <u>\$</u> 1 4	45.42
81.33	31.57	3bc.	46.12
82.07	31.35	.1828	47 • 27
83.00	32 • 2 +	.3656	47.57
84.00	32.53	.3656	48 • 26
85.13	33.12	• 4571	49.24
83.00	33.43	.2742	49.71
87.00	33.53	.2742	50.02
83.01	33.35	.2742	55.22
89.JC	3++24	• 27±2	F(• 36
9 " • 3 :	34.73	.457:	51 • 36 51 • 36
91.12	37.23	•4571	_
92.[37.23	-, ()	5 • 35
93.00	33.23	• • . 5 11	50.35
9 1.31	37.23	- • • • • • •	5 . 36
93.51	37 • 23	16	5° • 36
9	33.23	-•	5 . 36
97.10	37.13	- • · §1 ·	5: • 36
93.	35.13	- · (. i	51 • 35
93.0.	3″ • - •	-, 91 %	5° • 3°
1t • 01	3 5 5		5 36

SITE DATA--X-ROTATED Y-ROTATED

14.3	39.00	1), i	1
53. 3	2", 1	100 100	1
26 3	35.07	194	1
53.41	+ 2. 1)	1: 1	1
81.3	3 (- 1	101 110 110	1
7 3	46.11	1	- 1
18. 3	4 7. 1 t	1	4
18.73 56.73	400	1 (4
84. 3	7	1 () 1 () 1 ()	1
62 1	24 5 1	1.	4
55.1)	2 De 17	175 175	1
20.0	3/47:	41 d	<u> </u>
69.42	4 %	1.0	1
27	> ∠• ``	1	1
15.13	+ 9.	1.1	1
39 • / ?	5	1:3	1
50±33	18.11	111	1
15.33 39.72 50.73 18.63 71.72	2, 🐧	11	1
71. ?	54.77	1. L	1
94.33	55.11	1 (0	1
94.33 57.13 38.13 56.13	50.1	1.0	1
38 • 11	17.7:	1 1 u	1
56.11	15.11	106	1
18 • 33 35 • 33	1 3. 1)	100	1
35 . 13	21.53	1. 6	1
67.13	59 .]	1:6	1
47.73	14. 17	100	1
60.01	6.71	106	1
55.11	22.11	1: 0	<u> </u>
55.03 18.03 63.33	54.31	166	1
83. 1	59.73	1. 4	- 1
76.1	5, 8, 3, 9	1.6	1
9	40,11	100 100 100 100 100 100 100 100 100 100	1
78.1	18.11	41.	1
94.)	12.11	1 1	ī
45 1	323 + 3 + 4 + 2 23 + 5 1	1.4 1.4 1	111111111111111111111111111111111111111
6. • 13	2.11	1	4
55 · · · ·	4. 1	4.	1
89 1		1) 1: t	1
89.1	4 7 € 1	1	1

PROGRAM OUTPJT---

CETATOR-X	A-colttes	ANG(FAD)	EXPOSURE
1. 7.1	34.51	- 417	2 • 19
2.16	31.12	57:	6.37
3.	37.52	157	0 • 40
33	33.33	4570	1.39
5.80	32.5 +	4571	6.00
5. 3	32.35	4571	0.30
7.15	31.55	157	មិត្ត ដែធិ
3.7'	31.17	ールレデフリ	• 22
9.01	31.57	57.	• 44
120	3 . 13	45 7:	•51
11.00	29.53	457.	• 58
12.34	29.11	457	•58
13.70	23.5L	4571	.58
14.2.	29.51	11.	.58
15.53	23.51		.58
15.00	29.12	2571	.58
17.55	23.21	£14	.99
15.02	23.31	614	1.41
19.36	29.73	44.71	2.69
2,.00	28.73	- 211	3.46
	23.73	: ((. (4.22
21.01	28.73		4.22
22.[[ukuu .u914	5.74
23.)	28.33	• 1914 • 1914	
24.00	25.37		5.5 0
25.00	29.47	•4571	8.14
26.33	23.55	• 0 914	9.58
27.04	29.55	•· 914	9.83
23.03	3 • 1 •	• 4571	10.02
29.00	31.23	.0914	10.23
33.00	31.52	.3656	10.45
31.80	31.33	•3€5€	10.52
32.00	31.13	.1828	10.66
33,5€	31 • 37	•1828	1F.• <u>7</u> 9
3+.11	31.55	•1828	11.73
3 ° • 1 :	31.74	•182b	10.79
35.10	31.655	1828	10.79
37.001	31 5	= • · 515	1.479
38.17	31.23	1828	16.73
33.30	31 . +5	.1628	11.79
4	₹1.55	·1825	10.79
41.5	71.37	.1825	10.73
4.2.	32. 2	.1828	16.79
43.	30.21	.1828	16.79
4 3 .	37.33	.1824	179
i	27.57	.1828	11.02
- •	•		

45.00	~~ ~~		
	32.35	•3656	11.24
47.66	33.14	•1828	11.40
48.06	33.52	•3656	12.52
49.00	33.51	•3914	13.92
50.00	37.71		
51.33		.0914	15.18
	33.71	 € ↓.	18.45
52.3i	33.73	•• (((21.99
B3.26	37.73	- 1 (1)	23.62
54.50	77.7)	51.(25.54
57.01	37.71	- · · i t. i	
5			27 • 33
	33.73	=•⊍€88	28.52
57.0	37.32	3656	3: •73
£3.80	37.5L	•1828	32.61
£7.86	37.59	.182ª	34.41
€ 1.5°	77.33	1828	
£1.3.	35.33		36 • 22
		457	35.68
€ 2. T €	33.23	3914	37.15
E 3. 30	32.33	457	38 • 1.5
6 31	32.33	•182E	36.26
65.76	37.17	.1828	
€6.A.	37.35		38.46
E7.01		•1828	38.53
	32.35	4571	36.59
€5•41	32.53	2742	38 ∙ 66
E9.30	32.13	+ 5 7	38.73
7 3.30	32.53	.4576	
71.30	32.43	1914	38.73
72.01			38.73
77 00	32.31	1826	39 • 1 4
73.00	32.12	1828	39.56
7	31.3+	- • 1£ 28	41.84
73.16	31.75	1828	42.13
76.00	31.57	1828	72.013
77.39			43.69
78.50	32.15	•4571	4° • 33
	31.37	1628	45 • 84
79.01	31.73	914	48.35
82.00	31.59	914	49.04
81.00	31.53	- octub	49.74
82.70	31.37	.1828	
83.00			5° ∙ 69
84.70	32.25	•3656	51.29
	32.5+	•3656	51 • 83
95.30	33.13	•4570	52 • 86
85.20	33.41	.2742	53.33
87.ú:	37.53	•2742	53.64
60.50	37.93	27-2	
83.11	34.25		53.84
9		• 27 - 2	53.98
	35 + 75	•457.	53.98
91	35 • 2+	• = \$ 7 ;	53.98
92.07	35.2+	- •	53.95
93.0	₹# + 2+	- 1	
0 ,	35.2	*	53.93
0			53 .9 3
0.5	37 • 2 •	- • • • • •	53.98
91,11	3= • 1 5	916	53.98
c •	36.15	1.	53.93
95.31	77.5	91:	53.98
90.	7,5		
i iti	3/4.57		53.95
	7 * 6 7 /	45 7	53.98

SITE DATA--X-ROTATED Y-ROTATED

14)	39. 1	100	1
	2 * • * * *	1 ()	1
24.	35.11	1::	1
53.33	. 2. 1	177	1
81.)	3	10. 100 177 100	1
7: . 3		1	•
53.3 24.3 53.3 81.3 71.3	1 7 1 1 1	4	1 1
18. 3	÷ 0•	A	4
56.23	4.6	1	1
86.1	¥ 3•	1	1
	2:.	1 4	1
55.00	ų j	100 100 100 100 100 100	1
69 . 3	4 O . 1 *	1'	1
27. 1	b 2 , •	116	1
15. 3	C.	1	1
841 621 551 691 153 391	54.	1.1	1 1 1 1 1
E 3	18. 1	1::	1
18.	2. ' '	1 * a	1
55. 3 55. 3 15. 3 57. 3 15. 3 67. 3	52.00	1'0 1'' 1'' 1'' 1'' 1''	1
9: 3		1 ()	ī
57 3		1000 1. i	ī
573 383	4.7	4	1
30 • . 1	4.5 13	1.4 1(8) 1.0	4
55.)) 18.1)	1	<u> </u>	4
18. J	1.5.17	1 0 1 d 6	1 1 1 1 1 1 1 1 1
36 • J	21	144	1
67. 3	59.1	100	}
47.03 60.03 55.03 18.03 83.03 76.23	14.	100 100 100 100 100 100 100	1
60.01	6. 1	1 6.5	1
55.11	22.11	1 0.0	1
18.03	54.17	100	1
83. 3	59.13	1 ' F	1
75.33	5.8.1.3	1: 0	1
903	40. 1	16 A 16 G	1
75.22	18.11	1.0	1
943	323434442442445551112511 2516246.	1. c 1. c 1. c 1. c	1
45 - 13	5 11	1	ī
45 • 13 6 • 13 66 • 13	2. * *	1.	1
65.:	16.11	1	ī
65	# ₩ ₩ ``	1.	1
89. 1	I C • · ·	4 . '	•

PROGRAM OUTPJI---

1.1. 39.51457	X-FCT1TED	Y-P)[1[]	ANG (FAD)	EXPOSURE
3.01 33.52457. 5.00 4.11 33.1357. 5.30 32.54457. 7.00 0.06 32.15457. 7.00 0.07 31.35457. 7.00 0.08 31.37457. 7.00 0.08 31.37457. 7.00 0.08 31.37457. 7.00 0.08 31.37457. 7.00 0.08 31.37457. 7.00 0.08 31.38457. 7.00 0.08 31.38457. 7.00 0.08 31.39457. 7.00 0.08 31.30 23.51457. 7.00 0.08 23.51457. 7.00 0.08 23.51457. 7.00 0.08 23.51457. 7.00 0.08 23.51457. 7.00 0.08 23.51457. 7.00 0.08 23.51457. 7.00 0.09 11 0.01 23.51 0.09 11 0.01 23.51 0.09 11 0.01 1.01 0.02 23.73157. 7.00 0.09 11 0.01 1.01 0.02 23.73157. 7.00 0.09 11 0.01 1.01 0.02 23.73157. 7.00 0.09 11 0.01 1.01 0.09 11 0.01 1.01 0.00 23.73157. 7.00 0.00 23.7				را ق • ∪
3.01 33.52E7.	2,5%	32	4.5.7	″ • J .
4.11 33.13		33.52		
5.30 32.54 4571 0.30 7.01 31.35 4571 0.30 7.01 31.35 4571 0.30 3.11 31.37 4571 0.22 3.11 31.37 4571 0.51 13.30 31.33 4571 0.58 12.05 23.11 4571 0.58 13.31 23.51 4571 0.58 13.31 23.51 4571 0.58 13.32 23.51 4571 0.58 13.40 23.51 4571 0.58 13.41 23.51 4571 0.58 13.42 23.51 4571 0.58 13.51 23.51 4571 0.58 13.51 23.51 4571 0.58 13.51 23.51 4571 0.58 13.51 23.51 4571 0.58 13.52 23.73 0.911 1.41 0.99 13.52 23.73 0.911 1.421 0.911 0.911 0.911	4.00	33. 13	57.	
0.56 32.17 4571 0.00 31.75 31.75 4571 0.00 31.77 31.77 4571 0.00 31.77 31.77 4571 0.00 31.77 4571 0.00 31.77 4571 0.00 31.77 4571 0.51 31.50 23.51 4571 0.58 13.51 23.51 4571 0.58 13.51 23.51 4571 0.58 13.51 23.51 4571 0.58 13.51 23.51 4571 0.58 13.51 23.51 4571 0.58 13.51 23.51 4571 0.58 13.51 23.12 4571 0.58 13.51 23.12 0.911 1.41 13.51 23.12 0.911 1.41 13.51 23.73 411 0.91 1.41 13.51 23.73 411 0.91 1.41 13.51 23.73 411 0.91 1.41 13.51 23.73 411 0.91 1.41 1.41 23.73 411 0.91 0.91 0.52 23.73 411 0.91 0.55 23.45 23.73 411 0.58 23.45 23.45 23.47 0.514 0.59 23.45 23.47 0.514 0.52 23.47 0.514 0.52 23.47 0.514 0.52 23.55 0.514 0.52				
7.60 31.35 457. 0.00 3.11 31.17 457. 22 3.11 31.17 457. .44 10.10 30.37 457. .51 11.60 23.53 457. .58 12.15 23.51 457. .58 13.51 23.51 457. .58 13.51 23.51 457. .58 13.51 23.51 457. .58 13.52 23.51 457. .58 13.52 23.51 457. .58 13.51 23.51 457. .58 13.52 23.51 457. .58 13.51 23.51 457. .58 13.51 23.51 457. .58 13.51 23.51 457. .58 13.51 23.51 457. .58 13.52 .21 .457. .58 13.53 .21 .457. .58 13.53 .21 .457. .457. .457				
3.11 31.174.77 .22 3.12 3.575.71 .44 15.30 31.334.571 .51 11.00 23.534.571 .58 12.30 23.114.571 .58 13.30 23.514.57 .58 14.30 23.514.57 .58 14.30 23.514.57 .58 14.30 23.514.57 .58 14.30 23.514.57 .58 14.30 23.514.57 .58 15.30 23.514.57 .58 17.30 23.514.57 .58 17.30 23.734.51 .991 .991 .45 13.30 23.734.51 .491 .491 .491 13.30 23.734.51 .491 .491 .492 23.30 23.734.51 .491 .594 23.51 23.734.51 .991 .655 24.51 23.33 .514 .4570 .814 .893 24.51 23.37 .991 .991 .993 27.30 23.55 .991 .991 .993 27.30 23.55 .991 .993 .993 28.01 23.55 .991 .993 28.01 31.33 .1828 .10.66 33.00 31.31 .35 .991 .991 .10.79 35.30 31.37 .1828 .10.66 33.00 31.37 .1828 .10.67 37.30 31.35 .991 .991 .10.79 35.30 31.55 .991 .10.79 35.30 31.55 .991 .10.79 35.30 31.55 .991 .10.79 37.30 31.55 .991 .10.79 37.30 31.55 .991 .10.79 37.30 31.55 .991 .10.79 37.30 31.57 .10.11 .10.79 43.10 32.23 .91 .10.79 43.10 32.23 .91 .10.79 43.10 32.23 .91 .10.79				
3.17 3.57		31 7		
19.38	3.11			
11.00 23.53657 .58 12.00 23.1)4576 .58 13.00 23.514576 .58 14.00 23.51457 .58 14.00 23.51457 .58 17.00 23.51457 .58 17.00 23.73457 .58 17.00 23.73457 .58 17.00 23.7366 .59 15.06 23.7366 .59 23.07 23.7366 .59 23.07 23.7366 .59 23.07 23.7366 .59 23.07 23.7366 .59 23.07 23.7366 .59 23.07 23.7366 .59 23.07 23.7366 .59 23.07 23.7366 .59 23.07 23.7366 .59 23.07 23.7366 .59 23.07 23.7367 .67 .67 .67 .67 .67 .67 .67 .67 .67				
12.00				
13.11				
14.31				
14.11		23.51		
12.00				
17.00 28.21		29.12		
15.20		28.21		
13.37 28.73				
2 .3u				
21.[1 29.73		29.73		
22.20 22.73		20 73		
23.01 28.33				
24.11 29.97 914 6.55 26.07 29.47 571 8.14 26.07 29.55 514 9.58 27.00 29.55 914 9.83 28.01 37.14 4571 10.02 29.00 37.23 914 10.23 39.00 37.52 3656 10.45 31.00 31.01 3656 10.45 32.01 31.13 1828 10.66 33.00 31.37 1828 10.79 34.01 31.37 1828 10.79 35.01 31.37 1828 10.79 35.01 31.37 1914 10.79 35.01 31.37 1914 10.79 37.01 31.37 1914 10.79 37.01 31.37 1914 10.79 4.01 32.11 114 10.79 4.01 32.11 114 10.79 4.01 32.11 114 10.79 4.01 32.23				
26.07 27.47 .457t 8.14 26.07 22.55				
26.00 29.55				
27.33 29.55 914 9.83 28.51 37.14 .4570 10.02 29.80 31.23 914 10.23 33.00 31.52 .3656 10.45 31.00 31.31 .3656 10.45 32.00 31.31 856 10.52 32.00 31.37 828 10.66 33.00 31.37 828 10.79 35.00 31.35 914 10.79 35.00 31.55 914 10.79 37.30 31.37 914 11.79 37.30 31.37 914 11.79 33.00 31.37 914 11.79 40.31 32.11 914 11.79 40.31 32.11 914 10.79 40.31 32.23 914 11.79 40.31 32.23 914 11.79 40.31 32.23 914 11.79				
28.31 37.14 .4570 10.02 29.80 31.23 .9914 10.23 33.00 34.52 .3656 10.45 31.00 31.31 .3656 10.45 32.00 31.31 .3656 10.52 32.00 31.37 .1828 10.66 33.00 31.37 .1828 10.79 35.00 31.35 .1914 10.79 35.00 31.35 .1914 10.79 37.30 31.37 .1914 11.79 37.30 31.37 .1914 11.79 4.31 32.11 .1914 11.79 4.31 32.11 .1914 10.79 4.330 32.23 .1914 11.79 4.330 32.23 .1914 11.79 4.330 32.23 .1914 11.79 4.331 32.33 .1914 11.79				
29.80 31.23 .0914 10.23 33.00 34.52 .3656 10.45 31.00 31.31 .3656 10.52 32.60 31.31 .1628 10.66 33.00 31.37 .1628 10.79 34.33 31.45 .0914 10.79 35.00 31.33 .0914 10.79 36.00 31.34 .0914 11.79 37.30 31.33 .0914 11.79 33.00 34.32 .0914 11.79 4.31 32.11 .0914 11.79 4.31 32.11 .0914 11.79 4.31 32.11 .0914 11.79 4.330 32.23 .0914 11.79 4.330 32.23 .0914 11.79 4.330 32.23 .0914 11.79 4.330 32.23 .0914 11.79 4.330 32.23 .0914 11.79				
39.90 31.52 .3656 10.45 31.00 31.31 .3656 10.52 32.00 31.13 .1828 10.66 33.88 31.37 .1828 10.79 34.10 31.45 .6914 10.79 35.00 31.55914 11.79 37.00 31.55914 11.79 37.00 31.33914 11.79 37.00 31.33914 11.79 4.31 32.11 .914 10.79 4.31 32.11 .914 10.79 4.310 32.23914 11.79 4.300 32.23914 11.79 4.300 32.23914 11.79				
31.00 31.31 .3656 10.52 32.00 31.13 .1828 10.66 33.88 31.37 .1828 10.79 34.13 31.45 .6914 10.79 35.00 31.55 .6914 10.79 36.20 31.55 .6914 10.79 37.30 31.54 .6914 10.79 37.30 31.33 .6914 10.79 33.80 31.33 .6914 10.79 4.31 32.31 .914 10.79 4.31 32.31 .914 10.79 4.330 32.23 .914 10.79 4.330 32.23 .914 10.79 4.330 32.23 .914 10.79				
32.00 31.13 .1828 10.666 33.00 31.37 .1828 10.79 34.10 31.45				
33.88 31.37 .1828 10.79 34.23 31.45 .6914 10.79 35.00 31.55 .6914 10.79 36.20 31.55 .6914 10.79 37.00 31.55 .6914 10.79 33.00 31.33 .6914 10.79 39.00 31.33 .6914 10.79 4 .31 32.11 .914 10.79 4 .31 32.11 .1914 10.79 4 .30 32.23 .914 10.79 4 .31 32.23 .914 11.79 4 .31 32.23 .914 11.79				
39.23 31.45				
35.00 31.55914 10.79 36.00 31.54914 11.79 37.00 31.73914 11.79 37.00 31.73914 11.79 39.00 31.33914 11.79 400 32.01914 10.79 400 32.01914 10.79 400 32.01914 10.79 400 32.01914 11.79 400 32.01914 11.79 400 32.01914 11.79				
36.00 31.5+				
37.31 31.73			0.91	
33.07 31.33				
39.30 34.92 .916 11.79 6.31 32.11 .516 10.79 61.00 32.11 .1516 10.79 42.10 32.13 .516 11.79 43.30 32.23 .516 11.79 40.31 32.33 .1916 11.79				
4 .31 37.11 .514 10.79 41.10 37.11 .1514 179 42.11 37.13 .514 11.79 43.30 32.23 .514 11.79 43.31 32.33 .1914 11.79	31.11			
#1.00 32.13 .1514 10.79 #2.10 32.13 .514 11.79 #3.00 32.23514 11.79 #3.01 32.33914 11.79				
42.11 32.13 . 514 11.79 43.00 32.23514 11.79 43.01 32.33514 10.79				
43.30 32.23	F 1.50			
4 - • 31 32 • 33 • : 914 11 • 79				
41.00 32.47 .c914 11.21				
	4 T + 3 C	32.47	• (91L	11.21

46.00	32.55	011	11.63
	3:4/3		11.85
47.90	32.53	.7911	
6 A. 7 3	32.7+	• 610	12.61
49.30	32.33	. 514	13.37
5 1.30	32.33	• . ^C 1 4	14.77
51.20	37.12	• 514	15.98
52.00	37.12	• • • • •	17.18
	37.12		18.93
£3.00	22012	- · · · · · ·	
F 11	32 - 3	- • * i	19.97
P (4. 3.3	32,12	-	21.02
5 m . T .	37.12	- · · · · · ·	22.56
57.0L	37.72	• • • • • •	22.78
53.03	33.31	.2742	23.36
F 3	37.53	.27.12	23.94
€ ,30	33.35	27 2	24.36
	3 2 +	.3f5(24 • 64
£1.75		03536	
€2.21	35 + 3 +		24.73
63.31	3 . 2	. 51.	24.93
64.30	3 5 2	• .911	25.19
Essi	3+.51	911	25.19
EA. Di	34.73	.3914	25.19
£7.31	34.21	4 5 7 1	25.19
		- 19	25.19
F 20 % L	37.72	7	
E3.1	32.31	•162F	25.19
7:	33.+1	L F 7 (25.19
71.31	37.31	· 1576	25.13
72.31	33.52	2742	2: • 51
73.: .	37.74	271.2	26.02
76.55	33.05	2742	27.31
		2742	28.59
75.01	37.73		
76.21	32.51	272	30.15
77.::	32.71	457.	31.71
78.81	31.91	91 ^L	33.22
79.60	31.32	016	34.73
851	31.73	514	35 • 43
81.50	31.73	t . t	36.12
82.36	31.32	.1828	37 ⋅ 08
	32.13	.1825	37.67
83.05			38 • 27
84.Ju	32.23	.1829	39 • ú 2
85.30	32.73	•457	
86.20	32.35	.1828	39.78
87.00	37.35	.3156	40.12
83.00	33.73	.3655	41.27
89.10	34.11	•3E5E	41.041
93.01	34.51	• 5 F 7	40.41
91.30	35.33	· 457	42.41
9 2 . û t	37. 3		4 41
			1:0 • +1
93.23	37.13	• • • •	
Q ', • ~ .	35.23	 (:,	4.41
93.00	37.13	→ → → √ → □	4 .41
95.31	35.73	• · · ·	4 . 41
97.00	35.23	⇒Cst	4" • 41
94.0	35.13	- • '	41.41
64.	77.11	916	441
	34.51	57	441
1f	J = ♦ ∋ L	• = -	· · · · · · · · · · · · · · · · · · ·

SITE DATA--(ETATOS-Y CETATOS-X

14. 7	39.11	1 20	1
53	39.11	1 . c	1
14? 531 240 531	3	1. 9	1
53. 1	÷ 2•	1:	1
81	31.	1	1
76.3	44,53	1 :	1
18.3	31.0 1 01.01 13.11	10.0	1
38.12	44, 23	1. 0	1
53.3.3 53.3.3 53.3.3 53.3.3 53.3.3 53.3.3 53.3.3 53.3.3 53.3.3 5 5 5 5	44.13 43.13 43.13 42.13 42.13 43.13 13.13	11 0 1 0 1 0 1 0 1 0 1 0 1 0 1 0 1 0 1 0	111111111111111111111111111111111111111
52.3	25.11	100	1
55. 3	2.50	10)	1
59.1	49,11	1.1	1
27. 1	₹ 2• 3	1 -	1
15.11	+ d* , ,	1 L	1
39.41	54.55	1. (1
£(• .)	18.17	1.(1
18. 1	2. 11	1 i i	1
71.13	54. " 3	1 11	1
9 ·	55.73	1 0 ∂	1
57.:3	5 . • 7.7	1.0	1
38.]	17.73	1.0	1
56.13	1 F. 13	1. :	1
18.:3	1 3. 7 7	1. i	1
36	21.77	10%	1
57.3	59.11	100	1
47.3	14,13 6,13 22,11 54,13	1 0 a	1
56.4.13	6.1	106	1
55.11	22.11	1 1	1
18.73	54.79	1 v û	1
83.1)	59.11 58.13	100	1
76. ?	58.)	1	1
90.03	49.73	1 ~0	1
76	18.77	1 '	1
38. 1 58. 1 18. 1 57. 1 57. 1 55. 1 55. 1 57. 1	23 43 + 1 4 4 2 2 4 5 5 1	1.0 1.0 1.0 1.0 1.0 1.0 1.0 1.0 1.0 1.0	1 1 1 1
45.0	500	1:.	1
50.03	2.11 16.11 18.17	11.	1
66. 3	2•11 16•11 18•17	1	1
89. 1	1 8. 17	1 : 6	1

PROGRAM OUTPJI---

X-ROTATED	Y-ROTATED	ANG (PAD)	EXPOSURE
1.60	34.51	~.4.71	Ç • • • •
2.00	3:,12		[•∪[
3.07	33.52	117	ំ 🗸 🦸
2 . 2 L	33.13	4:71	e • 0 \$
5.30	32.54		0.13
5.36	32. , 5	- 04 5 74	1.00
7.20	31,55	- + + 5.71	0.40
30.01	31.)	71	• 22
3,10	37.57	4576	.44
1	3:.33	5 7 :	•51
11.33	23.53	71	• 53
12.50	29.13	57	•58
13.00	23.51	 ≒571	•58
1 4. 3.	23.51	→ • 1 %,	•53
15.35	23.51	- out ut	•53
15.35	23.12	- · · 57 :	•58
17.80	23.21	.1914	. 99
13.54	23.31	91L	1.41
19.25	29.73	·4571	2.69
2	23.73		3.40
21.37	23.73	- Cut	4.22
22.50	24.77	i i t	4.98
23.33	25.83	. 914	5.74
24.10	29,37	. 914	6.5)
25.31	23.+7	• + 570	8.14
25.56	23.55	. 014	9.58
27.50	23.55	. 914	9 • 8 ਹੋ
28.35	3 . 14	.657	10.02
29.31	31.23	. 61L	10.23
3	30.52	.3656	19.45
31.00	31.1	.3656	19.52
32.13	31.13	.1828	17.66
33.01	31.37	.1828	179
34.31	31.55	.1828	11.79
35.25	31.5+	. 914	1:.79
35.30	31.74	- 914	15.79
37.1.	31.33	914	1:.79
30.07	71. 72	611	11.79
39.01	77.)1	914	1 .79
L 7, 7,	77.13	61	16.79
41	32.13	914	10.79
# 2. C.	72.23	914	11.79
43.9.	32,33	• . 51k	1 .79
4 70 € 6 6 4 6 5 1	32.47	- 914	10.79
4	32.55	. 914	11.21
4 • • •	: 1 0 2 3	● v フ. ™	77 0 5 7

46.35	32.3+	• 27 4 2	11.43
47.00	32.37	011	11.05
43.33	33.13	•- <u> </u>	12.42
49.30	33.13	3620	13.40
53.51	37.13	- •∂638	14.38
51.Ju	37.33	 .(∪[15.37
52.00	33.13	1:	15.35
53.Ju	37.33	- •0€ H	17.56
Sa no	37.03	+ • ≥ () ≥ 6	18 • 46
F5.33	33,13	- • 1 of	18.57
F-1.00	37, 13		19.57
57.00	33.73)til	19.64
£8.J.	33.31	.2742	20.23
	33,53		21.00
E 3. 20		•2742	
€01	37,37	.2742	2: • 43
61.51	35.25	.3156	21.05
62.1.	34 . 34	916	21.13
€ 3. 3 .	3+++	• (91 4	21.19
64.20	34,53	. 514	21.26
€5.30	34,52	•. 914	21 • 26
65.00	34.71	. 5514	21.25
67.00	34.22	1 . 7 .	21 • 2ს
€3.t.	33.73	571	21.25
	37.31	1628	21 • 26
E3.01			
7 - 30	33.42	157	21 • 26
71.30	37.31	•457;	21 • 25
	33.53	2742	21.55
72.33			
73.00	37.35	2742	22 • 119
74.33	33.37	27 42	23.38
75.31	32.73	27-2	24.00
76.00	32.51	2742	26.22
77.31	32.71	4571	27.78
73.35	31.33	1626	29.29
79.35	31.7+	016	31.81
84.50	31.55	: 914	31.50
81.80	31.55	tif	32.19
82.3L	31.33	.1828	33.15
63.00	32.21	.3656	33 • 74
8 + • 9 8	32.5)	•365E	34 • 34
85	33.39	.4575	35.32
86.10	37.37	.2742	35.78
87.00	37.55	.2742	35.09
2 8.35	37.37	.2742	36.30
83.33	34.21	.2742	35.44
Q 1.	3 7 1	•4570	36 . 44
91.11	35, 23	. 45.7	35 . +4
92.10	37.21	the state of the	35.44
93.11	37.23	-, t.;	35.44
9-10:	37.21	-	35.44
Ç').	37.21	!:i	35 • ₩4
0.5	37.21	- · . ·	35 . 4
	713	,514	35.44
97.00			
Ç:.:	74.17	• • • •	35 . 44
gray and	33.11	- c1t	36 . 44
		- 4 5 7	
10 .37	→ 7 €		36 • 44

Appendix C4
Automatic Model Output
(see Table 3, 4 KM Column)

OPTIMUM FLISHT PATH INPUT SUMMARY---

NPAYS=11	04ECKPT 1 2
VO STEPS=25	CHECKPT 1 1.175, 35.373 2 133.375, 35.333
NO SITES= 38	A/C VELCCITYMIN=648.0 MAK=722.0
NO SITE TYPES=	
••	CORRIDOR WINTHE 2

PK DATA--EACH SITE TYPE

NO. OF MEASUREMENTS=

۲.

SIGMA(RANGE) =

0.35

SISMA(ANGLE) =

... DFG.

NPAYS=11

40 STEPS=25

SITE TYPE= 101 SITE TYPE NO= 1

.075	.231	.2 × 1	.272	.133	
• 14	.137	5 t ē	. 217	. <u>1.25</u>	C 0L U 4 4 5 = 2 [
-	633	(T)	17	-	S=5
6 () ()	1 T	هــو ر ه) •	در. ايا		1 5

SITE DATA--X-ROTATED Y-ROTATED

14.53	39.00	1.34	4
53	27.1	1 1	1 1 1 1 1 1 1 1
24.00	27.U0 35.00 42.00	1 2 5	-
53.63	42.00	1 1 1	1
81. 7	3: .01	4 * *	4
75.03	44.16	1	1
18.00	43.38	1 17	1
56 3	30.83 44.36 43.38 44.00	1 (•
34.53		127	1
62. 1	25.3)	1:.	1
56.01	27.53	153	1
69.03	49.50	157	1
27.33	೯₽.30	1.5	1
15.37	49.30	10	1 1 1
39.03	64.55	137	1
5 • 13	10.30	12.	1
31.03 73.03 18.03 56.03 56.03 56.03 15.03 15.03 18.03 71.03 57.03 38.03 56.03	25.00 27.50 49.00 49.00 49.00 18.00	101	1 1 1 1 1
71.03	• • • • · · · · · · · · · · · · · · · ·	123	1
94.31	65.30 61.30	107	1
57.0)	62.00	10.	1
38.00 56.00 18.00 36.00	17.00	103	1
56. UT	15.03 13.00 20.00 59.00	100	1
13.03	13.30	10)	1
36.00	20.00	133	1
.7 na	29.00	163	1
67.33 47.03 63.00	14.00	100	1
55. 3	14.03 6.03 22.00	100	1
55.13 18.03	64.00	4 2 3	1 1 1
93.00	69.00	100	<u> </u>
75.11	58.70	103	1
91.11	60.00 58.00 49.00	100	1
93.09 76.33 93.33 76.33	18.33	1 / 3	1
3+.)	62.43	1 1	ī
	FL.]	1.	1
45.03 53.03 66.30 49.03	£4j 2.0)	1:	:
66.30	16.3]	100	1
39.01	F4.13 2.03 16.03 18.00	100 100 100 100 100 100 100 100 100 100	ī

PROGRAM OUTPUT---

X-R0111EF	Y-FOTATED	ANG (RAC)	EXPCSIRE
4.17	35. €3	0000	6 . D 3
8.11	35.01	មប្រឹង	5.13
12.j	35.€3	 0000	11.69
16.13	35.~3		20.02
21.13	35.03	0037	27.51
24.13	35.€3	3031	329
28.13	35.€3	2332	33.31
32.13	35.03	3033	33.87
36.13	35.CJ	00;;	33.87
40.33	35.:1	:2	33.87
44.]3	34.63	0914	33.87
48.13	34.27	1914	35.64
52.13	34.63	.3914	42.62
56.13	35.(3	.:914	45.51
69.13	34.26	1828	47.27
64.13	32.23	457U	5(.58
68.33	31.93	1914	51.84
72.33	31.55	2914	52.51
76.11	31.55	0188	58.76
88.13	33.53	• 4573	€2.58
84.11	34.27	.1828	€6.49
88.10	34.63	.0914	67.31
92.11	34.63	1111	67.31
96.11	34.63	:102	67.31
13 37.	32.67	457	67.31

SITE DATA--CETATOR-Y CETATED

14.00	39. 0 0	100	1
57.	27.30	1.3	1
24.17	35.00	1('	1
24.03 53.03	42.30	165	1
81 1	30.00	1 00	1
24.13 53.03 81.13 71.03	44.36	1 3 3	1
18.31	43.30	100	1
81.01 71.01 19.01 56.01	44.55	1.7	1
34)	43.83	133	1
52.11	25.03	10.	1
56.00	30.00	100	1
63.31	49.00	133	1
27.17	50.00	111	1
15.33	49.33	13.	1
39.03	27.30 27.30 47.00	130	1
64.00	18.30	163	1
34.13 52.13 56.03 63.37 27.63 15.13 39.03 63.03 18.13 71.03	18.30 2.30	157	1
71.uJ	64.00	1 🕹 🕽	1
94.03 57.63 38.33	64.00 65.00 61.00	100	1
57.[]	66.00	100	1
39.11	1/.11	160	1
56.33	15.00 13.00 20.30 50.00	101	1
18.03 35.33 57.33	13.00	1.5	1
35.33	20.30	160	1
57. 33	50.00	10 0	1
M / a 1	14.33	100	1
60.00	14.03 6.00	103	1
55.03	22.00	د 10	1
18.33	22.00 64.00	130	1
60.00 55.00 18.10 83.00 76.00	60 53	100	1
76.00	58.03	163	1
# 1 a 2 h	49.20	11)	1
76.33	18.00	100	1
9 +• }	58.00 58.00 18.00 62.00 54.00 16.00	1000 1000 1000 1000 1000 1000 1000 100	111111111111111111111111111111111111111
45.13	54.00	1 C 3	1
51.11	2.11	133	1
56.77	16.00	1 3.	1
83.33	18.03	1 5]	1

PPOSRAM CUTPUT---

X-ROTATED	Y-FOTATED	ANC (RAC)	EXPOSURE
4.37	35	12	•96
8.10	35.00	:::::	5.13
12.17	35.11	•.3233	11.69
15.17	35.1,1		21.02
20.30	35		27.51
24.11	35.71	:333	30.29
28.10	35.13	1233	33.31
32.11	35.71	• • 1 1 2 2	33.87
36.13	34.63	0914	33.87
40.13	34.27	1914	33.87
44.30	33.91	1914	33.87
48.10	34.27	. 1914	37.31
52.13	33.93	:314	47.49
56.11	33.53	2914	53.44
60.11	32,73	1828	66.38
64.33	30.83	4573	63.93
68.13	30.46	3914	54.24
72.13	33.83	. 3914	65.91
76.13	31.13	.3914	72.16
80.33	29.23	4573	74.93
84.33	30.35	.2742	77.31
88.33	32.32	.457	79.29
92.13	33. 6	.1828	78.29
96.11	34.18	. 2742	78.29
100.10	32.22	- 457]	78.29

SITE DATA--X-ROTATED Y-ROTATED

14.57	30.00	1 7 1	1
53.11	27.06	13.1	1
24.03	35.00	1.7	
	42.00	1 1	<u>.</u>
53.13 81.33	36.00	10	4
81.07 70.03	44.30	10.	4
18.03	43.30	1	÷
56.03	44.00	4	÷
84.0j	43.00	1	1
60 60	25.00	103	1
52.67	20.00 20.00	133 143 143 163 163 163 163 163	<u>.</u>
50 C I	32.50 49.00	100	4
27 11	52.33	1 1	1
45 11	49.00	A G S	4
120 an	64.00	4 1 3	1
3 7 9 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3	48 83	1 3	1
18 1	70.01	400	1
52.61 55.01 59.01 27.01 15.11 39.00 61.01 18.11	18.03 2.00 64.00	1 1 1	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
94.37	64 00 65 05	1)	1
57.00	65.83 60.00	100	
57.00 38.00	17.85	103	1
39.03 56.00	15.01	123 103 103 103 103 103 103 103 103 103	1
13.33	47 C)	1 3 3 4 1 H	1
	13.01 20.00	1 U U	1
36.00 67.01	26.00 ED 00	100	1
47.02	59.00 44.85	199 190 199	1 1
60.00	14.00	163	1
55.03	6.03 22.û3	# # J	1
18.00	64.00	400	1 1
	£0.00	153	1
83.03	E G - UU	. ↓ ↓ J 4 * 3	1 1
76.31 99.03 76.21	58.30	100	1
76 17	49.00 18.0	4 7 7	1
94.	62.60	1 0 0	1 1
94.7) 45	62.00 54.01 2.00	100 100 100 100 100 100 100 100 100 100	1
5 ()	9 4 • • • •	1	1
5 (•) 55• 01	16.03	131	1
89	18.60	4 1 1	1
110	→ 0 • 0 0	1	1

PROSPAM DUTPUT---

X-ROTATED	Y-FOTATED	ANG (RAE)	EXPOSURE
4.30	33.63	4571	0.05
8.11	33.53	0338	1.67
12.13	33.∩3	1110	4.71
15.17	33.3	3333	7.24
211	33.13	••1111	14.04
24.33	33.53	23.5	16.82
28.11	33.13	••3333	15.84
32.17	33.03	• • 13.1	21.40
36.1)	33.23	5333	21.43
40.33	33.43	.1914	20.49
44.33	33.41	0330	28.45
48.) 0	33.77	.3914	23.84
52.33	33.4)	1314	34.02
56.]]	33.77	. 1914	37.76
60.77	33.03	1828	41.93
64.)3	31.05	4575	45.53
68.13	315	3	45.80
72.33	31.76	3333	47.47
76.10	31.16	0310	53.71
80.13	33.∂3	•4573	57.53
84.10	33.77	• 1929	61.44
88.10	34.51	.1328	62.27
92.11	34.51	0000	62.27
36.11	34.87	.1914	62.27
132.33	32.91	- • 4573	62.27

SITE DATA---Y-ROTATED Y-ROTATED

14.37 53.23 24.60 53.00 81.00	39.00 27.00	11111111111111111111111111111111111111	1 1
24.00	35.00 42.00	153	1
53	42.00	133	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
81.33	₹(•⊌0 // 00	1.	:
18.00	44.00 43.00	A D	1
18.07 56.07	44.00 43.00 44.00 45.00 35.00 59.00 59.00 64.00 64.00	1 1	1
84	43.00	1.3	1
62.33	25.33	123	1
84.03 62.03 56.03 57.03 15.03 33.03 61.03 71.03 94.03 57.03 36.03 67.03 47.03	35.03	1 3 3	1
63.12	49.90	1.3	1
27.33	52.00	10.	1
15.0	49.55	103	1
333	64.00	111	1
6.01	200	1 2	1
74 3	2 . UJ 64. AN	4.3.1	+ -
94.97	65.00	133	4
57. čJ	65.00 65.00 17.00	110	1
38.31	17.50	1);	i
56.0J	15.0J 13.00 20.00 59.0J 14.0J 6.00 22.00	100	1
18.00	13.88	1 ∪ ∂	1 1 1
35.03	26.00	133	1
67. 33	59.00	10 J	1
47.03	14.03	155	1
60.03	6.UU	1 6 J	1
55.00 18.00 83.00 76.00	64. an	150	1 1 1 1 1
83.01	60.00	113	1
76.00	58.03	133	ī
90.00	40.00	100	1
75.33	18.03	1 🕻 .	1
94.00	62.00	103	1
90.03 76.33 94.00 45.33 6 .33	54.33	133	1 1 1 1
6 . 33	2.00 16.00	1 ,	1
66.00	16.00	1	1
89.CJ	18.30	101	1

PROGRAM OUTPHI---

X-ROTATED	CETATOR-Y	ANG (GAE)	EXPOSURE
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12.33 15.33	33.13 33.63	0;;; 0]};	4.71 7.24
21.1)	333		14.34
24.30	33.03	1730	16.82
28.11	33.13	0303	19.84
32.13	33.13	••]].	26.40
36.17	33.03	÷.7332	2(• 4)
40.10	33.43	.0914	20.40
44.70	33.77	.1914	20.42
48.13	34.13	.1914	23.84
52.10	33.77	0914	34.02
56.);	33.43	:914	37.76
60.10	33.63	3914	41.93
64.11	31.07	4573	45.53
68.17	31.07	0000	45.8G
72.11	31.17	v 3 12	47.47
76.13	31.07	3332	53.71
80.33	33.63	.4573	57.53
84.11	33.77	.1828	61.44
88.13	34.51	.1828	E 2.27
92.13	34.51	3303	62.27
36.11	34.89	.3314	£2.27
100.33	32.91	4573	52.27

SITE DATA--CETATOR-Y CETATCF-X

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14.00 53.01	27.11	4.1	4
26.01	75 33		7
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81.27	7 - 400	1.5	4
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56.13 84.13	44.06	103	1
34. 13	43.37	1 2	4
52.11	25.00	1	1
55.01	37 5A	4 1 1	•
14.03 53.03 24.03 53.03 7.00 18.03 56.03 56.03 56.03 56.03 57.03 15.03	39.00 37.00 35.00 42.00 43.00 43.00 43.00 45.00	4.0	4
09.1	49.34	1.	<u>.</u>
27.33 15.03	57・41	139	1
1 5•€3	49.53	101	1
30.0	€4.33	100	1
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39.00 56.00	17.30	1.1	4
90.1	47 30	100	<u>.</u>
18.03 36.03	10.03	1 1 3	1
35.03	20.00	111	
67.01	FG . 00	1 J I	1
47.03	14.00	1	1
60.00 55.00	0.00	134	1
55.J]	22.00	13.	1
18.07 83.00	22.00 54.00	123	1
83.00	69.00	101	1
76 00	58.9¢	4.53	•
76.00 90.00	20.46	4 5 .	
90.00	49.0) 18.00	1 4 3	1
76. J. 3	18.50	1	1
34.13	F2.03	133	1
45.J]	62.03 54.00	133	1
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65.03	16.30	1 1 1	1
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36.13 33.77 25.	
41.33 33.771201 21.	
44.33 33.431914 21.	
48.12 33.131914 25.	
52.17 33.47 .0914 35.	29
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63.33 33.43 .0914 49.	42
64.)] 31.43457) 52.	47
68.13 31.430.23 52.	73
72.33 31.430000 54.	43
76.10 31.435008 60.	64
80.33 33.43 .4573 64.	
84.10 34.14 .1828 68.	
88.]] 34.51 .3914 69.	
92.10 34.510000 69.	
96.13 34.87 .0914 69.	
100.13 32.914572 69.	

SITE DATA--X-ROTATED Y-ROTATED

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42.30	13.	1
35.00	10.	1
44.50	163	1
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44.00	103	1
43.00	1.	1
25.00	133	1
47.30	100	1
40.33	1.	1
52.00	100	1
49.50	100	1 1 1 1 1 1
64.00	1.00	1
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2 • û Đ	103	1 1
64.03	13.	1
65.00	103	1 1
60.30	100	1 1 1 1
17.03	101	1
15.00	103	1
13.60	100	1
20.00	100	1
59.00	133	1
14.00	133	1
6.00	193	1
22.80	100	1
64.00	103	1
69.00	100	1
	103	1
49.00	10)	1
18.00	13.	1
62.00	103	1
54.33	1:1	1 1 1
2.00	150	1
16.00	100	1
10.00	100	1
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PROGRAM OUTPUT---

X-ROTATED	Y-FOTATED	4NG (94E)	EXPOSURE
4.33 3.33 12.33 16.33 27.33 24.33	35.03 35.03 35.03 35.03 35.03 35.03 34.63	0000 0000 0000 0000 0000 0000	3.00 5.13 11.69 20.02 27.51 39.11
28.13	34.27	1914	33.13
32.13	33.91	1914	33.69
36.11	33.9)	:003	33.69
43.33	33.91	0333	33.69
44.33	33.53	0914	33.69
48.13	33.17	3914	36.73
52.13	33.53	.3914	43.71
56.33	33.17	3914	45.71
60.13	32.83	0914	47.42
64.33	30.83	4573	50.47
68.13	30.47	0914	50 .73
72.33	30.83	• 0914	52.40
76.33	31.27	. 6914	58.64
80.10	29.23	4570	61.42
84.11	33.36	.2742	63.80
88.11	32.33	• 457)	84.73
92.33	33.15	.1828	54.78
96.33	34.19	.2742	64.78
100.33	32.22	- • 457	54.78

SITE DATA--X-ROTATED Y-POTATED

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533 513 73 18.03 56.3	44.00 47.00 44.00 43.00 25.00	1	1
56.73	44.33	135	1
84.00	43.03	1. 1	1
62.03	25.53	100	1
14.07 53.07 24.07 53.03 51.03 7.003 18.03 56.03 54.03 56.03 69.03	42.53	1 3 3	1
69.11	49.33	1	1
31.33 71.33 18.03 56.33 84.03 62.03 56.03 69.03 27.33	47.53 49.13 52.13	11.	1
151 39.00	49.33	1	1
39.03	54.00	137	1
60.00	40 33	11.	1 1 1 1 1
84.03 62.03 56.03 63.03 45.03 45.03 61.93 18.03 94.03	10.33 64.33 65.33 67.33 17.33 15.00 13.00 20.00	100	1
71.63	64.33	162	1
94.01	65.00	133	1 1 1 1 1 1 1
57.43	61.33	121	1
39.JT 56.01	17.30	1.	1
56.01	15.00	100	1
18.61	13.00	103	1
35.JJ	20.06	11.	1
67.33	59 .JC	1:1	1
9700	14000	133	1
60.03	6.00	133	1
55.00	22.66 64.33	100	1
18.60 83.60	64.00	100	1
83.60	69.0]	100	1
76.07	58.00	133	1
76.01 90.01	49.00	10)	1
76.03	18.0J 62.00	1	1
94.13 45.33	62.00	137	1
	54.13 2.00 16.00 18.00	1.3 1.3 1.3 1.3 1.3 1.3 1.3 1.3 1.3 1.3	1 1 1 1
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89.0	18.00	15	1

PROGRAM OUTPUT---

A-BOLVIED	4N3 (8AC)	EXPOSURE
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35•′)		11.59
35.∂0	3330	26.02
35	:31	27.51
35∙⊎3	:300	31.29
35.00	3333	33.31
34.€3	1314	33.87
34.27	3914	33.87
33.93	0914	33.87
33.53	J914	33.87
33.53	0190	35.64
33.93	.3914	43.47
33.53	3914	41.87
32.73	1828	43.02
35.83	4573	46.27
33.45	0914	46.33
33.83	.0914	48.0C
31.19	.0914	54.24
29.23	4573	57.02
30.35	.2742	F9.41
32.32	. 457 1	€0.38
33. 5	.1828	50.38
34.13	.2742	6t • 3 8
32.22	457	6:.38
	35.13 35.13 35.13 35.13 35.13 35.13 37.13	35.133333 35.033330 35.033330 35.033330 35.033330 35.033330 35.033330 34.633314 33.533914

Appendix C5
Uncertain Model Output
(see Table 5)

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Appendix C6
Manual Model Output
(see Table 6)

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                                           35.81
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                           -. 4571
                                            45.56
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18 ....
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77, 71
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              37,23
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10 .50	37.15	1 5 7	24.94
15 .35	37.37		25.33
10 . 0	37.17	! ?	25.1
	37.23	- 67	31 • 3 •
	37.1+	• • • • •	
11		- • 1 : 7	35 • 43
11	37.27	- • ! ! ? · · · · · · · · · · · · · · · · ·	25.55
10 .:	37.27		3 . 41
16 % .	37.35	• • • • •	31.65
11 .03	37.74	6 1 7 .	33 • / 2
11 .03 15 .	37.13	167	23 • 26
11.7.35	77,23	- 6 5 7 1	28.50
10	77.75	-	28.66
16 .50	32.33	1571	39.33
17.00	37.29	7	31.26
18.051	77,27		31.00
	37.25	- 1 5 7	27 • 42
15 .		- 1 ~ f	
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11 .::	33.32	5 7 .	35.17
11	77.13	- 5 ?	31.39
10.001	37.14	- · · · · · · · · · · · · · · · · · · ·	34.63
16.001	37.15	- · · · · · ·	27.61
11.076	32,22	157	38 • 18
10	33.13	- · · ! 7,	27.78
10 - 33	33.21		29.65
100.00	37.25	-· +7.	31 . 25
10 .05	37.23	1571	30 ⋅ ∞2
10 . 01	37.15	657	33.59
10 .36	33.22	4176	31.33
16 . 1.	37.25	457	28 • 17
16 90	37.15	4571	36.79
100.05	₹₹ . 27	57	28.64
11			35 • 74
	33.27	457	
10 1.35	37.11	LF7	23 • +4
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16 • 31	37.25	- • 1· F. 7 ·	31 • 8 3
11 00	33.35	- • • ₽ ? .	29.17
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Lf .i.	34.13	57	312
11 1	37.13	- 1.57	34 . (4

Appendix C7

Data for Points on Figure 13

163.00	37.31	4576	47.36
10 . 5 "	33.32	4576	46.36
10 . 1	37.14	4570	58.84
10	37.35	4571	52.33
16 .73	33.32	4570	49.46
	37.32	- 4576	53.19
10			
10	33.31	4576	49.15
18 .5.	33.17	4576	50.29
162.33	33.5+	4576	49.94
10 .3	33.23	457	45 • 34
10 .0.	37.53	4F76	39.71
150.00	37.2+	457.	43.44
1875	33.51	4573	46 • 67
10 / . 55	33.13	4570	45.11
10 .72	33.+3	4570	42.31
10 .00	33.33	1570	46.77
16 2. 27	37.55	4571	48.69
160.56	33.5+	F70	43.07
16 '- 55	33.)+	4570	44 . 24
10 .00	33.31	4570	44.00
11 .[1	37.12	4570	46.48
101.50	33.23	4570	42.05
100.30	33.35	457	44.52
18 03	33.15	4570	42.15
10 .60	32.35	4570	48.70
			43.83
10 % 30	33. +5	4570	
100.01	33.51	4575	45.31
100.00	33.15	4570	42.30
16 . 07	33.25	4570	43.23
107.10	33.31	4570	45 • 32
107.00	33.43	4570	53.80
100.00	33.37	4570	45 • 28
10 1.00	37.21	4570	43.89
16 .00	33.23	4571	51 • 54
10::30	33.35	4570	47.10
100.00	32.35	3571	44.44
163.58	33.37	4571	45.48
101.00	37, 32	4571	45.23
100.30	33.+1	4F76	45.39
133.00	37.31	4575	45 . 29

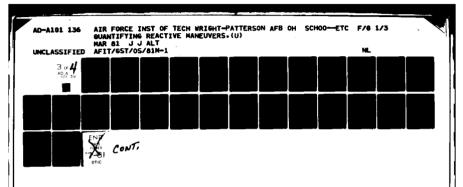
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33.31'
                        - . . 571
                                        48.87
100.00
10 . 10
             33.3+
                        -.4F76
                                        45.34
             33.17
                        -- 5570
                                        61.85
160.20
             33.73
                        -. 457(
                                        52.78
10:00
             33.31
                        -.4571
10 - . 50
                                        51.17
10..00
             33.35
                        - . 4571
                                        55.79
100.00
             33.31
                        - . 4570
                                        49.44
             33.15
                        - . 45.75
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37.97
                        - . 4570
                                        51.33
16 1.12
             33.25
                        -. 3570
                                        44.91
10....
15 . 70
                        -. 357"
             33.43
                                        41.31
16 .::
                        - . 4571
             33.13
                                        42.18
100.00
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16 .. 50
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16 . . . . .
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                                        42.61
10 .03
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                                        47.37
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             33.+5
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100.33
             33.53
                        -.4576
                                        44.88
             33.32
                        - . 4571
                                        48.78
100.00
             33, 25
                        -- 4471
                                        45.86
10 '- 00
             33.12
                        - . 4. F. 71
                                        46.16
160.33
                        -.457(
             33.23
                                        46.81
100.00
             33.43
                        -.457L
                                        44.38
101.00
             33.17
                        - . . 570
                                        43.16
                                        49.27
             37.31
                        - . 457L
100.00
160.00
             33.43
                        -.4570
                                        47.35
101.00
102.00
             33.75
                        -.4575
                                        45.56
                        -.4575
             33.15
                                        41.68
                        -.4576
160.00
             33.15
                                        43.41
100.00
             33.31
                        -.4575
                                        44 . 13
             33.44
                        -.4571
                                        53.50
10..00
                        -. 4F71
10 . 23
             37.13
                                        45.15
100.00
             33.23
                        -.4570
                                        43.92
107.50
             33.22
                        -- 4570
                                        54.61
10:.00
             33.72
                        -.4571
                                        53.43
101.10
             32.3+
                                        44.27
                        --4576
100.00
                                        44.38
             33.35
                        -.+570
160.30
             37.35
                        - . uf71
                                        44.03
10 .. 00
             33.41
                        -.457(
                                        45.68
100.91
             33.23
                        -.4571
                                        46.91
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107.09	33.27	4570	53.72
10 3.33	33.13	u 57 i	44.83
103.00	33.11	+575	55.87
180.50	33.7+	4570	57.78
1033	33.22	457(54.58
107.50	33.35	4570	55.28
160.05	33.25	4570	43.30
157.33	33.23	4571	54.98
103.50	33.57	4570	51.67
1003	33.23	4576	47.84
109.93	33.51	457i	46 • 86
10 1.33	33.1+	457'	47.64
16 . 50	33.51	+576	47.93
10 . 50	33.23	4571	51.40
15 1.22	33.+1	4570	49.46
100.00	33.23	4575	51.54
100.00	33.41	4570	52.47
160.50	37.25	4570	48.27
10 '.00	33.34	4571	48.98
100.35	33.21	4570	47.65
103.00	33.25	4579	41.11
183.35	33.22	4570	47 • 98
109.36	33.45	4576	48.52
100.33	33.37	4570	46.54
107.30	32.35	+570	5 2•92
100.30	33.+4	4576	45 • 48
100.05	33.52	4570	48.56
169.00	33.21	4570	48.92
180.00	33.17	4570	46.5 6
100.35	33.23	4576	50.89
16'.33	33.44	457C	54.01
100.00	32.33	4576	45 • 29
103.00	33 • 23	4570	45.73
10 1.00	33.25	-•+57í	47.58
10 '. 00	33.52	1571	53.56
167.60	32.35	4570	44.69
10 (. 30	37.37	4571	48.01
10 .30	33.35	4570	50.49
109.31	33.+5	- • 1 57 5	51.63
100.00	33.21	-•4576	47.65

			~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~
109.00	33.34	4570	52.22
1031	33.34	457:	54.01
100.30	33.75	4570	60.86
1650	33.73	4576	52.39
100.08	33.3)	4570	53.56
100.00	33.27	1575	54.43
10 .00	33.31	4570	51.71
100.00	33.17	4570	52.64
103.00	33.92	4570	61.67
100.00	33.23	4576	47.10
104.00	33, 72	4575	51.02
101.00	33.27	4576	48.43
100.00	33.32	4578	52.72
101.00	33.35	4570	49.12
10	37.53	4576	47.48
104.00	33.31	4571	50.11
100.01	33.+7	4570	52.87
10.00	33.35	4570	48.24
101.06	33.84	+573	50.82
100.00	33.25	4578	51.90
10 1.03	33.17	4576	51.60
184.80	33.23	4570	47.47
107.05	33.42	4570	44.32
100.00	33.11	4570	45.57
109.30	33.29	4570	50.35
160.00	33.34	4570	55.94
107.00	33.73	4570	46.47
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160.50	37.32	4570	45.52
109.00	33.33	4570	46.50
100.00	33.39	4576	57.52
163.60	33.21	4570	50.25
100.00	33.13	4578	42.44
101.00	33,25	4571	54.52
100.05	37.57	4570	56.94
100.00	33.15	4575	48.56
189.30	33.24	4570	49.69
100.00	37.35	5576	52.81
167.20	37.31	1571	49.51
10:000	37.23	576	48.53

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100.00	33.22	4570	41.14
163.06	33.25	4575	39.81
1633	37.23	4570	49.95
100.00	33.73	4:71	46.99
100.00	33.2+	4570	38.12
100.00	33.21	4570	46 • 118
151.33	33.23	4570	38.69
102.00	33.35	4576	44.74
100.30	37.33	4570	54.88
	33.22	4571	
101.35			42.11
100.00	33.47	4576	31.50
101.00	33.23	4570	38 • 55
107.80	33.33	4570	47.28
100.00	3 ³ •35	4578	39.89
100.00	33.42	4578	38 • 47
101.75	33.23	4576	49.83
10".26	33.45	4571	44.91
18 . 33	33.13	4570	44.68
103.00	33.3+	4571	44.49
101.03	33.17	4570	40.83
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103.03	33.15	4576	40.82
1000	37.37	4570	40.89
100.00	33.11	4570	36.68
18 50	33.25	4576	45.00
169.38	33.33	4576	43.83
103.03	33.75	4570	43 • 29
100.00	33.15	4575	37.42
163.85	33.37	4578	30.69
10%.00	33.23	4570	35 • 97
107.00	33.41	4570	47.15
10:.00	33.37	4576	36.59
103.33	33.29	4570	37.58
100.30	33.27	4575	44.75
		4570	
167.36	33.53		49.18
10 1.73	33.13	4575	38.97
10:00	33.25	4570	36 • 47
103.33	33 • 23	4570	44.18
103.30	33.33	4570	39.41
10).55	33.7+	4573	36.14

100.70				
103.00 33.27 4570 34.61 103.00 32.37 4570 44.61 10.00 33.73 4570 34.83 103.00 37.30 4570 34.83 103.00 33.27 4570 37.22 107.00 33.33 4570 35.20 107.01 33.33 4570 35.20 103.01 37.13 4570 31.39 103.01 37.47 4570 31.39 103.01 33.13 4570 31.54 103.01 33.31 4570 31.54 103.01 33.33 4570 32.44 103.01 33.33 4570 32.44 103.01 33.33 4570 32.84 103.01 33.35 4570 32.89 103.01 33.35 4570 32.53 103.01 33.35 4570 32.39 103.01 33.35 4570 32.39 103.01 33.31 4570 36.42 103.01 <td< td=""><td>100.00</td><td>33.24</td><td>4570</td><td></td></td<>	100.00	33.24	4570	
103.10 103.30 103.734570 39.51 103.30 173.734570 34.83 103.00 33.274570 33.81 103.10 33.334570 33.81 103.11 33.334570 37.82 103.51 33.334570 37.82 103.51 33.334570 37.82 103.51 33.474570 31.39 103.51 33.474570 32.44 103.60 33.334570 33.84 103.11 33.534570 33.84 103.11 33.534570 32.53 103.274570 33.84 103.20 33.354570 32.39 103.27 33.354570 32.39 103.27 33.354570 32.39 103.27 33.354570 32.39 103.30 33.314570 30.94 103.00 33.334570 30.94 103.00 33.334570 30.94 103.00 33.334570 30.94 103.00 33.334570 30.62 103.00 33.334570 30.62 103.00 33.334570 30.62 103.00 33.334570 30.62 103.00 33.334570 30.62 103.00 33.334570 39.85 103.00 33.334570 39.85 103.00 33.374570 39.85 103.00 33.374570 39.85 103.00 33.374570 39.85 103.00 33.374570 39.85 103.00 33.374570 39.85 103.00 33.274570 39.85 103.00 33.274570 39.85 103.00 33.274570 39.85				
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103.00 33.27 4570 33.81 103.11 33.33 4570 35.20 103.50 33.32 4570 37.82 103.01 33.13 4570 31.39 103.01 33.13 4570 32.44 103.00 33.31 4570 31.54 103.00 33.34 4570 31.54 103.00 33.34 4570 35.26 103.10 33.35 4570 33.84 103.00 33.45 4570 33.84 103.00 33.35 4570 32.53 103.00 33.35 4570 37.16 103.00 33.35 4570 37.16 103.00 33.15 4570 31.90 103.00 33.15 4570 30.42 103.00 33.35 4570 30.94 103.00 33.35 4570 30.94 103.00 33.35 4570 30.94 103.00 33.37 4570 30.94 103.00 <t< td=""><td></td><td></td><td></td><td></td></t<>				
107.57 33.33 4576 35.20 103.52 33.32 4576 37.82 103.61 33.13 4576 31.39 103.61 33.13 4576 32.44 103.61 33.31 4576 31.54 103.60 33.81 4576 31.54 103.60 33.37 4576 31.54 103.60 33.37 4576 35.26 103.60 33.37 4576 32.53 107.61 33.57 4576 32.53 107.61 33.37 4576 32.53 107.61 33.37 4576 32.53 107.60 33.13 4576 32.39 107.60 33.13 4576 37.16 107.60 33.15 4576 31.90 107.60 33.45 4576 30.94 107.60 33.37 4576 29.20 107.60 33.37 4576 27.58 107.60 33.32 4576 29.46 107.60 <t< td=""><td></td><td></td><td></td><td></td></t<>				
103.52				
10 1.00 37.13 4576 31.39 10 1.00 37.47 4570 24.80 10 1.00 33.13 4570 31.54 10 1.00 33.31 4570 31.54 10 3.35 33.34 4570 35.26 10 1.00 33.53 4570 28.94 10 1.00 33.45 4570 32.53 10 1.00 33.35 4570 32.53 10 1.00 33.15 4570 32.39 10 1.00 33.15 4570 31.90 10 1.00 33.15 4570 30.94 10 1.00 33.15 4570 30.94 10 1.00 33.31 4570 30.94 10 1.00 33.31 4570 30.94 10 1.00 33.37 4570 29.20 10 1.00 33.32 4570 30.94 10 1.00 33.32 4570 29.46 10 1.00 33.32 4570 29.46 10 1.00 33.32 4570 29.46 1				
10 1.57				
100.01				
102.00 33.314570 31.54 103.00 33.344570 35.26 103.10 33.594570 28.94 103.00 33.454570 32.53 103.00 33.154570 31.90 103.00 33.154570 31.90 103.00 33.154570 30.42 103.00 33.154570 30.94 103.00 33.314570 30.94 103.00 33.314570 30.94 103.00 33.314570 30.94 103.00 33.314570 30.94 103.00 33.314570 30.94 103.00 33.314570 30.94 103.00 33.314570 30.94 103.00 33.314570 30.62 103.00 33.324570 30.62 103.00 33.334570 39.85 103.00 33.374570 39.85 103.00 33.374570 39.85 103.00 33.334570 39.85 103.00 33.334570 39.85 103.00 33.334570 39.85 103.00 33.334570 39.85 103.00 33.334570 39.85 103.00 33.334570 39.85 103.00 33.334570 39.85				
103.36				
103.10	100.00	33.81	457(31.54
100.30 33.27 4570 33.84 100.00 33.35 4570 32.53 100.00 33.35 4570 37.16 100.00 33.13 4570 31.90 100.00 33.15 4570 31.90 100.00 33.15 4570 36.42 100.00 33.31 4570 29.20 100.00 33.34 4570 38.97 100.30 33.33 4570 29.46 100.30 33.31 4570 29.46 100.31 33.31 4570 29.46 100.31 33.31 4570 29.46 100.31 33.32 4570 29.46 100.30 33.33 4570 29.46 100.30 33.33 4570 29.46 100.30 33.33 4570 29.12 100.30 33.33 4570 29.12 100.30 33.23 4570 29.65 100.30 33.71 4570 35.90 100.30 <t< td=""><td>100.05</td><td>33.34</td><td> 4570</td><td>35.26</td></t<>	100.05	33.34	4570	35.26
101.01	160.20	33.53	4571	28.94
107.30	100.30	33.27	4570	33 . 84
107.30	101.71	33.45	4570	40.28
100.00 33.134570 32.39 101.27 33.134570 31.90 101.00 33.134570 31.90 101.00 33.134570 36.42 101.00 33.424570 29.20 101.10 33.344570 30.94 101.00 33.454570 29.46 101.00 33.314570 29.46 101.00 33.754570 27.58 101.31 33.314570 30.94 100.31 33.324570 30.62 100.00 33.214570 30.62 100.00 33.334570 39.85 100.00 33.334570 39.85 100.00 33.334570 39.85 100.00 33.234570 39.85 100.00 33.234570 39.85 100.00 33.234570 39.85 100.00 33.234570 39.85 100.00 33.234570 39.85 100.00 33.234570 39.85	107.30		457(32.53
100.00 33.134570 32.39 101.27 33.174570 31.90 101.00 33.174570 36.42 101.00 33.174570 29.20 101.00 33.344570 30.94 101.00 33.454570 29.46 101.00 33.774570 27.58 101.01 33.314570 30.94 100.00 33.214570 30.62 100.00 33.314570 30.62 100.00 33.334570 39.85 100.00 33.334570 39.85 100.00 33.334570 39.85 100.00 33.334570 39.65 100.00 33.234570 39.65 100.00 33.234570 35.90 100.00 33.234570 35.90 100.00 33.234570 35.90 100.00 33.234570 35.90 100.00 33.234570 35.90 100.00 33.234570 35.90				37.16
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10 1.00 33.154570 29.20 10 1.00 33.424570 29.20 10 1.10 33.344570 30.94 10 1.00 33.454570 29.46 10 1.00 33.754570 27.58 10 1.30 33.314570 30.94 10 1.00 33.754570 26.81 10 1.00 33.324570 30.62 10 1.00 33.334570 39.85 10 1.00 33.334570 39.85 10 1.00 33.234570 39.85 10 1.00 33.234570 35.90 10 1.00 33.234570 35.90 10 1.00 33.234570 35.90 10 1.00 32.554570 36.86 10 1.00 32.554570 36.86 10 1.00 33.224570 36.86				
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188.33				
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103.00 33.334570 39.85 103.00 33.374570 29.12 100.10 33.234570 29.65 107.00 33.254570 35.90 107.10 33.714570 36.86 1010 32.854570 29.20 1010 33.224570 30.52 100.10 33.354570 28.77 107.00 33.274570 32.75				_
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100.33 33.234576 29.65 103.00 33.254576 35.90 103.30 33.734576 36.86 1030 32.354576 29.20 1030 33.224576 36.52 100.33 33.354576 28.77 103.00 33.274576 32.75				
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1030 32.856570 29.20 1730 33.226570 30.52 100.03 33.354570 28.77 100.00 33.274570 32.75				
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103-60 33-274576 32-75				
10 • 24 55 • +1 • • 4575 33 • 73				
	10 .74	55 • + L	4575	55.73



10 1.00	33.32	4576	27.10
10".00	37.95	4571	27.64
100.35	3?.+3	4570	33.32
169.00	37.17	4576	39.31
18 /. 10	32.45	4579	33.83
10 .00	33.25	571	33.65
16 .00	33.55	5570	28 • 27
10 - 01	32.53	457	34 • 66
11 33	37.11	45 75	39.31
16 . 63	33.51	4571	29.13
10 20	32.57	571	35 • 36
109.00	32.29	4579	34.35
10 3.39	33.53	4576	28 • 27
1033	33.53	4571	29.17
16 .3	32.43	657.	34.11
10 .00	32.35	4570	35.36
103.02	33.33	4571	25.41
180.00	32.57	4570	33.15
103.00	32.32	4570	35 • 8 3
10 . 9:	33.53	4570	29.23
101.55	32.34	457	34.68
110.00	33.3+	4570	38 • 87
110.00	37.53	4570	35.26
10:.33	33.43	4570	29.33
10 / 30	33.4+	 ≒57€	37 • 84
100.00	33.12	4576	34.09
100.00	33.53	4576	29.09
160.00	₹₹.59	4570	27.74
100.00	32.44	4570	33.57
100.00	33.47	4576	33 • 39
105.30	32.43	4575	33.63
160.00	33.37	457ú	32.96
103.50	32.51	4570	38 • 8 3
103.01	33.51	571	27 • 66
1080	32.45	4576	33.67
107.00	32.57	4571	35.53
11 .50	37.31	576	26 • 41
100.00	33.75	4577	37.33
11 60	32.31	2571	36 • 7 3
103.00	33.43	4571	28.73

10 20	33.34	4571	39.01
10 . 30	37.53	4571	41.39
1038	32.57	6876	47.96
10.000	37.72	45.71	5° • 75
10 35	33.25	4571	49.69
10 .35	37.33	+571	43.99
16 % (8	33.57	4571	38.31
1000	32.51	457	42.94
10 51	33.13	6571	51.21
107.20	33.53	4576	37.19
16 . 6	37.31	4F7	44.35
10 .00	32.3?	4571	47.44
10	33.53	4576	46.77
10 .00	32.23	4570	43.51
18 1.08	37.31	457i	45.14
1032	32.95	4576	45.93
100.76	33.27	4571	39.72
10: . 23	32.27	4571	44.76
160.06	32.33	570	46.23
18 .00	37.33	-0.576	43.37
10 .3.	33.52	457	37.52
100.75	37.37	457[52.22
16 3, 25	33.7.	4571	45.58
100.39	37.52	4571	43.75
10 30	33.25	4575	49.64
100.00	33.55	4571	39.01
10 00	33.55	4570	40.26
100.00	33.31	4570	41.98
10 33	32.55	4570	46.01
103.00	32.57	4571	47.00
100.00	32,23	4576	44.80
187.50	32.23	4575	44.29
103.33	32. +5	4571	42.84
100.00	33.53	457.	41.37
10 100	32.29	4576	45.71
10	33.33	4570	45.60
15 1. 3.	33.62	- · · F 7'	37.76
1030	33.17	4571	49.37
10 1.51	32.51	157	43.23
10 .03	73.52	570	49.99

109.00	33.73	1576	39.92
102.00	33.33	4571	41.07
100.00	32.53	4578	45.26
103.30	33.55	4 5 7 7	51.33
100.00	33.7+	4 F7(38.89
100.70	33.3+	457i	47.20
107.05	33.32	4570	44.07
100.00	32.25	4576	48.46
10 /.05	33.13	4575	51.53
10 .00	33.53	4575	39 • 4 ü
16 1.00	33.31	4578	45.22
103.00	32.57	457(45.93
100,00	33.57	+•177L	39.72
187.09	33.13	4576	44.13
10 00	33.33	457t	46 • 9 û
10 % 5 5	37.33	4571	45.31
100.00	33.43	457;	41.49
180.00	32.24	4571	45.38
100.00	32.32	4571	44.46
10 % 07	33.53	4576	43.81
100.00	32.53	576	45.41
169.00	33.33	+578	51.71
100.0c	32.53	4570	45 • 35
100.00	33.57	4570	40.44
100.00	33.19	4576	45.18
100.30	33.52	+F70	39 • 35
100.00	33.51	4570	39 • 96
100.00	33.73	4570	45.58
100.38	32.42	4576	46.63
160.5E	33.94	4576	44.31
180.00	32.27	457r	47.90
107.00	32.99	4570	45 • 83
103.00	32.53	4570	45.11
100.00	33.54	4570	41.29
107.38	33.+3	4570	45 • 65
101.00	33.33	4575	47.96
100.00	33.31	4571	39 • 44
167.51	33.33	4578	49 • 40
100.00	37.73	4570	41.13
169.83	33.52	457	40.93

107.00	33.37	4570	53.49
169.00	33.83	4570	40.18
10):	32.4+	4570	49.17
11 - 01	33.31	4571	44.76
167.23	33.75	4570	45.43
10.00	33.35	4570	45.99
1039	33.31	4570	52.64
10,.00	32.5+	4570	51.92
100.06	33.11	4570	57.12
100.00	33.51	4576	4r.28
180.50	32.43	4570	47.02
10 . 2	32.51	4571	47.76
10 .00	33.52	457!	45 • 24
10,.33	33.53	4576	45.65
100.00	33.72	4575	51.67
103.30	33.3+	570	39.60
10 .00	33.47	4576	45.20
107.33	32.51	4571	46 • 27
100.00	32.5+	4575	51 • 17
103.30	33.57	4576	45 • 95
167.00	33.45	4570	50.67
163.30	33.53	4570	45.69
100.00	32.43	4576	55 • 24
100.00	37.+3	4570	42.80
105.39	33.52	4570	43.89
105.00	32.++	4570	48.61
104.78	33.57	4570	45.46
1651	37.75	4575	51.92
100.30	32.+7	4570	51.63
160.06	33.57	4576	54 • 15
100.00	32.57	4570	50.63
10.000	33.31	4571	40.85
100.33	32.49	4570	47.76
100.00	33.31	4570	40.58
101.00	33.21	4570	51.49
180.03	33.77	457!	45.98
10:00	33.33	1576	39.48
1631	33.13	657	49.23
100.00	33.72	4570	45.24
10 .53	33.43	457i	53.13

100.00	33.33	457(31.49
10 1.55	33.17	4571	39.31
107.00	32.44	4571	40.52
16 % 60	37.25	171	34.13
10 . 13	33.34	457	37.66
169.11	33.35	576	35.44
180.00	33.23	4571	37.14
			46.24
101.00	32.55	1570	
150.05	33.12	457[33.69
160.01	32.35	457.	33.06
100.33	33.55	+570	32.16
10 - 21	32,51	57	35.91
100.00	33.57	4570	30.06
1853	33.53	4571	30.71
1001	32.43	4571	39.17
102.30	33.74	4570	33.83
16 80	37.+5	4571	31.45
100.30	32.52	457t	35.91
10:.33	32.54	4570	35.67
100.30	33.53	57(34.38
10.00	37.25	4571	34.38
100.00	33.55	4570	30.75
100.01	33.73	4570	36.73
160.50	33.+3	1571	30.79
100.00	33.52	457.	37.06
160.60	33.13	4570	37.62
100.00	37.53	4570	44.68
100.00	33.75	4570	36.23
109.33	32.45	4571	36.13
169.55	33.7+	570	35.18
107.00	32.44	4570	35.97
100.35	33.52	45.70	29.76
105.03	32.51	4570	35.73
100.00	33.43	4570	38.79
107.50	33.33	4571	40.38
163.55	32.33	4576	35 . 69
109.00	33.77	4575	30.95
100.00	32.43	4571	36 .96
100.50	33.73	417.	29.70
100.00	33.43	4570	34.78

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101.20
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10 50	34.12	57.	29.22
16	34.17	1.57	32.36
16 .50	33.55	6570	34.04
100.00	37.57	57	33.91
10 .53	33.53	4576	34.32
15 .00	33.53	- · · 57'	34.24
11 .33	33.31		35.28
11.	37.53	57.	36.96
11	33.34	4575	37.31
15 .5	35.15	57/	30.88
1(37.75	1170	32.82
11 ."	33.43	957	35.20
113.00	33.35	157.	36.71
10.2	33.53	571	35.17
10 .1	37.72	- · · · 5.7	33 • 11
10	37.+1	: 57.	32.67
11 .50	34.71	457.	34 • 36
10 .00	33.41	- • = 57 6	37 • 49
10 .1:	33.45	57	32 • 6 3
10 .00	33.9.	4575	35.00
15	37. 73	1571	31.71
10 .00	33.73	557(34.62
1000	37.55	1 57	33.74
10 .33	34.17	3570	30.56
16 . 56	33.37	4576	33.97
100.00	37.55	4570	33.55
10:00	3 11	4576	33.24
100.00	37.74	>570	36.33
1039	33.55	457î	34.85
10 .00	37.52	457(33.55
10 .00	33.53	4576	34.63
10:016	33.41	l. 57 t	34.73
16 .00	33.33	4570	36.85
11 .32	37.55	1571	35 • 05
10 426	33.47	4575	38 • 26
16	37.55	571	37.94
10,000	33.33	-4-576	32.47
10	37.43	457	36.93
17	33.+5	- •! ₹?	34.28
15 .00	33.95	4571	38.59

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10 1.00	33.59	57(47.33
11	37.33	1571	+3.65
1023 10	33.53	:576	45.66
10	37.3+	1575	47.36
11 .33	37.43	457(44.56
10	33.35	4570	45.74
10 .00	3++ 27	(F75	45.55
10	37.53	4.E7.	45.96
10 .11	37.31	4571	47.73
11	34.73	4570	40.86
16 .77	37. +5	1571	44.35
11 . 0	37.49	1576	45.66
10 . 1	37.75	4570	49.85
10 .01	33.45	4576	48 • 47
1f •10	37.4+	-,:57	47 • 36
10	33. +3	1576	45.30
15.00	34.13	4571	41 • 68
17 . 0:	37.+5	4570	48.90
16 . 1.	33.53	457i	44.59
17 .20	33.31	4171	48 - 45
16 .00	33.53	4576	47.43
16 .52	37.33	4576	48.77
100.26	33.43	4576	45.81
15 .5:	31.15	- •457!	39.91
100.30	33.73		47 . 44
16%.8%	33.51	4576	41.87
10 .33	33.37	-•457t	45.87
160.90	37.74	4571	46.87
16 37	33.33	4571	46 • 58
107.57	37.55	4571	46.62
100.33	33.55	4576	43.33
10 . 00	33.57	4576	44.53
10 1.00	3×.57	4575	42.31
100.00	33.4.	1575	43.72
105(37.33	570	48.67
101.00	33.52	3570	48.66
10 1.67	37.52	~. . £7 ,	40 . 20
15 .37	33.55	- 417	49.59
10 .7.	₹₹. • +		39 • 8 3
16 .10	37.95	 457.	49.29

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10 +30	3 to 15	- •457.	44.58
16	31.11	157	46.25
10 .00	37,57	4575	49.63
10.00	37.33	57:	56.19
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1052	3' • 15	RE76	45.32
15 • 5	33.37	F ?	53.00
15 .35	33.31	4571	52.13
10.001	37.53	£71	49.50
10 - 30	33.35	- 1.7	53.65
	31.17		
15 • 7		4571	46.13
1035	33. ¥£	4571	51.13
25 . 50	37.52	57:	49.91
1/ 1.33	3	576	53.68
17 .70	33.57	457	48.62
1801	37. / L	4571	
44 65			47.57
11 .5	37. +2	- • 457	48.16
15	35.00	F7.	49.22
103.00	33.+2	157:	52.3?
10 .50	33.73	557t	51.73
16 . 50	33.91	457.	51.37
11 .56	37.75	457.	49.76
100.33	33.35	E71	47.00
16	37.55	4F7°	48 • 34
15 .35	33.74	4576	46.29
164.00	33.97	857(48.53
10 . 53	33.51	1.571	51.06
16 .00	34.17	4570	49.43
15 .30	37.73	to # 7 *	50.49
16 .36	3 · 5 /	4570	50.64
10 1.30	37.71	4570	49.77
16 .33	33.+2	578	52.40
160.96	33.+3	6576	45.92
10,00	33.43	4570	48.63
15 . 31	37.45	- · 4.5.7.1	49.13
16 .00	33.35	+57	50.43
16 .	37.57		49.91
10-10-	32.15	1571	42.20
10	33.75	- • • 571	50.45
11 .33	33. 15	•• F7t	45.29
14	33.37	 3571	55.78

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10 . 31	34.34	570	44.19
11	3' • 25	 57.	41.61
10.33	34.13	576	45.37
11	313	- •≒₹?६	48.63
1030	33.31	4570	46.77
10 .05	37.37	4571	45.82
10 .10	33.33	7 :	44.77
10 .	3×, 33	557	45 • 62
10	3 23	4571	49.53
16 .::	3 1 +	571	41.19
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15 .37	3113	4571	45 • 27
18 - 90	34.73		46 • 19
10 .50	33.91	57.	45.90
10 .11	35.53	4571	46.37
10 .11	33.31	3571	45.76
10 .00	34.13	1571	41.61
10 .10	33.33	4476	49.97
17 .00	37.32	4571	44.91
1031	3 2 -	*576	48.37
10	33.93	457	45 • 8 3
163.30	34.25	4570	49.41
100.05	33.93	4575	46.67
10 .00	34.23	4570	43 • 84
100.35	33.37	4575	47.77
100.10	34.31	6571	45.81
107.54	33.97	4571	47.51
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10 1.33	34.13	-•457(45.65
100.00	34.33	4571	43.78
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10 57	37.82	4571	46.24
15:.36	3 - 1 1	4577	45.72
10 000	3 . 31	- · · · · · · · · · · · · · · · · · · ·	4n • 63
18 .50	33.95	1 5.7	49.22
10	37.33	- • • 171	43.65
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16 :. 30
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10 ....
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                                           31.32
18 1.04
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16 % 45
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15 ....
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10 ...
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10 .00
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                                           28.38
101.00
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10 ... T
              37.05
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10 ...
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10 0000	3:•23	! ! 7	43.36
16	34.33	(17)	41.72
1'	34.35	4.571	41.42
1f	3 • 2*	4571	42.16
10 .99	34.33		43.39
	37.73	57	42.78
1: . 1: .:	76.75	(67)	
	36.35	4570	46.52
1	31.77	fr.	43.74
16 •	3 1 • 13	-• : 7:	42.61
1	34.35	427	41.12
11 .	34.45	- · · 17	42.53
11	3++23	57.	41.13
1	34.33		44.16
14 .35	3 4 . 22	- · · · · · · · · · · · · · · · · · · ·	43 • 41
10 .	31.37	17	42.02
17 • 1	35.41	- •0₹7t	41.23
1 " ·	31 • 21	• • • •	34.87
14 .17	3 • • 3	57;	43.56
10	3 75	57	45.42
10	35.37	6 5 71	44.42
16	3 ?)	- 4571	39.12
16 .50	31.37	£7(42.37
10	34.33	457	45.12
10 .00	34.15		41.71
10 .00	34.43	57	41.82
10	34 • 45	:571	41.96
16	34.25	457	40.13
10:418	34 • 25	457	47 • 24
16 . 41	34.32	457	42.23
11 .31	3′••)		
18 .3(34.35	- 4 5 7 1	44.30
15 .55		457	38.65
10.00	3+1	- 65 7 °	44.41
	34.13	4576	37.62
15 - 22	34.33	- 1571	43.91
	34 • 13	- • • • 7 6	4! . 88
10 .36	37.31		41.86
11	34.23	457:	37.88
1f .ce	34.42	- · · · · · · · · · · · · · · · · · · ·	41.73
10 .00	3' • + 1	1171	45 · 27
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10 .01	3 - • 2 *	1.57	47.37
16 .	3 1	571	51.76
10 .07 16 .0 17 .0	33.41	457	47.70
1 .	3		51.12
11	3 4 35	4571	48.17
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10	34.33	- 657	51.18
15 .7	3' • 75	- 457: - 457: - 457:	53.45
10	3 • 13	T • 1 1 1 1	5" • 37
10	3 · • • 3	* • · * y *	51 • 86
10.	3' • 25	4571	49.59
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15.00	34.75	457	52.53
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10 :.35	34.25	457	48.54
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10 . 7	3 . +7	570	49.23
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11	34.34	4575	57.35
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10.00	3 37	1575	49.38
10 . 00	34.33	457	49.43
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VITA

John Joseph Alt was born on August 31, 1949 in Ft. Wayne, Indiana. He was raised in the midwest and graduated from Andrean High School in 1967. He attended St. Joseph's College (Indiana) from which he received the degree of Bachelor of Science in Physics in June 1971. He was commissioned in the United States Air Force on completion of Officer Training School in September 1971. He completed navigator training and received his wings in July 1972 and electronic warfare training in February 1973. He served as a B-52 electronic warfare officer (EWO), an instructor EWO, and an EWO flight examiner with the 596th Bomb Squadron (Heavy), Barksdale AFB, Louisiana (1973-1977) and with the 43rd Strategic Wing, Andersen AFB, Guam (1977-1979). He entered the Air Force Institute of Technology in August 1979. He is married to the former Sharon E. Rayfield of Lancaster, S. C. He and his wife have a daughter, Heidi, and a son, Jeremy.

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QUANTIFTING REACTIVE MANEUVERS.(U)
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SUPPLEMENTARY

INFORMATION

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ERRATA

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